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TRANSONIC FAN/COMPRESSOR ROTOR DESIGN STUDY

Volume IV



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General Electric Company
Aircraft Engine Business Group
Advanced Technology Programs Dept.
Cincinnati, Ohio 45215

February 1982

Final Report for Period September 1980 - February 1982

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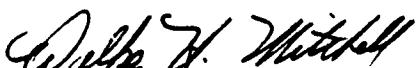
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This technical report has been reviewed and is approved for publication.



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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Volumes I through VI of this report describes the aerodynamic design of a series of five transonic rotors all parametrically related to a base- line design documented in Technical Report AFAPL-TR-79-2078. Each of the five designs deviate from the base line, in so far as practical, by a variation of one parameter only. The parametric variations are specified at the rotor tip. The original hub characteristics were preserved to the maximum extent practical. The varied parameter was adjusted along the span.		

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This volume describes the aerodynamic design details of the Phase III rotor. The Phase III rotor was designed to have a steeper average suction surface angle in the supersonic region ahead of the shock than the baseline rotor. This results in a smaller cascade throat area in the outer 80% of the blade than the baseline rotor. The hub region was kept essentially the same as the baseline rotor. The location of maximum airfoil thickness is 70% of length at the tip and 56% at the hub which is the same as the baseline rotor.



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VOLUME IV

PHASE III ROTOR DESIGN

Foreword

This Final Technical Report was prepared by the Advanced Technology Programs Department, Aircraft Engine Business Group, General Electric Company, Evendale, Ohio for the United States Air Force Systems Command, Air Force Wright Aeronautical Laboratories Wright-Patterson Air Force Base, Ohio under Contract F33615-80-C-2059. The work was performed over a period of one year starting in September 1980. Effren Strain (Captain USAF) was the Air Force Project Engineer for this program.

This report describes the results of an effort to aerodynamically define five rotor designs, all parametrically related to a base line design which could be evaluated by future testing in order to define the sensitivity of transonic blade rows to several design variables.

For the General Electric Company Mr. D.E. Parker was the Technical Program Manager for this program. Mr. M.R. Simonson was the principal investigator. Mr. A.J. Bilhardt was the overall Program Manager.



VOLUME IV

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LIST OF SYMBOLS AND ABBREVIATIONS

1. Used in Circumferential Average Flow Output Tables

STA	calculation station number	
WTF	total airflow	
PSIC	stream function (0 = tip (OD), 1 = hub (ID))	
z	axial location	inches
R	radius	inches
PHI	streamline slope	degrees
CURV	streamline curvature \curvearrowleft = neg., \curvearrowright = pos.	1/inches
VM	meridional velocity	ft/sec
CU	absolute tangential velocity	ft/sec
ALPHAM	absolute flow angle on stream surface	degrees
MM	meridional Mach number	
SL	calculation streamline number	
BLDBLK	flow blockage factor	$(\text{free area} - \text{blocked area})/\text{free area}$
PS	static pressure	psia
PT	total pressure	psia
TT	total temperature	degrees
BETAM	relative flow angle on stream surface	degrees
UREL	relative velocity	ft/sec
MREL	relative Mach number	
VABS	absolute velocity	ft/sec
MABS	absolute Mach number	
GAMMA	specific heat ratio	
PT-RAT	total pressure/inlet total pressure	
TT-RAT	total temperature/inlet total temperature	
RCU	radius x tangential velocity	in-ft/sec
CZ	axial velocity	ft/sec
PCT IMM	percent annulus immersion from tip (OD)	
RAD	average of leading and trailing edge streamline radii	inches
ACC PT	cumulative total pressure ratio	
RATIO	cumulative total temperature ratio	
ACC TT	cumulative total temperature ratio	

LIST OF SYMBOLS AND ABBREVIATIONS

1. Used in Circumferential Average Flow Output Tables (Cont'd)

AD.	adiabatic efficiency
POLY	polytropic efficiency
Axial VEL R	axial velocity ratio across blade row

2. Used in Stream Surface Blade Coordinate Tables

PT	point number	
PCT X	fraction of meridional distance from leading edge	
X	meridional coordinate on meanline	inches
Y	tangential coordinate on meanline	inches
B*M	meanline angle on stream surface	degrees
T(M)	thickness of blade perpendicular to meanline	inches
XS	meridional coordinate on suction surface	inches
YS	tangential coordinate on suction surface	inches
XP	meridional coordinate on pressure surface	inches
YP	tangential coordinate on pressure surface	inches

3. Used in Plane Section Coordinate Tables

Z	axial coordinate of stacking axis	inches
R	radius of coordinate system origin	inches
MU	tilt angle in axial direction	degrees
ETA	tilt angle in tangential direction	degrees
RHO	section height	inches
PT	point number	
ALPHA	axial coordinate	inches
ZETA*	meanline angle from axial	degrees
UPSILON	coordinate perpendicular to ALPHA and radius	inches
PCT AL	fraction of axial distance from leading edge	
T/C	local thickness/chord ratio	

SECTION XIII
DESIGN OF PHASE III ROTOR

1. INTRODUCTION

The efficiency of a transonic rotor is influenced by shock losses as well as losses due to cascade diffusion and secondary flow effects. The magnitude of the shock losses increases rapidly as the inlet Mach number increases. The average Mach number just ahead of the leading edge passage shock is influenced by the shape of the blade suction surface ahead of the shock. Increasing the average suction surface angle ahead of the shock reduces the average Mach number ahead of the shock and presumably reduces the shock losses. However, this results in a reduced cascade throat area. If the throat is too small, the cascade will not pass the design flow and may not achieve the attached shock pattern which is desired for minimum loss.

Also if the blade suction surface angle is made steep ahead of the cascade mouth, or covered portion, it may be necessary to have a rapid change in blade meanline angle at the cascade mouth to prevent the throat from becoming too small within the covered channel. A rapid change of suction surface angle increases the surface Mach number ahead of the shock and tends to worsen the shock-boundary layer interaction. This consideration may influence the optimum throat margin for best efficiency.

For cascades having an inlet Mach number greater than about 1.3, it is possible to achieve a net precompression of the flow ahead of the passage shock and still maintain a throat area sufficiently large to pass the flow.

The maximum flow that a transonic cascade can pass is set by the average suction surface angle in the flow induction region ahead of the first captured Mach wave, provided that the throat area is sufficient and not limiting. Hence any increase in suction surface angle must take place aft of the flow induction region or there will be a reduction of flow.

To get more definitive data on the effect of the suction surface shape ahead of the leading edge passage shock, and on the interrelation of the suction surface shape and the cascade throat area, the Phase III and the Phase IV blades were designed to have smaller throat areas in the outer 80% of the blade than the baseline rotor. While the throat areas of the Phase III and Phase IV rotors

were essentially the same, the Phase IV blade has some what less external compression and some what more internal compression. As a result, the Phase III blade has greater suction surface (and meanline) curvatures in the region of the cascade mouth than does either the Phase IV or the baseline rotors.

2. DESIGN PROCEDURE

The "data match" circumferential average flow solution and the Stream Surface Blade Sections (SBS) analysis of the baseline rotor previously described in Volume I were used as a starting point for the design of the Phase III rotor. However, a higher efficiency was assumed for the outer 80% of the flow since it was believed that the Phase III blade should have reduced shock losses and increased efficiency in this region. The rotor exit total pressure was maintained the same as the baseline rotor while the total temperature was reduced to reflect the assumed higher efficiency. The assumed chord-wise distribution of work was iteratively adjusted to obtain a calculated chord-wise distribution of static pressure which had a shape similar to that of the data match calculations of the base line rotor. However, the level of static pressure was higher in the outer portion at the rotor exit as a result of the assumed higher efficiency and consequent reduced energy input of the Phase III rotor. The resulting streamline static pressure distribution for the Phase III blade is compared with the data match of the baseline rotor on Figure 44.

The assumed streamline work input (as a fraction of the total streamline work) is plotted versus percent axial projection in Figure 45. The tip streamline is the one on the left. Each subsequent streamline is indexed to the right by the value of its stream function (fraction of the total flow from the tip). The dashed lines are lines of constant percent axial projection.

The blade meanline departure angles (the difference between the air angle and the meanline angle) were adjusted to achieve the desired suction surface contour in the forward part of the blade and yet maintain sufficient throat area to pass the desired flow.

After each modification to the chordwise work distribution and/or departure angles, revised blade annulus blockage and blade lean angles were calculated and input to the circumferential Average Flow Determination (CAFD) computer program

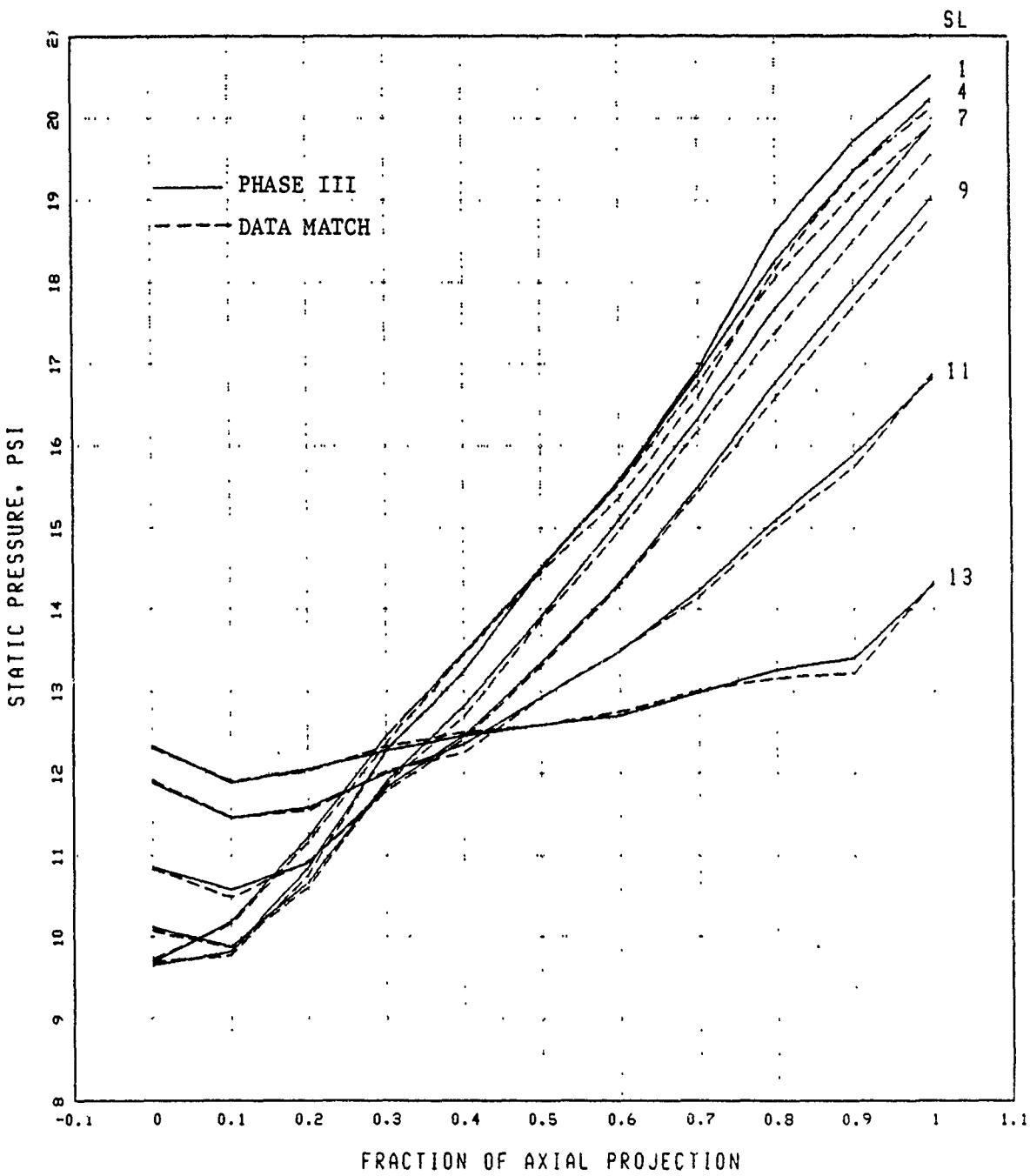


Figure 44. Phase III Rotor Static Pressure Distribution

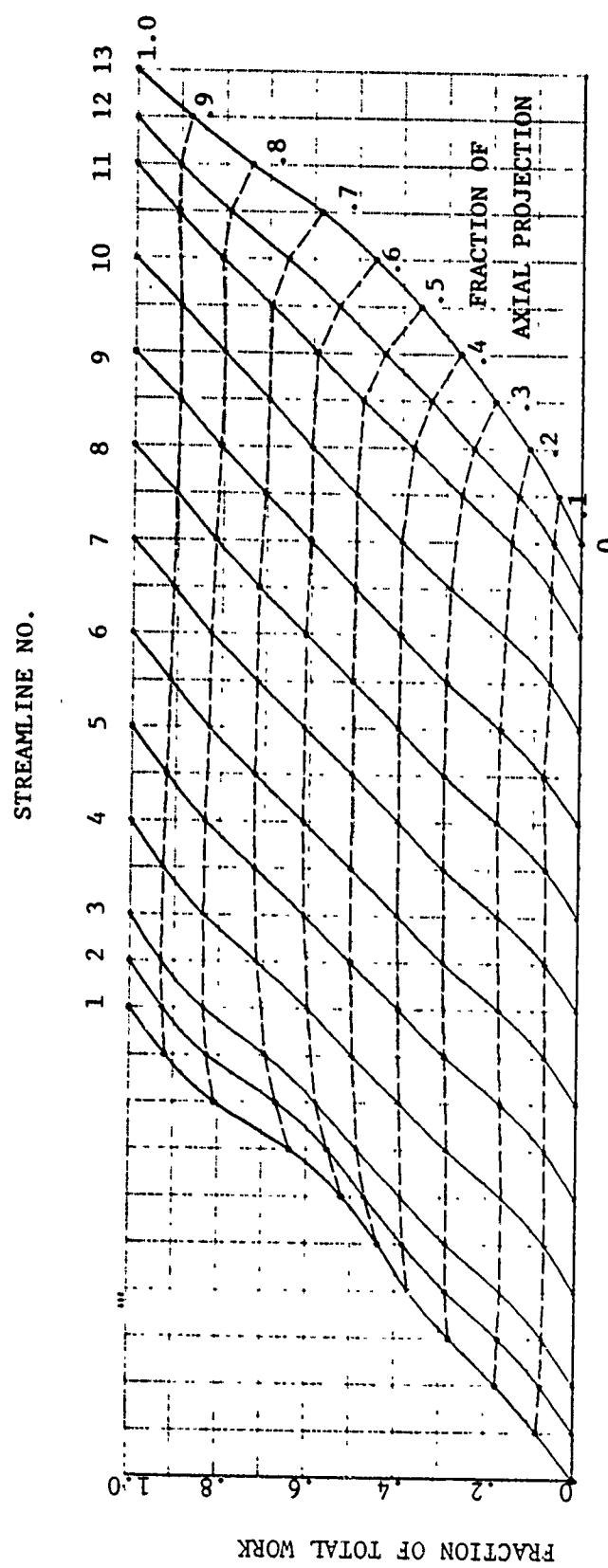


Figure 45. Phase III Rotor Intrablade Work Distribution

for the next iteration.

A method of characteristics computer program was used to analyze the flow in the cascade flow induction region for streamlines 3 and 6 to assure that the rotor would achieve the design flow. For other streamlines the differences between the suction surface angle and the "free-flow" streamline angle was compared with similar data from the data match calculations of the baseline rotor. This, then, was used as a guide in setting the suction surface angle in the flow induction region.

Figure 46 shows the radial distribution of the Phase III rotor throat margin compared with the data match case. It can be seen that the throat margin was set at essentially 5 percent in the outer 80 percent of the blade, except for locally at the tip where the margin is a little over 6 percent. The throat margin for a streamsurface blade section is defined here as the percent of excess throat area over and above the minimum theoretical area required to pass the streamtube flow at a throat Mach number of 1.0 and assuming a total pressure loss equivalent to a normal shock at the upstream Mach number. In a rotor the effect of radius change (between the leading edge and throat) on the relative total enthalpy and pressure is included.

The radial variation of incidence angle for the Phase III blade is shown on Figure 47. Since the blade thickness distribution and leading edge wedge angle are the same as the baseline rotor, the incidence angles were kept very close to the data match values of the base line rotor.

A modified version of Carter's Rule was used to calculate a reference deviation angle for the baseline rotor. This procedure converts the vector diagrams (from the data match calculations) to an equivalent two-dimensional set of vectors which would produce the same circulation as the actual blade taking into account the change in streamline radius and meriodional velocity. The difference between the deviation angle implied by the data match calculations and the reference deviation angle was then added to the reference deviation angle calculated from the modified Carter's Rule for the Phase III blade.

The radial distribution of the Phase III rotor deviation angle is shown on Figure 48 and the deviation angle minus the reference deviation angle is compared with the data match values in Figure 49.

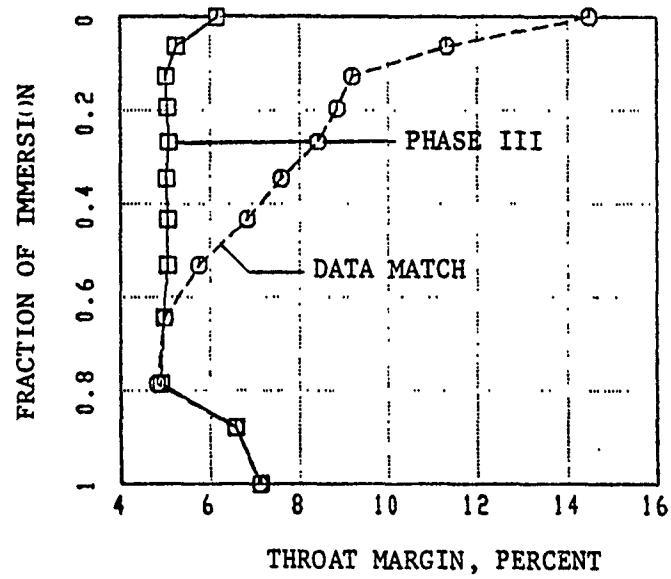


Figure 46. Phase III Rotor Throat Margin Compared With Data Match

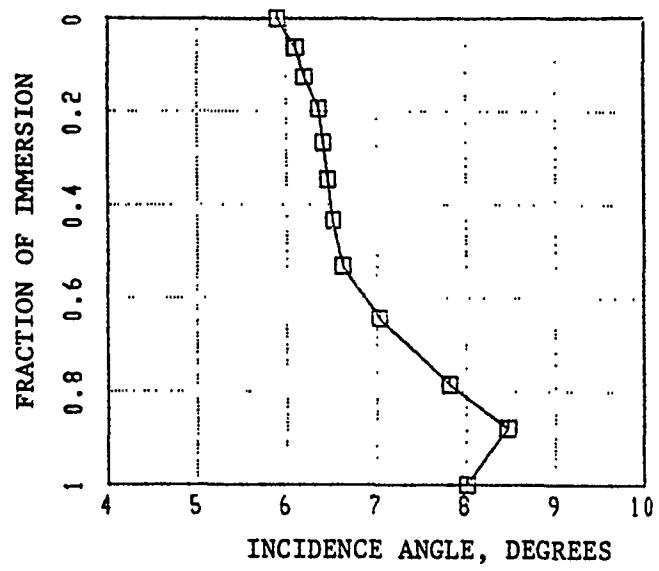


Figure 47. Phase III Rotor Incidence Angle Versus Fractional Immersion

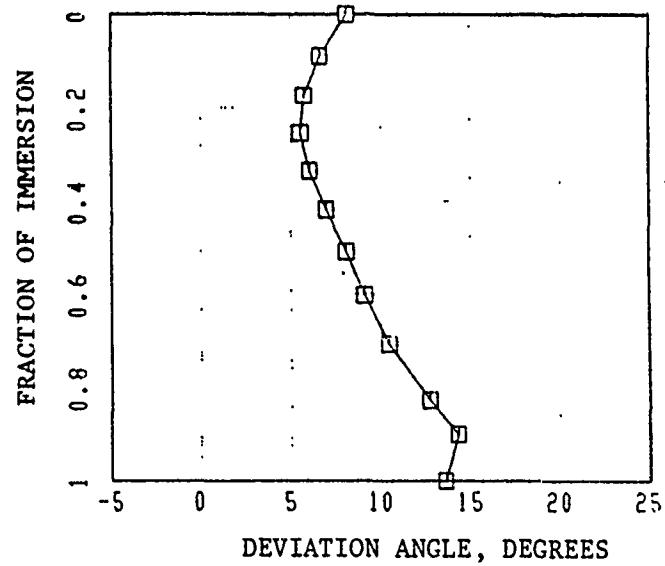


Figure 48. Phase III Rotor Deviation Angle Versus Fractional Immersion

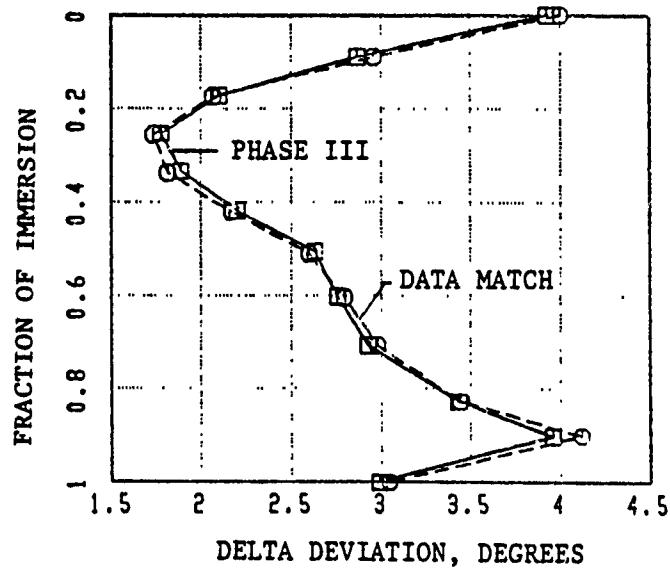


Figure 49. Phase III Rotor Deviation Angle Minus Reference Deviation Angle Compared With Data Match

A plot of departure angles for each streamsurface section is shown in Figure 50. Once the intrablade work distribution was chosen these departure angles were required to satisfy the desired incidence angles, deviation angles, and passage area ratios. The resulting streamsurface tip section of the Phase III rotor is compared to that of the baseline rotor in Figure 51.

The radial distribution of the stator incidence angle for the Phase III rotor is compared with the data match of the base line rotor in Figure 52. The lower stator incidence angle in the outer 40 percent of span for the Phase III design is largely the result of the assumed higher efficiency in the outer portion of the rotor.

Details of the Phase III rotor design are given in Section XV.

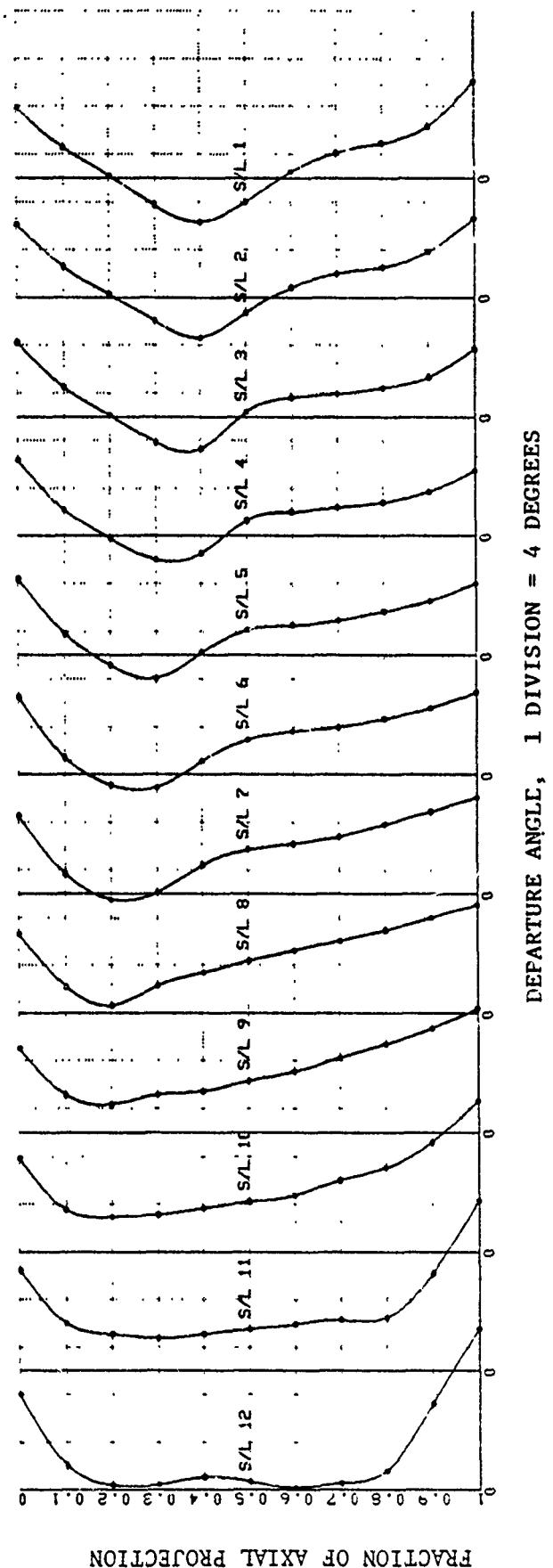


Figure 50. Phase III Rotor Intrablade Departure Angle Distribution

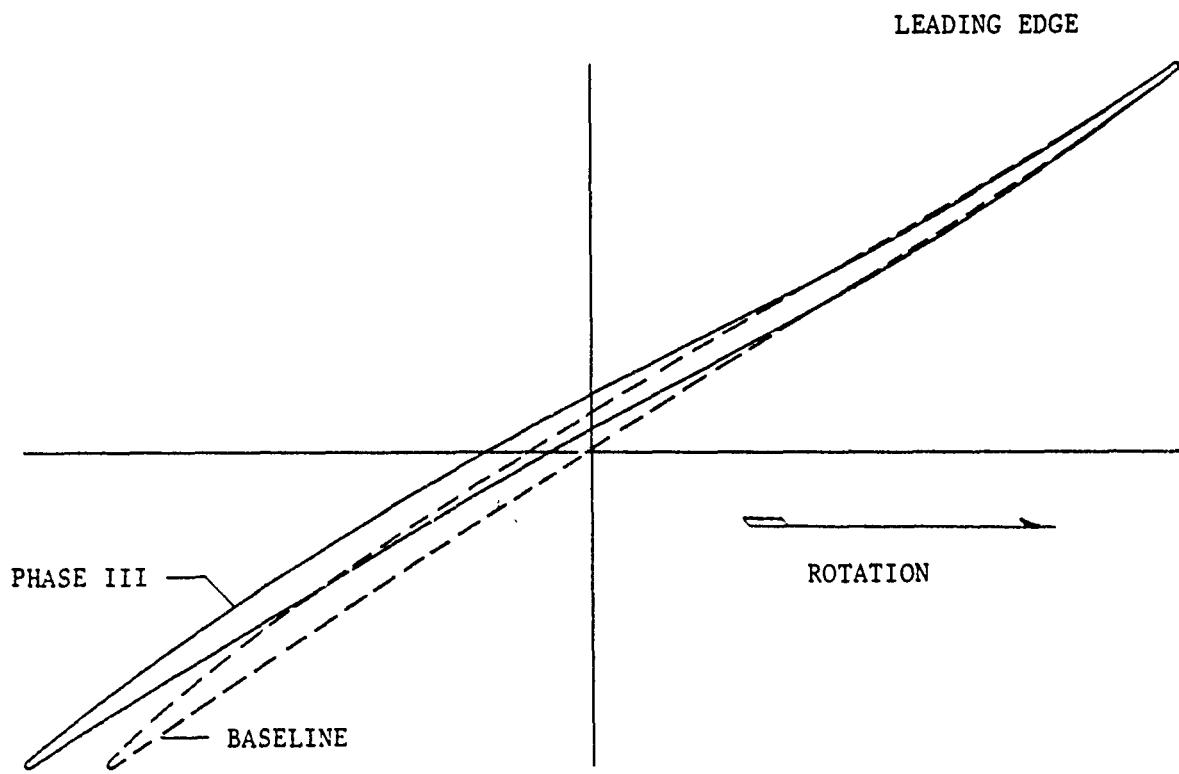


Figure 51. Phase III Rotor Streamsurface Tip Section Compared with Baseline Design

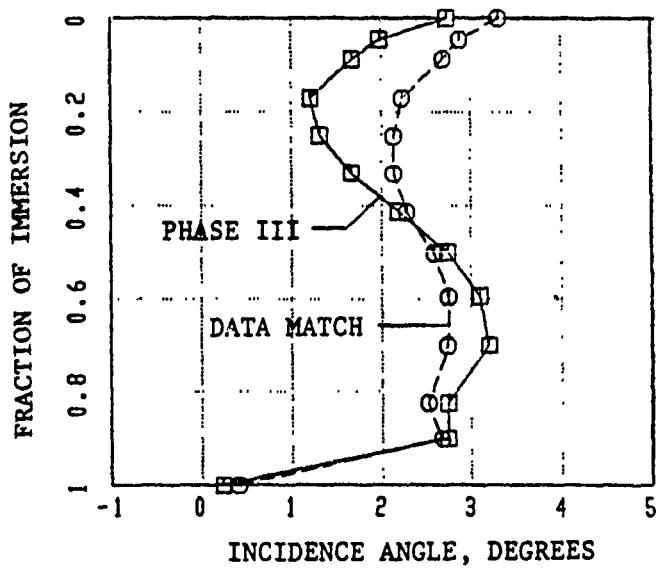


Figure 52. Phase III Stator Incidence Angle Compared With Data Match

SECTION XIV
DETAILS OF PHASE III ROTOR DESIGN

1. CIRCUMFERENTIAL AVERAGE FLOW SOLUTION

The following tabulation presents the detail results of the Phase III rotor circumferential average flow computation. Each page of the tabulation gives results for one calculation station. Figure 53 shows the calculation station locations within the gas flowpath. At each calculation station various aerodynamic parameters are given on each of thirteen calculation streamlines. Also given are several mass averaged station flow properties.

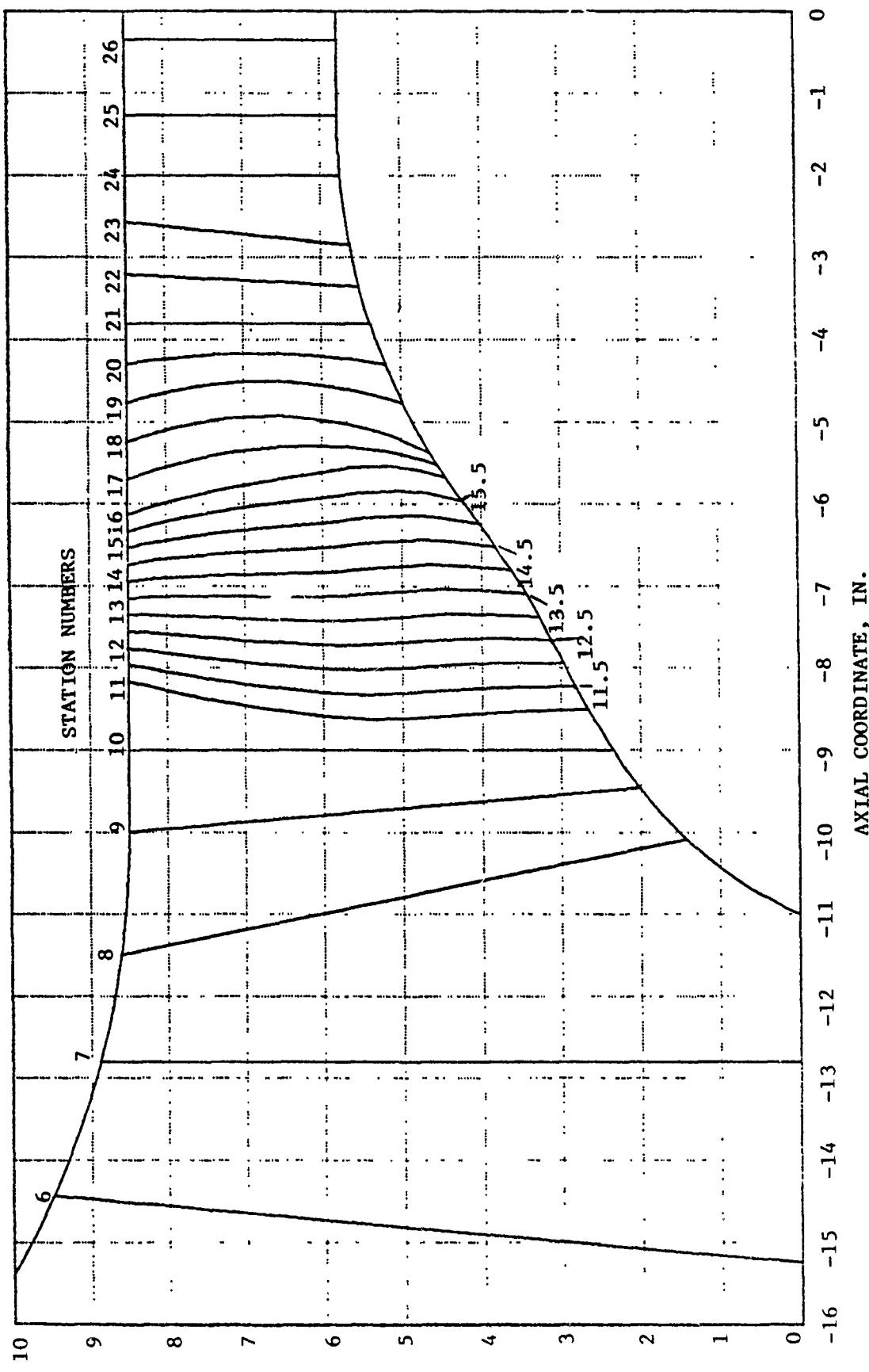


Figure 53. Compressor Flypath With Calculation Stations

INLET I= 1 STA= 5.000 AFLOW= 478 13 D+C=0. FREE
 WIF= 61.365 OPTX=DPP OPTY=FKEE ITYPE=O INBR=O D+H=0.
 PSIC Z R PHI CURV VM ALPHAM MM
 0. -18. 800 13. 207 -50. 10 0. 0.0831 150. 4 0. 0. 0.135
 0. 0.050 -18. 800 12. 564 -43. 54 0. 181. 0 0. 0. 0.163
 0. 100 -18. 800 12. 020 -40. 31 0. 195. 9 0. 0. 0.176
 0. 200 -18. 800 11. 027 -34. 70 0. 218. 6 0. 0. 0.196
 0. 300 -18. 800 10. 099 -29. 90 0. 237. 1 0. 0. 0.213
 0. 400 -18. 800 9. 193 -25. 65 0. 252. 4 0. 0. 0.227
 0. 500 -18. 800 8. 277 -21. 78 0. 265. 1 0. 0. 0.239
 0. 600 -18. 800 7. 319 -18. 17 0. 275. 9 0. 0. 0.248
 0. 700 -18. 800 6. 277 -14. 68 0. 284. 9 0. 0. 0.257
 0. 800 -18. 800 5. 083 -11. 19 0. 292. 5 0. 0. 0.264
 0. 900 -18. 800 3. 569 -7. 35 0. 298. 9 0. 0. 0.270
 0. 950 -18. 800 2. 516 -4. 92 0. 301. 6 0. 0. 0.272
 1. 000 -18. 800 0. 000 0. 0. 3C4. 0 0. 0. 0. 0.274

SL BLDBLK PS PT TT BETAM VREL MREL VABS MABS
 1 0.997 14.510 14.696 518.7 86.31 2335.4 2.095 150.4 0.135
 2 0.997 14.427 14.696 518.7 85.33 2224.5 1.997 181.0 0.163
 3 0.997 14.382 14.696 518.7 84.72 2130.2 1.913 195.9 0.176
 4 0.997 14.3C5 14.696 518.7 83.59 1958.1 1.760 218.6 0.196
 5 0.997 14.237 14.696 518.7 82.42 1797.9 1.617 237.1 0.213
 6 0.997 14.177 14.696 518.7 81.16 1641.9 1.477 252.4 0.227
 7 0.997 14.124 14.696 518.7 79.71 1484.6 1.337 265.1 0.239
 8 0.997 14.077 14.696 518.7 77.94 1320.6 1.190 275.9 0.248
 9 0.997 14.037 14.696 518.7 75.58 1143.7 1.031 284.9 0.257
 10 0.997 14.002 14.696 518.7 71.94 943.5 0.850 292.5 0.264
 11 0.997 13.972 14.696 518.7 64.61 697.2 0.629 298.9 0.270
 12 0.997 13.959 14.696 518.7 55.81 536.8 0.484 301.6 0.272
 13 0.997 13.947 14.696 518.7 0.00 304.0 0.274 304.0 0.274

STA 5.000 MASS AVERAGED PROPERTIES
 PT= 14.696 TT= 518.69 GAMMA= 1.4015 PT-RATE= 1.000 TT-RATE= 1.000
 RCU= 0. VM= 255.3 C7= 233.4 MM= 0.230 MABS= 0.230 MREL= 1.300

INLET	1= 2	STA= 6.000	AFLOW= 277.56	D+C=O.	FREE
WTF= 61.365	OPTX=DPP	OPTY=FREE	I TYPE=O INRR=O	ABC=O.	D+H-O.
PSIC Z R	PHI CURV VM CU ALPHAM MM				ABH=O.
0.050 -14.431 9.481	-24.96 -0.0952	514.6	0.	0.	0.471
0.050 -14.450 9.254	-24.10 -0.1028	507.6	0.	0.	0.464
0.100 -14.470 9.020	-22.95 -0.0955	501.1	0.	0.	0.458
0.200 -14.513 8.532	-20.65 -0.0825	489.4	0.	0.	0.447
0.300 -14.558 8.010	-18.38 -0.0712	478.4	0.	0.	0.436
0.400 -14.606 7.446	-16.13 -0.0614	467.8	0.	0.	0.426
0.500 -14.660 6.829	-13.87 -0.0529	457.2	0.	0.	0.416
0.600 -14.719 6.141	-11.59 -0.0455	446.4	0.	0.	0.406
0.700 -14.787 5.352	-9.23 -0.0390	434.9	0.	0.	0.395
0.800 -14.869 4.402	-6.73 -0.0330	422.1	0.	0.	0.383
0.900 -14.978 3.142	-4.03 -0.0257	406.9	0.	0.	0.369
0.950 -15.057 2.234	-2.57 -0.0190	398.1	0.	0.	0.361
1.000 -15.250 0.000	0.	387.5	0.	0.	0.351

SL	BUDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.997	12.623	14.696	518.7	72.90	1750.4	1.601	514.6	0.471
2	0.997	12.677	14.696	518.7	72.73	1710.1	1.564	507.6	0.464
3	0.997	12.726	14.696	518.7	72.53	1668.8	1.525	501.1	0.458
4	0.997	12.812	14.696	518.7	71.99	1583.2	1.445	489.4	0.447
5	0.997	12.892	14.696	518.7	71.30	1492.2	1.361	478.4	0.436
6	0.997	12.968	14.696	518.7	70.41	1394.8	1.271	467.8	0.426
7	0.997	13.042	14.696	518.7	69.22	1288.9	1.174	457.2	0.416
8	0.997	13.116	14.696	518.7	67.61	1172.0	1.066	446.4	0.406
9	0.997	13.193	14.696	518.7	65.27	1039.7	0.945	434.9	0.395
10	0.997	13.277	14.696	518.7	61.48	884.2	0.803	422.1	0.383
11	0.997	13.374	14.696	518.7	53.72	687.7	0.624	406.9	0.369
12	0.997	13.429	14.696	518.7	44.72	560.3	0.508	398.1	0.361
13	0.997	13.493	14.696	518.7	0.00	387.5	0.351	387.5	0.351

STA 6.000 MASS AVERAGED PROPERTIES
 PT= 14.696 TT= 518.69 GAMMA= 1.4016 PT-RAT= 1.000 TT-RAT= 1.000
 RCU= 0. VM= 455.5 CZ= 438.5 MM= 0.415 MABS= 0.415 MREL= 1.120

INLET STA= 7.000 AFLOW= 244.35 D+C=0. FREE D+H=0.
WTF= 61.365 M+IP= 27 OPTY-FREE ITYPE=0 INRR=0 ABC=0. ABH=0.
PSIC Z R PHI CURV VM CU ALPHAM MM
 0 -12.800 8.880 -15.47 -0.0952 625.2 Q. 0. 0.578
 0 0.050 -12.800 8.675 -14.65 -0.0872 617.8 O. 0. 0.571
 0 100 -12.800 8.464 -13.90 -0.0849 610.4 O. 0. 0.564
 0 200 -12.800 8.021 -12.40 -0.0755 595.1 O. 0. 0.549
 0.300 -12.800 7.546 -10.86 -0.0736 579.6 O. 0. 0.533
 0.400 -12.800 7.032 -9.28 -0.0680 563.9 O. 0. 0.518
 0.500 -12.800 6.468 -7.60 -0.0629 547.6 O. 0. 0.502
 0.600 -12.800 5.837 -5.79 -0.0587 530.4 O. 0. 0.486
 0.700 -12.800 5.112 -3.79 -0.0560 511.3 O. 0. 0.468
 0.800 -12.800 4.237 -1.46 -0.0558 488.4 O. 0. 0.446
 0.900 -12.800 3.064 1.52 -0.0633 455.8 O. 0. 0.415
 0.950 -12.800 2.206 3.55 -0.0759 428.6 O. 0. 0.390
 1.000 -12.800 0.000 O. 0. 0. 0.348

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.998	11.714	14.696	518.7	68.25	1687.2	1.560	625.2	0.578
2	0.998	11.779	14.696	518.7	68.02	1650.8	1.525	617.8	0.571
3	0.998	11.843	14.696	518.7	67.77	1613.5	1.490	610.4	0.564
4	0.998	11.974	14.696	518.7	67.20	1535.5	1.415	595.1	0.549
5	0.998	12.105	14.696	518.7	66.48	1452.4	1.337	579.6	0.533
6	0.998	12.235	14.696	518.7	65.56	1363.1	1.253	563.9	0.518
7	0.998	12.366	14.696	518.7	64.37	1266.0	1.162	547.6	0.502
8	0.998	12.502	14.696	518.7	62.76	1158.6	1.061	530.4	0.486
9	0.998	12.649	14.696	518.7	60.46	1036.9	0.948	511.3	0.468
10	0.998	12.819	14.696	518.7	56.85	893.0	0.815	488.4	0.446
11	0.998	13.051	14.696	518.7	49.87	707.2	0.644	455.8	0.415
12	0.998	13.234	14.696	518.7	42.25	579.0	0.526	428.6	0.390
13	0.998	13.517	14.696	518.7	0.00	383.6	0.348	383.6	0.348

STA 7.000 MASS AVERAGED PROPERTIES
 PT= 14.696 TT= 518.69 GAMMA= 1.4017 PT-RAT= 1.000 TT-RAT= 1.000
 RCU= 0. VM= 539.1 CZ= 532.1 MM= 0.495 MABS= 0.495 MREL= 1.109

INLET		I= 4	STA= 8.000	AFLOW= 224.07	D+C=0.	FREE
WTF=	61.365	OPTX=DPP	OPTY=FREE	ITYPE=0	INBR=0	D+H=0.
PSIC	Z R	PHI CURV	VM	CU	ALPHAM	ABH=0.
0	-11.499	8.608	-8.21	-0.0953	712.0	0.
0.050	-11.461	8.412	-7.49	-0.0954	703.2	0.
0.100	-11.421	8.211	-6.86	-0.0909	693.8	0.
0.200	-11.339	7.790	-5.58	-0.0816	675.7	0.
0.300	-11.250	7.341	-4.24	-0.0744	658.7	0.
0.400	-11.155	6.858	-2.77	-0.0695	642.1	0.
0.500	-11.052	6.333	-1.09	-0.0568	625.0	0.
0.600	-10.938	5.753	0.90	-0.0667	606.2	0.
0.700	-10.809	5.096	3.37	-0.0695	583.8	0.
0.800	-10.656	4.320	6.70	-0.0768	554.5	0.
0.900	-10.459	3.318	12.11	-0.0934	510.5	0.
0.950	-10.323	2.629	17.76	-0.1212	468.0	0.
1.000	-10.086	1.421	47.99	0.1910	433.9	0.

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.997	10.919	14.696	518.7	64.89	1677.6	1.567	712.0	0.665
2	0.997	11.003	14.696	518.7	64.65	1642.6	1.533	703.2	0.656
3	0.997	11.092	14.696	518.7	64.41	1506.5	1.497	693.8	0.647
4	0.997	11.259	14.696	518.7	63.82	1531.8	1.425	675.7	0.628
5	0.997	11.415	14.696	518.7	63.05	1453.3	1.349	658.7	0.611
6	0.997	11.564	14.696	518.7	62.05	1370.1	1.269	642.1	0.595
7	0.997	11.716	14.696	518.7	60.78	1280.6	1.184	625.0	0.578
8	0.997	11.879	14.696	518.7	59.16	1182.5	1.091	606.2	0.559
9	0.997	12.070	14.696	518.7	57.01	1072.2	0.987	583.8	0.538
10	0.997	12.311	14.696	518.7	53.97	942.6	0.865	554.5	0.509
11	0.997	12.655	14.696	518.7	48.91	776.8	0.710	510.5	0.467
12	0.997	12.966	14.696	518.7	44.75	659.0	0.601	468.0	0.427
13	0.997	13.200	14.696	518.7	30.03	501.2	0.456	433.9	0.394

STA 8.000 MASS AVERAGED PROPERTIES

PT= 14.696 TT= 518.69 GAMMA= 1.4017 PT-RAT= 1.000 TT-RAT= 1.000
 RCU= 0 VM= 612.1 C2= 604.7 MM= 0.566 MABS= 0.566 MREL= 1.140

INLET	WTF =	I = 5	OPTX-DPP	OPTY-FRFE	LTYPE=0	MTIP = 53	STA= 9 000	AFLOW= 211.87	D+C=0.	FRFE
0	-9.999	8	500	0	0	758.9	0	0	0.713	
0	-9.984	8	315	-1.10	-0.0543	751.9	0	0	0.706	
0	0.050	-9	125	-0.88	-0.0526	744.0	0	0	0.698	
0	100	-9	968	-0.25	-0.0510	730.2	0	0	0.684	
0	200	-9	935	0.62	-0.0514	716.0	0	0	0.669	
0	300	-9	900	1.78	-0.0535	700.5	0	0	0.653	
0	400	-9	862	3.30	-0.0577	682.6	0	0	0.635	
0	500	-9	821	6.359	-0.0623	660.8	0	0	0.613	
0	600	-9	776	5.816	5.23	-0.0633	650.0	0	0	
0	700	-9	725	5.201	7.75	-0.0711	633.0	0	0.586	
0	800	-9	665	4.475	11.16	-0.0788	596.4	0	0.550	
0	900	-9	587	3.542	16.60	-0.0812	546.2	0	0.501	
0	950	-9	536	2.919	21.65	-0.0910	515.3	0	0.472	
1	0.996	-9.460	2.011	38.65	0.1881	511.4	0	0	0.468	
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS	
1	0.996	10.466	14.696	518.7	63.16	1681.0	1.580	758.9	0.713	
2	0.996	10.534	14.696	518.7	62.87	1648.8	1.548	751.9	0.706	
3	0.996	10.612	14.696	518.7	62.58	1615.3	1.515	744.0	0.698	
4	0.996	10.745	14.696	518.7	61.83	1546.9	1.448	730.2	0.684	
5	0.996	10.881	14.696	518.7	60.95	1474.6	1.378	716.0	0.669	
6	0.996	11.028	14.696	518.7	59.91	1397.4	1.303	700.5	0.653	
7	0.996	11.196	14.696	518.7	58.69	1313.5	1.223	682.6	0.635	
8	0.996	11.397	14.696	518.7	57.23	1220.7	1.133	660.8	0.613	
9	0.996	11.646	14.696	518.7	55.41	1115.0	1.032	633.0	0.586	
10	0.996	11.963	14.696	518.7	52.94	989.6	0.912	596.4	0.550	
11	0.996	12.377	14.696	518.7	48.85	830.1	0.762	546.2	0.501	
12	0.996	12.618	14.696	518.7	44.99	728.7	0.667	515.3	0.472	
13	0.996	12.648	14.696	518.7	34.76	622.4	0.569	511.4	0.468	

STA 9.000 MASS AVERAGED PROPERTIES
 PT= 14.696 TT= 518.69 GAMMA= 1.4018 PT-RAT= 1.000 TT-RAT= 1.000
 RCU= 0 VM= 663.8 CZ= 655.4 MM= 0.617 MABS= 0.617 MREL= 1.178

INLET		I = 6	STA= 10,000	AFLOW= 204.13	D+C=O.	FREE	D+H-O.	
W1F =	61.365	OPTX=DPP	OPTY=FREE	ITYPE=O	INBR=O	ABC=O.	ABH=O.	
PSIC	Z	R	PHI	CURV	VM	CU	ALPHAM	MM
0.	-9.000	8.500	0.	0.	774.8	0.	0.	0.730
0.	-9.000	8.317	0.34	0.0031	775.1	0.	0.	0.730
0.100	-9.000	8.130	0.65	-0.0024	775.0	0.	0.	0.730
0.200	-9.000	7.743	1.46	-0.0129	772.3	0.	0.	0.727
0.300	-9.000	7.333	2.52	-0.0224	765.8	0.	0.	0.720
0.400	-9.000	6.897	3.91	-0.0326	754.8	0.	0.	0.709
0.500	-9.000	6.425	5.65	-0.0422	738.5	0.	0.	0.692
0.600	-9.000	5.906	7.94	-0.0578	714.6	0.	0.	0.668
0.700	-9.000	5.320	10.76	-0.0720	680.6	0.	0.	0.633
0.800	-9.000	4.625	14.37	-0.0853	634.6	0.	0.	0.587
0.900	-9.000	3.734	19.55	-0.0855	575.6	0.	0.	0.530
0.950	-9.000	3.141	23.74	-0.0851	535.8	0.	0.	0.491
1.000	-9.000	2.340	32.46	0.1922	543.6	0.	0.	0.499
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	MABS
1	0.994	10.309	14.696	518.7	62.68	1688.3	1.590	774.8
2	0.994	10.306	14.696	518.7	62.16	1659.8	1.563	775.1
3	0.994	10.306	14.696	518.7	61.62	1630.7	1.536	775.0
4	0.994	10.333	14.696	518.7	60.52	1569.5	1.478	772.3
5	0.994	10.398	14.696	518.7	59.39	1503.7	1.414	765.8
6	0.994	10.506	14.696	518.7	58.19	1432.1	1.345	754.8
7	0.994	10.665	14.696	518.7	56.92	1353.2	1.268	738.5
8	0.994	10.895	14.696	518.7	55.57	1263.7	1.181	714.6
9	0.994	11.214	14.696	518.7	54.06	1159.5	1.079	680.6
10	0.994	11.631	14.696	518.7	52.13	1033.9	0.957	634.6
11	0.994	12.138	14.696	518.7	48.86	874.9	0.805	575.6
12	0.994	12.459	14.696	518.7	45.97	771.0	0.707	535.8
13	0.994	12.398	14.696	518.7	37.22	682.6	0.626	543.6

STA 10,000 MASS AVERAGED PROPERTIES

PT= 14.696 TT= 518.69 GAMMA= 1.4018 PT-RATE= 1.000 TT RATE= 1.000
RCU= 0. VM= 705.9 CZ= 694.5 MM= 0.660 MABS= 0.660 MREL= 1.217

ROTOR1 1= 7 STA= 11.000 LE ROTOR
 WTF= 61.365 M1IP= 79 AFLOW= 197.36 0*C=O.
 OPTX=DPP OPTY=FREE ITYPE=4 INBR=3 0*H=O.
 PSIC Z R PHI CURV VM CU ALPHAM
 0 -8.166 8.500 0. O. 835 4 0. 0.
 0.050 -8.204 8.322 0.40 -0.0055 837 3 0. 0.
 0.100 -8.242 8.140 0.78 -0.0035 838 5 0. 0.
 0.200 -8.322 7.763 1.77 -0.0033 839 3 0. 0.
 0.300 -8.396 7.364 3.15 -0.0140 834 3 0. 0.
 0.400 -8.465 6.938 4.88 -0.0303 820 2 0. 0.
 0.500 -8.531 6.477 6.88 -0.0486 794 6 0. 0.
 0.600 -8.592 5.968 9.19 -0.0480 760 3 0. 0.
 0.700 -8.624 5.396 12.24 -0.0622 718 8 0. 0.
 0.800 -8.604 4.734 16.11 -0.0630 669 3 0. 0.
 0.900 -8.548 3.904 21.67 -0.0678 605 3 0. 0.
 0.950 -8.526 3.362 25.96 -0.0633 560 7 0. 0.
 1.000 -8.507 2.653 31.20 -0.1471 551 6 0. 0.
 SL BLDBLK PS PT TT BETAM VRFL MREL VABS MARS
 1 0.990 9.697 14.696 518 7 60.89 1716.9 1.631 835.4 0.794
 2 0.990 9.677 14.696 518 7 60.31 1690.5 1.607 837.3 0.796
 3 0.990 9.664 14.696 518 7 59.73 1663.3 1.581 838.5 0.797
 4 0.990 9.657 14.696 518 7 58.51 1606.6 1.527 839.3 0.798
 5 0.990 9.708 14.696 518 7 57.30 1544.2 1.467 834.3 0.792
 6 0.990 9.852 14.696 518 7 56.18 1473.7 1.397 820.2 0.777
 7 0.990 10.111 14.696 518 7 55.19 1392.0 1.315 794.6 0.750
 8 0.990 10.452 14.696 518 7 54.17 1299.0 1.221 760.3 0.715
 9 0.990 10.854 14.696 518 7 52.95 1193.1 1.115 718.8 0.672
 10 0.990 11.319 14.696 518 7 51.30 1070.4 0.995 669.3 0.622
 11 0.990 11.887 14.696 518 7 48.70 917.1 0.846 605.3 0.559
 12 0.990 12.260 14.696 518 7 46.62 816.3 0.750 560.7 0.515
 13 0.990 12.334 14.696 518 7 40.33 723.5 0.664 551.6 0.506

STA= 11.000 MASS AVERAGED PROPERTIES
 PT= 14.696 TT= 518.69 GAMMA= 1.4018 PT-RAT= 1.000 TT-RAT= 1.000
 RCU= 0. VM= 756.5 CZ= 742.2 MM= 0.712 MABS= 0.712 MREL= 1.261

ROTOR 1		STA 11.500 IN ROTOR							
WTF =	365	I = 8	WTIP = 92	AFLOW = 180.99	D+C=0.				
PSIC	Z	OPTX-TT	OPTY-PT	ITYPE=5	INBR=3	ABC=0.			
PHI	R	PHI	CURV	VM	CU	ALPHAM	MM		
0	-7.963	8.500	0	0.	848.4	45.6	3.08	0.798	D+H=0.
0	-7.991	8.324	0.30	0.0205	851.7	41.3	2.77	0.803	ABH=0.
0	-8.020	8.143	0.68	0.0182	859.3	39.2	2.62	0.811	
0	-8.083	7.770	1.73	0.0098	875.2	37.6	2.46	0.829	
0	-8.143	7.378	3.29	-0.0050	887.0	39.6	2.56	0.842	
0	-8.198	6.962	5.22	-0.0149	886.2	42.0	2.71	0.841	
0	-8.251	6.512	7.38	-0.0134	874.7	44.9	2.94	0.828	
0	-8.301	6.017	9.83	-0.0279	846.7	47.5	3.21	0.799	
0	-8.324	5.464	12.83	-0.0052	813.8	56.0	3.94	0.764	
0	-8.299	4.824	16.59	0.0102	765.9	52.2	3.90	0.715	
0	-8.247	4.027	22.17	0.0134	714.1	61.7	4.93	0.662	
0	-8.231	3.508	25.98	0.0615	678.9	65.0	5.47	0.627	
0	-8.224	2.817	29.07	0.0800	654.0	68.9	6.01	0.603	
SL	BLOBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.955	10.182	15.513	530.1	59.75	1683.7	1.584	849.6	0.799
2	0.955	10.102	15.460	528.8	59.18	1662.4	1.566	852.7	0.804
3	0.954	10.003	15.440	528.1	58.42	1640.8	1.549	860.2	0.812
4	0.952	9.820	15.432	527.3	56.72	1595.1	1.511	876.0	0.830
5	0.948	9.718	15.474	527.3	54.91	1542.8	1.464	887.8	0.843
6	0.942	9.745	15.506	527.3	53.25	1480.9	1.405	887.2	0.842
7	0.934	9.881	15.525	527.3	51.62	1408.7	1.334	875.8	0.830
8	0.925	10.175	15.518	527.1	50.15	1321.3	1.246	848.1	0.800
9	0.914	10.573	15.587	527.7	48.14	1219.5	1.144	815.8	0.765
10	0.899	10.954	15.430	526.1	46.22	1106.9	1.033	767.7	0.717
11	0.881	11.459	15.419	526.0	42.26	964.9	0.895	716.8	0.665
12	0.848	11.745	15.354	525.4	39.22	876.3	0.810	682.0	0.630
13	0.810	11.888	15.248	524.4	33.22	781.7	0.721	657.6	0.607

STA 11.500 MASS AVERAGED PROPERTIES
 PT = 15.471 TT = 527.14 GAMMA = 1.4018 PT-RAT = 1.053 TT-RAT = 1.016
 RCU = 287.0 VM = 827.0 C2 = 809.8 MM = 0.779 MABS = 0.780 MREL = 1.274

ROTOR 1		I = 9	STA = 12,000	AFLLOW = 170,36	D+C=0.	IN ROTOR
WTF =	61.365	OPTX=TT	MTIP=105	ITYPE=5	INBR=3	D+H=0.
PSIC	2	R	OPTY=PT	CURV	CU	ABC=0.
0.	-7.759	8.500	0.	0	96.7	0.755
0.	-7.778	8.324	-0.02	0.0326	822.8	6.54
0.	-7.798	8.145	0.33	0.0372	831.3	92.2
0.	-7.844	7.777	1.53	0.0187	852.1	92.2
0.	-7.889	7.393	3.22	0.141	871.3	97.0
0.	-7.931	6.987	5.29	0.0057	882.0	103.1
0.	-7.972	6.549	7.51	-0.0023	881.4	110.0
0.	-8.010	6.068	9.98	0.0103	875.1	118.7
0.	-8.024	5.532	12.79	0.0098	856.3	124.7
0.	-7.995	4.915	16.49	0.0006	820.5	131.4
0.	-7.946	4.149	21.85	0.0209	771.3	142.5
0.	-7.936	3.649	25.21	0.0211	736.4	147.4
0.	-7.941	2.971	28.21	0.0136	688.0	149.9
1.000	-7.94				12.29	0.633
SL	BLDBLK	PS	PT	TT	BETAM	VREL
1	0.932	11.217	16.462	542.9	59.79	1623.8
2	0.931	11.160	16.479	541.8	59.10	1602.1
3	0.930	11.052	16.482	540.8	58.29	1581.3
4	0.926	10.861	16.544	539.8	56.35	1537.9
5	0.920	10.709	16.655	539.8	54.19	1489.2
6	0.910	10.639	16.747	539.9	52.02	1433.3
7	0.896	10.668	16.799	539.9	49.87	1367.6
8	0.880	10.749	16.832	539.9	47.42	1293.2
9	0.863	10.900	16.760	539.0	44.84	1207.7
10	0.842	11.197	16.633	537.7	41.89	1102.2
11	0.820	11.583	16.461	536.1	37.39	970.8
12	0.774	11.801	16.281	534.5	34.00	888.3
13	0.730	12.048	15.985	531.8	28.55	783.2

STA 12,000 MASS AVERAGED PROPERTIES
 PT = 16.616 TT = 539.05 GAMMA = 1.4018 PT-RAT = 1.131 TT-RAT = 1.039
 RCU = 691.5 VM = 839.7 CZ = 821.8 MM = 0.782 MAB = 0.790 MREL = 1.229

IN ROTOR									
	STA=	12.500	AFLOW=	161.72	D+C=O.	D+H=O.	ABH=O.	MM	
WTF=	61.365	OPTX=TT	I TYPE=PT	1NBR=3	ABC=O.				
PS1C	2	R	PHI	CURV	VM	CU	ALPHAM	MM	
0.	-7.556	8.500	0	0	778.4	155.1	11.27	0.706	
0.	-7.565	8.323	-0.26	0.00677	789.7	159.6	11.43	0.717	
0.	-7.576	8.146	-0.06	0.0240	796.2	159.8	11.35	0.725	
0.	-7.606	7.783	1.09	0.0467	817.7	161.5	11.18	0.747	
0.	-7.636	7.407	2.89	0.0320	840.3	168.4	11.33	0.771	
0.	-7.664	7.011	4.99	0.0333	854.5	176.9	11.70	0.786	
0.	-7.692	6.585	7.30	0.0281	862.5	185.2	12.12	0.795	
0.	-7.720	6.119	9.74	0.0180	865.5	196.0	12.76	0.798	
0.	-7.725	5.600	12.59	0.0135	858.0	208.7	13.67	0.792	
0.	-7.690	5.005	16.27	0.0237	840.6	224.7	14.97	0.775	
0.	-7.646	4.268	21.44	0.0242	805.9	239.6	16.56	0.742	
0.950	-7.641	3.786	24.93	0.0088	773.5	243.2	17.45	0.712	
1.000	-7.658	3.123	28.5!	-0.0461	712.6	236.1	18.33	0.654	

SL	BLDBLK	PS	PT	IT	BETAM	VREL	MREL	VABS	MABS
1	0.907	12.445	17.585	557.5	59.94	1553.9	1.410	793.7	0.720
2	0.907	12.453	17.789	557.8	58.90	1528.9	1.389	805.6	0.732
3	0.906	12.432	17.875	557.0	58.07	1505.5	1.370	812.1	0.739
4	0.901	12.276	18.034	555.7	55.99	1462.0	1.336	833.5	0.762
5	0.694	12.109	18.213	555.4	53.57	1415.2	1.298	857.0	0.786
6	0.881	12.002	18.344	555.2	51.14	1361.8	1.252	872.6	0.802
7	0.865	11.897	18.375	554.6	48.56	1303.2	1.201	882.2	0.813
8	0.847	11.820	18.368	554.0	45.60	1237.0	1.141	887.4	0.819
9	0.828	11.828	18.309	553.1	42.25	1159.2	1.070	883.0	0.815
10	0.803	11.892	18.182	551.8	38.07	1067.8	0.985	870.1	0.803
11	0.777	11.996	17.841	548.8	32.51	955.6	0.880	840.7	0.774
12	0.726	12.080	17.487	545.8	28.78	882.6	0.812	810.9	0.746
13	0.680	12.279	16.873	540.4	23.85	779.1	0.715	750.7	0.689

STA 12.500 MASS AVERAGED PROPERTIES
 $P_1 = 18.084 \quad TT = 553.54 \quad \Gamma = 1.4017 \quad P_1 - RAT = 1.231 \quad TT - RAT = 1.067$
 $P_{CL} = 1184.0 \quad VM = 829.5 \quad C_2 = 811.7 \quad MM = 0.762 \quad MABS = 0.783 \quad MREL = 1.160$

ROTOR 1		I=11		STA= 13 000		AFLOW= 154.98		D+C=0.		IN ROTOR	
WTF=	365	OPTX=TT	OPTY=PT	WTIP=131	ITYPE=5	INBR=3	CU	ABC=0.	D+H=0.	ABH=0.	
PSIC	Z	R	PHI	CURV	VM	MM	ALPHAM	MM			
0.	-7	352	8.500	0	0	748.8	205.9	15.38	0	670	
0.050	-7	352	8.322	-0.57	0.0450	767.9	212.8	15.49	0	688	
0.100	-7	354	8.145	-0.52	0.0480	783.4	215.7	15.40	0	704	
0.200	-7	367	7.786	0.58	0.0270	810.0	220.9	15.25	0	731	
0.300	-7	382	7.418	0.42	0.0323	830.6	227.2	15.30	0	753	
0.400	-7	397	7.033	4.57	0.0211	845.7	236.8	15.64	0	768	
0.500	-7	413	6.620	6.86	0.0261	853.1	245.9	16.08	0	777	
0.600	-7	429	6.168	9.38	0.0249	864.8	260.6	16.77	0	790	
0.700	-7	425	5.666	12.32	0.0169	868.7	277.7	17.73	0	796	
0.800	-7	386	5.092	15.96	0.0105	861.5	304.3	19.45	0	790	
0.900	-7	344	4.385	21.12	0.0098	833.0	327.8	21.48	0	764	
0.950	-7	346	3.923	24.99	-0.0155	804.2	328.3	22.21	0	738	
1.000	-7	375	3.281	29.92	-0.1052	736.3	319.0	23.42	0	674	
SL	BUDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS		
1	0.885	13.471	18.599	570.2	59.94	1495.1	1.337	776.6	0.694		
2	0.886	13.451	18.899	570.8	58.56	1472.0	1.319	796.8	0.714		
3	0.885	13.386	19.083	570.4	57.33	1451.2	1.303	812.6	0.730		
4	0.882	13.243	19.377	569.3	54.91	1409.2	1.272	839.6	0.758		
5	0.876	13.089	19.574	568.3	52.49	1364.0	1.236	861.1	0.780		
6	0.864	12.961	19.726	567.7	49.90	1313.0	1.193	878.2	0.798		
7	0.845	12.829	19.737	566.6	47.23	1256.4	1.145	887.8	0.809		
8	0.825	12.627	19.751	566.0	43.75	1197.2	1.094	903.2	0.825		
9	0.803	12.463	19.693	565.0	39.74	1129.7	1.035	912.0	0.835		
10	0.775	12.398	19.639	564.3	34.60	1046.6	0.959	913.7	0.838		
11	0.748	12.354	19.241	561.0	28.17	944.9	0.867	895.2	0.821		
12	0.700	12.296	18.696	556.6	24.36	882.8	0.810	868.6	0.797		
13	0.653	12.463	17.850	549.5	19.44	780.9	0.715	802.5	0.735		

STA 13 000 MASS AVERAGED PROPERTIES

PT= 19.412 TT= 565.94 GAMMA= 1.4016 PT-RAT= 1.321 TT-RAT= 1.091
 RCU= 1605.3 VM= 830.4 CZ= 812.6 MM= 0.755 MABS= 0.792 MREL= 1.114

IN ROTOR									
	STA= 13.500	MTIP=1.44	AFLOW= 150.80	D+C=0.	D+H=0.	ABC=0.	ABH=0.	MM	ALPHAM
WTF=	61.365	OPTX=TT	OPTY=PT	ITYPE=5	INBR=3				
PSIC	Z	R	PHI	CURV	VM	CU			
0.	-7.148	8.500	0.	0.	701.6	243.5	19.14	0.619	
0.050	-7.138	8.319	-0.76	-0.0142	730.7	260.7	19.63	0.646	
0.100	-7.132	8.142	-0.79	-0.0052	749.1	270.5	19.86	0.663	
0.200	-7.128	7.788	0.29	0.0162	778.0	280.2	19.81	0.692	
0.300	-7.129	7.428	2.02	0.0227	802.5	289.2	19.82	0.717	
0.400	-7.130	7.054	4.08	0.0431	820.1	297.8	19.96	0.735	
0.500	-7.133	6.653	6.40	0.0319	833.1	307.6	20.26	0.750	
0.600	-7.138	6.215	8.99	0.0207	847.9	324.8	20.96	0.766	
0.700	-7.125	5.731	12.00	0.0193	856.8	344.6	21.91	0.777	
0.800	-7.081	5.179	15.88	-0.0015	857.8	375.4	23.63	0.780	
0.900	-7.043	4.501	21.16	-0.0143	838.1	416.0	26.40	0.764	
0.950	-7.051	4.062	25.40	-0.0281	814.3	415.0	27.01	0.743	
1.000	-7.092	3.450	31.86	-0.0937	762.8	402.8	27.84	0.697	
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.877	14.512	19.371	579.6	60.82	1439.1	1.270	742.7	0.656
2	0.879	14.551	19.934	582.5	58.82	1411.3	1.247	775.8	0.686
3	0.880	14.575	20.312	583.5	57.29	1386.1	1.227	796.4	0.705
4	0.878	14.507	20.785	582.9	54.58	1342.6	1.194	827.0	0.735
5	0.871	14.351	21.082	581.9	51.85	1299.1	1.160	853.0	0.762
6	0.857	14.156	21.213	580.5	49.11	1252.7	1.123	872.5	0.782
7	0.837	13.915	21.204	578.9	46.12	1202.0	1.082	888.1	0.799
8	0.816	13.635	21.218	578.1	42.32	1146.6	1.036	908.0	0.820
9	0.793	13.348	21.137	576.8	37.89	1085.7	0.984	923.5	0.837
10	0.766	13.109	21.063	575.9	32.12	1012.9	0.921	936.4	0.851
11	0.737	12.925	20.793	573.8	24.29	919.5	0.838	935.7	0.852
12	0.699	12.712	20.072	568.3	20.33	868.4	0.793	914.0	0.834
13	0.649	12.589	18.981	559.6	15.11	790.1	0.722	862.5	0.788

STA= 13.500 MASS AVERAGED PROPERTIES
 PT= 20.807 TT= 578.32 GAMMA= 1.4015 PT-RATE= 1.416 TT-RATE= 1.115
 RCU= 2026.3 VM= 812.5 CZ= 794.5 MM=0.731 MABS=0.788 MREL=1.057

ROTOR 1		STA = 14.000	STA = 14.000	AFLOW =	147.45	D+C=O.	D+H=O.
WTF =	WTIP =	MIIP = 157	OPTX=TT	OPTY=PT	I TYPE=5	INRR=3	ABC=O.
PSIC	Z	R	PHI	CURV	VM	CU	ABH=O.
0.	-6.945	8.500	0.	0.	661.1	289.2	23.62
0.	-6.925	8.317	-0.60	-0.0117	695.9	307.4	23.83
0.050	-6.910	8.139	-0.65	-0.0166	724.5	323.3	24.05
0.100	-6.890	7.789	0.21	-0.0044	759.9	337.0	23.92
0.200	-6.875	7.436	1.78	0.0097	785.7	348.8	23.94
0.300	-6.863	7.071	3.68	0.0088	799.8	360.6	24.27
0.400	-6.854	6.683	5.92	0.0270	808.1	372.4	24.74
0.500	-6.847	6.260	8.58	0.0281	824.9	389.9	25.30
0.600	-6.825	5.794	11.74	0.0100	835.9	409.5	26.10
0.700	-6.777	5.266	15.94	-0.0045	843.5	444.1	27.77
0.800	-6.742	4.619	21.61	-0.0344	835.4	490.9	30.44
0.900	-6.756	4.204	26.15	-0.0524	818.8	494.8	31.15
0.950	-6.809	3.632	33.48	-0.0749	788.5	490.3	31.88
1.000							0.720

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.873	15.576	20.333	591.0	61.37	1379.6	1.202	721.6	0.629
2	0.876	15.601	20.975	593.9	59.05	1353.0	1.181	760.8	0.664
3	0.877	15.612	21.541	596.1	56.94	1328.0	1.161	793.4	0.694
4	0.874	15.559	22.193	595.9	53.78	1286.0	1.130	831.2	0.731
5	0.866	15.432	22.608	595.0	50.80	1243.2	1.098	859.6	0.759
6	0.852	15.308	22.827	593.7	47.97	1194.6	1.058	877.3	0.777
7	0.833	15.109	22.836	591.9	44.96	1142.0	1.015	889.8	0.791
8	0.813	14.731	22.800	590.5	40.91	1091.5	0.975	912.4	0.815
9	0.792	14.328	22.642	588.5	36.25	1036.5	0.930	930.9	0.835
10	0.768	13.926	22.559	587.5	29.90	973.1	0.877	953.3	0.859
11	0.737	13.490	22.274	585.4	21.21	896.1	0.811	969.0	0.877
12	0.714	13.136	21.506	579.9	16.78	855.2	0.777	956.7	0.869
13	0.662	12.705	20.328	571.1	10.81	802.8	0.733	928.6	0.847

STA = 14.000 MASS AVERAGED PROPERTIES
 PT = 22.266 TT = 590.66 GAMMA = 1.4014 PT-RAT = 1.515 TT-RAT = 1.139
 RCU = 2446.1 VM = 794.7 CZ = 776.3 MM = 0.708 MABS = 0.788 MREL = 1.003

ROTOR1		I=14	STA= 14.500	M1IP= 170	AFLOW= 145 80	D+C-O.	IN ROTOR
WTF=	61.365	OPTX=TT	OPTY=PT	ITYPE=5	INBR=3	ABC-O,	D-H=0,
PSIC	Z R	PHI	CURV	VM	CU	ALPHAM	ABH=0,
0.	-6.741 8.500	C.	0.	616.6	355.6	29.97	0.529
0.050	-6.712 8.315	-0.41	-0.0198	656.5	372.6	29.58	0.565
0.100	-6.688 8.137	-0.40	-0.0238	686.2	385.8	29.34	0.592
0.200	-6.651 7.790	0.31	-0.0116	729.3	401.6	28.84	0.632
0.300	-6.622 7.444	1.69	0.0033	754.9	412.0	28.63	0.658
0.400	-6.596 7.088	3.50	0.0158	770.6	423.1	28.77	0.675
0.500	-6.574 6.711	5.67	0.0040	778.9	435.7	29.22	0.685
0.600	-6.556 6.303	8.31	0.0040	790.5	454.2	29.88	0.698
0.700	-6.525 5.856	11.67	-0.0019	799.1	473.2	30.63	0.709
0.800	-6.473 5.353	16.17	-0.0208	808.6	506.2	32.05	0.721
0.900	-6.441 4.740	22.56	-0.0678	817.0	561.6	34.50	0.734
0.950	-6.461 4.352	27.47	-0.0874	810.1	580.4	35.62	0.731
1.000	-6.526 3.824	34.87	-0.0668	799.2	588.4	36.37	0.727

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	MABS
1	0.874	16.935	21.783	607.6	61.68	1299.9	1.115	711.9
2	0.877	16.936	22.483	609.8	59.05	1276.5	1.098	754.9
3	0.878	16.932	23.053	611.0	56.84	1254.4	1.082	787.2
4	0.875	16.868	23.871	610.7	53.15	1216.0	1.054	832.5
5	0.869	16.740	24.307	608.9	50.06	1175.9	1.025	860.0
6	0.857	16.570	24.524	606.9	47.05	1130.9	0.990	879.1
7	0.840	16.336	24.527	604.7	43.86	1080.3	0.949	892.5
8	0.824	15.969	24.463	602.9	39.78	1028.6	0.908	911.7
9	0.808	15.508	24.220	600.2	35.03	975.9	0.866	928.7
10	0.790	14.965	24.032	598.4	28.47	919.9	0.820	954.0
11	0.759	14.227	23.828	597.0	18.60	862.0	0.774	991.4
12	0.750	13.725	23.210	593.0	13.04	831.5	0.751	996.6
13	0.699	12.987	22.032	584.9	6.17	803.8	0.731	992.4

STA 14.500 MASS AVERAGED PROPERTIES

PT= 23.883 TT= 603.82 GAMMA=1.4012 PT-RAT= 1.625 TT-RAT= 1.164
 RCU= 2894.4 VM= 764.6 CZ= 745.6 MM=0.673 MABS=0.783 MREL=0.939

ROTDR1 **I=15** **OPTX=TT** **STA= 15.000** **AFLOW= 145.75** **D+C=0.**
WTF= 61 365 **MTIP= 18.3** **OPTY=PT** **TYPE=5 INBR=3** **D+H=0.**
PSIC Z R **PHI** **CURV** **VM** **ALPHAM** **MM**
 0. -6.538 8.500 0. 0. 569.6 451.1 38.37 0.479
 0. -6.499 8.314 0. 0.35 0.0088 614.2 460.4 36.85 0.519
 0. 0.050 -6.466 8.136 -0.33 0.0128 645.6 464.6 35.74 0.548
 0. 100 -6.466 8.136 -0.33 0.0128 645.6 464.6 35.74 0.548
 0. 200 -6.412 7.791 0.37 0.0028 692.7 469.3 34.12 0.593
 0. 300 -6.368 7.451 1.68 -0.0011 719.3 475.2 33.45 0.619
 0. 400 -6.329 7.104 3.42 -0.0053 731.1 483.1 33.46 0.632
 0. 500 -6.295 6.738 5.66 -0.0031 738.3 496.6 33.93 0.641
 0. 600 -6.265 6.345 8.38 -0.0120 746.4 512.9 34.50 0.651
 0. 700 -6.225 5.918 11.90 -0.0244 754.0 535.6 35.39 0.661
 0. 800 -6.168 5.442 16.69 -0.0364 763.2 570.5 36.78 0.673
 0. 900 -6.139 4.869 23.83 -0.0666 788.5 635.5 38.87 0.702
 0. 950 -6.166 4.511 28.90 -0.0607 800.3 670.2 39.94 0.718
 1.000 -6.243 4.026 36.12 -0.0570 807.1 710.3 41.35 0.731

SL BLDBLK PS PT TT BETAM VREL MREL VABS MABS
 1 0.884 18.620 23.968 631.4 61.50 1193.6 1.004 726.6 0.611
 2 0.887 18.548 24.616 631.2 58.61 1179.3 0.997 767.5 0.649
 3 0.888 18.459 25.056 629.8 56.39 1166.1 0.990 795.3 0.675
 4 0.886 18.270 25.716 626.2 52.59 1140.1 0.976 836.7 0.716
 5 0.881 18.092 26.092 622.8 49.42 1105.7 0.952 862.1 0.742
 6 0.873 17.925 26.238 619.6 46.51 1062.2 0.919 876.3 0.758
 7 0.861 17.679 26.247 617.1 43.17 1012.2 0.879 889.8 0.773
 8 0.850 17.270 26.080 614.4 39.11 962.0 0.839 905.7 0.790
 9 0.839 16.784 25.875 611.9 34.01 909.6 0.798 924.9 0.811
 10 0.827 16.161 25.673 610.0 27.06 857.1 0.756 952.8 0.840
 11 0.803 15.111 25.616 609.7 15.85 819.6 0.730 1012.7 0.902
 12 0.799 14.323 25.220 607.6 8.93 810.2 0.727 1043.9 0.936
 13 0.751 13.270 24.388 602.8 0.01 807.1 0.731 1075.2 0.974

STA 15 000 MASS AVERAGED PROPERTIES
PT= 25.689 TI= 617.96 GAMMA= 1.4010 PT-RAT= 1.748 TI-RAT= 1.191
RCU= 3375.9 VM= 727.2 CZ= 707.1 MM= 0.633 MAB= 0.780 MREL= 0.870

ROTOR 1
 WTF = 61.365 I=16 OPTX=TT
 PSIC Z R
 0. -6 334 8.500 0.
 0. 0.050 -6 286 8.312 0.39
 0. 0.100 -6 244 8.134 0.36
 0. 0.200 -6 173 7.793 0.42
 0. 0.300 -6 114 7.459 1.80
 0. 0.400 -6 062 7.120 3.65
 0. 0.500 -6 015 6.767 5.99
 0. 0.600 -5 974 6.389 8.88
 0. 0.700 -5 925 5.982 12.54
 0. 0.800 -5 864 5.536 17.45
 0. 0.900 -5 838 5.006 24.67
 0. 0.950 -5 871 4.677 29.77
 0. 1.000 -5 960 4.235 36.53

STA= 15.500 AFLOW= 146.77 IN ROTOR
 MTIP= 196 ITYPE=5 INBR=3 D=C=O.
 OPTV=PT CURV VM ALPHAM MM
 PHI 0. 0. 531.8 512.5 43.94 0.442
 0. 0. 578.0 515.8 41.75 0.483
 0. 0. 611.6 518.3 40.28 0.514
 0. 0. 660.3 520.0 38.22 0.560
 0. 0. 688.2 524.5 37.32 0.587
 0. 0. 700.0 532.7 37.26 0.600
 0. 0. 703.4 546.5 37.84 0.605
 0. 0. 707.2 565.4 38.64 0.611
 0. 0. 712.5 590.7 39.66 0.619
 0. 0. 724.8 629.8 40.99 0.633
 0. 0. 764.0 695.1 42.29 0.675
 0. 0. 787.1 740.3 43.24 0.702
 0. 0. 799.3 799.9 44.18 0.744

STA 15.500 MASS AVERAGED PROPERTIES
 PT= 27.244 TT= 629.44 GAMMA= 1.4008 PT-RAT= 1.854 TT-RAT= 1.214
 RCU= 3767.1 VM= 695.0 CZ= 674.1 MM= 0.600 MABS= 0.781 MREL= 0.817

TE ROTOR							
ROTOR 1		STA = 16.000		AFLOW= 148.33		D+C=0.	
WTR=	I=17	MTIP=209	OPTX=TT	ITYPE=6	INBR=3	D+H=0.	
PSIC	Z	R	PHI	CURV	VM	ALPHAM	NN
0.	-6	131	8.500	0.	501.0	557.9	0.413
0.050	-6	072	8.311	-0.13	-0.0402	551.2	45.34
0.100	-6	022	8.133	0.10	-0.0624	584.8	45.34
0.200	-5	935	7.795	1.08	-0.0862	634.0	43.65
0.300	-5	861	7.468	2.57	-0.0890	658.3	41.55
0.400	-5	795	7.139	4.45	-0.0788	668.6	40.87
0.500	-5	735	6.798	6.79	-0.0625	669.8	41.79
0.600	-5	683	6.437	5.64	-0.0419	671.7	42.76
0.700	-5	626	6.051	13.11	-0.0172	678.4	43.80
0.800	-5	560	5.633	17.60	0.0315	694.7	44.82
0.900	-5	537	5.144	24.00	0.0933	728.0	45.72
0.950	-5	576	4.845	28.35	0.1749	761.7	46.12
1.000	-5	677	4.442	35.49	0.0883	780.9	48.09
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL
1	0.926	20.517	26.560	658.0	62.00	1067.1	0.881
2	0.927	20.399	27.120	654.9	58.76	1062.9	0.882
3	0.927	20.322	27.560	652.0	56.32	1054.5	0.880
4	0.927	20.232	28.390	647.4	52.08	1031.6	0.868
5	0.928	20.173	28.940	643.7	48.66	996.6	870.5
6	0.928	20.088	29.280	640.8	45.39	952.1	809.0
7	0.928	19.910	29.380	638.3	41.90	899.9	768.3
8	0.928	19.559	29.350	636.2	37.47	846.3	725.0
9	0.928	19.025	29.280	634.4	31.60	796.5	686.6
10	0.928	18.193	29.160	633.0	23.62	758.2	657.9
11	0.930	16.832	28.920	631.6	12.49	745.6	654.0
12	0.914	15.685	28.770	631.5	4.73	764.3	676.0
13	0.875	14.309	28.650	632.3	-6.31	785.7	704.1

STA = 16.000 MASS AVERAGED PROPERTIES
 PT= 28.783 TT= 640.23 GAMMA= 1.4006 PI-RATE= 1.959 TI-RATE= 1.234
 RCU= 4135.2 VM= 663.7 C2= 644.1 MM= 0.568 MABS= 0.783 MREL= 0.768

AVERAGE	BLADE SPEED	ACC PT	EFFICIENCY	AXIAL
PCT 1MM RAD	IN OUT	RATIO	AD.	POLY VEL R
0.	8.500	1500.0	1.8073	0.686 0.711 0.600
3.7	8.316	1468.6	1.8454	1.2626 0.729 0.751 0.658
7.3	8.137	1436.5	1.8753	1.2570 0.766 0.786 0.697
14.6	7.779	1369.9	1.9318	1.2431 0.835 0.849 0.756
21.9	7.416	1299.5	1.9692	1.2410 0.887 0.897 0.789
29.5	7.038	1224.4	1.9924	1.2354 0.925 0.932 0.816
37.6	6.637	1142.9	1.9992	1.2306 0.950 0.955 0.843
46.4	6.202	1053.2	1.9971	1.2266 0.965 0.969 0.882
56.1	5.724	952.3	1.9924	1.2231 0.977 0.979 0.941
67.0	5.184	835.4	994.1	1.2204 0.982 0.984 1.030
80.3	4.524	689.0	907.8	1.2177 0.981 0.983 1.182
88.8	4.104	593.3	855.0	1.2175 0.974 0.976 1.330
100.0	3.547	468.2	783.8	1.2190 0.960 0.964 1.348

FREE								FREE									
WTF	I=18	STA= 17.000	MTIP=222	AFLOW= 146.21	D+C=0.	D+H=0.	ARH=0.	PS1C	Z	OPTX=DPP	OPTY=FREE	I TYPE=O	INBR=O	ABC=0.	MM		
	R	PHI	CURV	VM	CU	ALPHAM	MM		R	PHI	CURV	VM	CU	ALPHAM	MM		
0	-5.700	8.500	0	0	498.6	557.9	48.21	0.411	0	-5.639	8.314	0.0377	551.5	557.6	45.32	0.458	
0.050	-5.587	8.140	1.56	-0.0551	587.3	557.4	43.50	0.490	0.100	-5.500	7.811	2.94	-0.0627	642.9	560.7	41.10	0.541
0.200	-5.430	7.495	4.34	-0.0540	673.1	567.5	40.14	0.570	0.300	-5.375	7.178	5.87	-0.0389	690.3	578.8	39.98	0.588
0.400	-5.333	6.850	7.70	-0.0156	697.0	594.1	40.44	0.596	0.500	-5.305	6.503	9.96	0.0127	700.0	614.7	41.29	0.601
0.600	-5.294	6.129	12.83	0.0453	704.8	642.3	42.34	0.608	0.700	-5.302	5.713	16.72	0.0828	710.2	680.7	43.78	0.616
0.800	-5.350	5.225	22.55	0.1554	725.5	735.1	45.37	0.635	0.900	-5.405	4.934	26.21	0.2126	744.0	777.8	46.27	0.656
0.950	-5.521	4.550	33.90	0.2101	732.4	849.4	49.23	0.650	1.000								

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.940	20.520	26.533	658.0	62.11	1065.9	0.879	748.2	0.617
2	0.940	20.378	27.093	654.9	58.77	1063.6	0.883	784.3	0.651
3	0.940	20.278	27.532	652.0	56.25	1057.1	0.882	809.7	0.676
4	0.940	20.132	28.390	647.4	51.83	1040.2	0.876	853.1	0.718
5	0.940	19.997	28.940	643.7	48.28	1011.5	0.857	880.5	0.746
6	0.940	19.828	29.280	640.8	44.89	974.4	0.830	900.9	0.767
7	0.940	19.590	29.380	638.3	41.41	929.3	0.795	915.8	0.783
8	0.940	19.249	29.350	636.2	37.28	879.8	0.756	931.7	0.800
9	0.940	18.771	29.280	634.4	31.93	810.5	0.717	953.6	0.823
10	0.940	18.111	29.160	633.0	24.76	782.0	0.679	983.7	0.853
11	0.940	17.019	28.920	631.6	14.45	749.3	0.656	1032.8	0.904
12	0.940	16.109	28.770	631.5	7.11	749.8	0.661	1076.4	0.949
13	0.940	15.158	28.564	632.3	-3.64	733.9	0.652	1121.6	0.996

STA 17.000 MASS AVERAGED PROPERTIES
 PT= 28.777 TT= 640.23 GAMMA= 1.4006 PT-RAT= 1.958 TT-RAT= 1.234
 RCU= 4.35.2 VM= 675.8 CZ= 657.6 MM= 0.579 MABS= 0.788 MREL= 0.782

STATOR	WTR = 61.365	L=19	OPTX=DPP	OPTY=FREE	ITYPE=1	INER=4	ABC=0.	D+C=0.	D+H=0.	LE STATOR	
										VM	CU
PSIC	Z	R	PHI	CURV			ALPHAM	NM			
0	-5.250	8.500	0.	0.	538.2	557.9	46.03	0.445			
0.050	-5.192	8.323	1.45	-0.0105	581.3	557.0	43.78	0.484			
0.100	-5.143	8.157	2.47	-0.0165	612.4	556.2	42.25	0.512			
0.200	-5.064	7.839	3.97	-0.0199	665.2	558.7	40.03	0.561			
0.300	-5.004	7.532	5.20	-0.0164	697.0	564.8	39.02	0.592			
0.400	-4.963	7.223	6.47	-0.0121	717.5	575.2	38.72	0.613			
0.500	-4.938	6.905	7.98	-0.0086	728.2	589.4	38.98	0.625			
0.600	-4.933	6.568	9.89	-0.0057	735.0	608.7	39.63	0.633			
0.700	-4.952	6.204	12.40	-0.0018	742.7	634.4	40.50	0.643			
0.800	-5.003	5.800	15.88	0.0114	751.4	670.5	41.74	0.654			
0.900	-5.107	5.321	21.17	0.0299	762.0	721.8	43.45	0.669			
0.950	-5.201	5.029	24.68	0.0243	753.9	763.1	45.35	0.664			
1.000	-5.375	4.643	31.23	0.3271	803.7	832.3	46.00	0.718			
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	MABS	MABS		
1	0.940	20.121	26.533	658.0	60.26	1085.0	0.898	775.2	0.641		
2	0.940	20.054	27.093	654.9	57.48	1081.4	0.900	805.1	0.670		
3	0.940	19.993	27.532	652.0	55.26	1074.7	0.899	827.3	0.692		
4	0.940	19.864	28.390	647.4	51.11	1059.4	0.894	868.7	0.733		
5	0.940	19.702	28.940	643.7	47.64	1034.4	0.878	897.1	0.762		
6	0.940	19.486	29.280	640.8	44.27	1002.0	0.856	919.6	0.785		
7	0.940	19.201	29.380	638.3	40.82	962.3	0.825	936.9	0.804		
8	0.940	18.827	29.350	636.2	36.83	918.3	0.791	954.3	0.822		
9	0.940	18.336	29.280	634.4	31.80	873.9	0.756	976.8	0.846		
10	0.940	17.672	29.160	633.0	25.17	830.2	0.723	1007.0	0.877		
11	0.940	16.7C3	28.920	631.6	15.91	792.4	0.695	1049.6	0.921		
12	0.940	16.180	28.770	631.5	9.37	764.1	0.673	1072.7	0.945		
13	0.940	14.494	28.564	632.3	-0.92	803.8	0.718	1157.0	1.034		

STA 18.000 MASS AVERAGED PROPERTIES
 PT = 28.777 TT = 640.23 GAMMA = 1.4007 PT-RAT = 1.958 TT-RAT = 1.234
 RCU = 4135.2 VM = 707.1 CZ = 689.5 MM = 0.607 MABS = 0.806 MREL = 0.811

STATOR

STA = 19.000

WTF = 61.365 OPTX=OPP OPTY=BETM ITYPE=2 INBR=4 ABC=0 ABH=0.

PSIC Z R PHI CURV VM ALPHAM MN

0. -4.770 8.500 0. 0. 609.8 372.5 31.42 0.501

0.050 -4.724 8.335 1. 13. 0.0344 639.4 382.0 30.86 0.529

0.100 -4.683 8.177 2. 00. 0.0517 663.4 388.9 30.38 0.551

0.200 -4.617 7.870 3. 43. 0.0620 709.0 404.3 29.69 0.594

0.300 -4.567 7.572 4. 70. 0.0559 740.3 415.2 29.29 0.625

0.400 -4.532 7.272 6. 05. 0.0464 762.5 425.0 29.13 0.647

0.500 -4.512 6.964 7. 63. 0.0372 775.7 433.9 29.22 0.661

0.600 -4.508 6.642 9. 56. 0.0321 786.2 444.8 29.50 0.673

0.700 -4.524 6.298 11. 98. 0.0351 799.5 462.5 30.05 0.687

0.800 -4.565 5.922 15. 13. 0.0462 818.5 490.5 30.93 0.707

0.900 -4.640 5.496 19. 48. 0.0881 848.1 531.0 32.05 0.739

0.950 -4.696 5.253 22. 45. 0.1170 873.4 558.8 32.61 0.765

1.000 -4.770 4.975 26. 23. 0.1265 905.8 597.2 33.40 0.799

IN STATOR

MTIP=248 D+C=0. D+H=0.

OPTY=BETM ITYPE=2 INBR=4 ABC=0 ABH=0.

CURV VM ALPHAM MN

0.100 0. 0. 609.8 372.5 31.42 0.501

0.200 1. 13. 0.0344 639.4 382.0 30.86 0.529

0.300 2. 00. 0.0517 663.4 388.9 30.38 0.551

0.400 3. 43. 0.0620 709.0 404.3 29.69 0.594

0.500 4. 70. 0.0559 740.3 415.2 29.29 0.625

0.600 6. 05. 0.0464 762.5 425.0 29.13 0.647

0.700 7. 63. 0.0372 775.7 433.9 29.22 0.661

0.800 9. 56. 0.0321 786.2 444.8 29.50 0.673

0.900 11. 98. 0.0351 799.5 462.5 30.05 0.687

0.950 15. 13. 0.0462 818.5 490.5 30.93 0.707

1.000 19. 48. 0.0881 848.1 531.0 32.05 0.739

1.100 22. 45. 0.1170 873.4 558.8 32.61 0.765

1.200 26. 23. 0.1265 905.8 597.2 33.40 0.799

MASS AVERAGED PROPERTIES

PT = 28.777 TT = 640.23 GAMMA = 1.4005 PT-RAT = 1.958 TT-RAT = 1.234

RCU = 3037.7 VM = 766.1 CZ = 749.6 MM = 0.653 MABS = 0.757 MREL = 0.940

STATOR I=21 STA= 20.000 AFLOW= 118.87 IN STATOR
 WTF = 61.365 MTIP=261 OPTY=BETM ITYPE=2 INBR=4 D=C=0.
 PSIC Z R PHI CURV VM CU ALPHAM MM ABH=0.
 0 -4 300 8.500 0. 0. 649.8 260.2 21.82 0.534
 0.050 -4.277 8.341 0.63 0.0046 679.3 269.6 21.64 0.561
 0.100 -4.258 8.188 1.26 0.0092 701.6 276.2 21.49 0.582
 0.200 -4.225 7.890 2.55 0.0167 742.6 288.6 21.24 0.621
 0.300 -4.201 7.598 3.89 0.0211 769.3 295.8 21.03 0.648
 0.400 -4.184 7.306 5.34 0.0248 787.9 300.9 20.90 0.666
 0.500 -4.174 7.007 6.96 0.0309 798.5 304.3 20.86 0.678
 0.600 -4.173 6.696 8.85 0.0408 807.0 308.6 20.93 0.687
 0.700 -4.180 6.368 11.08 0.0540 818.7 316.7 21.15 0.699
 0.800 -4.199 6.C18 13.82 0.0747 836.4 331.0 21.59 0.717
 0.900 -4.232 5.632 17.32 0.0874 860.2 351.1 22.20 0.741
 0.950 -4.260 5.420 19.55 0.1000 876.8 363.5 22.52 0.758
 1.000 -4.300 5.188 22.50 0.1259 897.8 379.5 22.91 0.778

SL	BLDBLK	PS	PT	T1	BETAM	VREL	MREL	VABS	MABS
1	0.849	21.212	26.533	658.0	62.34	1399.7	1.149	699.9	0.575
2	0.851	21.184	27.093	654.9	60.53	1381.1	1.140	730.9	0.603
3	0.852	21.152	27.532	652.0	59.02	1363.1	1.131	754.0	0.625
4	0.855	21.075	28.390	647.4	56.07	1330.3	1.113	796.8	0.667
5	0.857	20.978	28.940	643.7	53.64	1297.7	1.092	824.2	0.694
6	0.859	20.856	29.280	640.8	51.44	1264.0	1.069	843.4	0.713
7	0.860	20.700	29.380	638.3	49.42	1227.5	1.042	854.5	0.725
8	0.860	20.484	29.350	636.2	47.25	1189.0	1.012	864.0	0.735
9	0.858	20.164	29.280	634.4	44.59	1149.7	0.982	877.8	0.750
10	0.855	19.671	29.160	633.0	41.15	1110.8	0.952	899.5	0.771
11	0.847	18.953	28.920	631.6	36.77	1073.9	0.926	929.0	0.801
12	0.840	18.485	28.770	631.5	34.07	1058.5	0.915	949.1	0.820
13	0.828	17.897	28.564	632.3	30.84	1045.6	0.907	974.7	0.845

STA 20.000 MASS AVERAGED PROPERTIES
 P1= 28.777 T1= 640.23 GAMMA= 1.4004 PT-RAT= 1.958 TT-RAT= 1.234
 RCU= 2126.4 VM= 789.7 CZ= 776.4 MM= 0.670 MABS= 0 719 MREL= 1.035

STATOR	WTF = 61	365	I=22	OPTX=DPP	MTIP=274	STA= 21.000	AFLOW= 115 15	D+C=0.
			Z					D+H=0.
PSIC	-3.800	8.500	0.	0.	0.	VM	CU	ABC=0.
0.	-3.800	8.346	0.	0.53	0.0027	696.6	173.9	ABH=0.
0.050	-3.800	8.197	1.08	0.0046	719.5	177.5	MM	ALPHAM
0.100	-3.800	7.907	2.24	0.0091	761.3	184.1	13.86	0.596
0.200	-3.800	7.624	3.47	0.0153	787.6	187.8	13.59	0.636
0.300	-3.800	7.340	4.81	0.0228	805.7	190.3	13.41	0.662
0.400	-3.800	7.051	6.28	0.0320	815.8	191.5	13.29	0.680
0.500	-3.800	6.752	7.95	0.0428	823.4	192.5	13.16	0.699
0.600	-3.800	6.439	9.87	0.0554	832.9	195.4	13.20	0.709
0.700	-3.800	6.109	12.12	0.0704	845.1	202.0	13.44	0.722
0.800	-3.800	5.757	14.94	0.0974	860.8	211.3	13.79	0.738
0.900	-3.800	5.571	16.61	0.1114	873.0	216.0	13.90	0.750
0.950	-3.800	5.376	18.63	0.1266	887.1	208.8	13.98	0.763
1.000	-3.800	5.376	18.63					
SL	BLOBLK	PS	PT	TT	BETAM	VREL	MREL	MABS
1	0.849	21.395	26.533	658.0	63.43	1488.7	1.221	686.8
2	0.850	21.373	27.093	654.9	61.79	1473.9	1.215	686.8
3	0.851	21.350	27.532	652.0	60.45	1458.8	1.209	686.8
4	0.853	21.296	28.390	647.4	57.85	1430.7	1.195	686.8
5	0.855	21.226	28.940	643.7	55.77	1400.1	1.177	686.8
6	0.856	21.128	29.280	640.8	53.90	1367.6	1.154	686.8
7	0.858	20.995	29.380	638.3	52.23	1331.9	1.128	686.8
8	0.858	20.813	29.350	636.2	50.50	1294.6	1.099	686.8
9	0.859	20.564	29.280	634.4	48.48	1256.6	1.070	686.8
10	0.858	20.224	29.160	633.0	46.03	1217.3	1.040	686.8
11	0.855	19.728	28.920	631.6	43.07	1178.4	1.010	686.8
12	0.854	19.391	28.770	631.5	41.30	1162.1	0.998	686.8
13	0.852	18.997	28.564	632.3	39.37	1147.5	0.986	686.8

STA 21.000 MASS AVERAGED PROPERTIES
 PI= 28.776 TT= 640.23 GAMMA= 1.4003 PT-RAT= 1.958 TT-RAT= 1.234
 RCU= 1341.2 VM= 803.0 CZ= 792.9 MM= 0.679 MABS= 0.698 MREL= 1.119

STATOR	I=23		M1IP=287		AFLOW= 114.73		D+C=0.		IN STATOR	
	WTF= 61.365	OPTX=DPP	OPTV=BETIM	11TYPE=2	VM	CURV	CU	ALPHAM	MM	D+H=0.
PSIC	Z	R	PHII							ZC1=0.
0	-3.204	8.500	0.	0.	656.8	82.8	7.19	0.537		
0.050	-3.211	8.351	0.54	-0.0034	688.4	85.5	7.08	0.566		
0.100	-3.218	8.207	1.04	-0.0020	711.2	87.0	6.97	0.588		
0.200	-3.232	7.928	1.99	0.0061	753.5	89.8	6.79	0.628		
0.300	-3.245	7.655	2.99	0.0149	780.2	91.1	6.66	0.654		
0.400	-3.259	7.382	4.09	0.0236	798.7	91.8	6.56	0.672		
0.500	-3.272	7.104	5.31	0.0320	808.9	91.9	6.48	0.683		
0.600	-3.286	6.817	6.69	0.0420	816.2	92.0	6.43	0.691		
0.700	-3.301	6.519	8.29	0.0540	825.2	92.9	6.43	0.701		
0.800	-3.316	6.205	10.32	0.0572	835.3	95.3	6.51	0.711		
0.900	-3.333	5.871	12.71	0.0638	842.8	98.3	6.65	0.719		
0.950	-3.341	5.696	13.98	0.0821	848.2	99.9	6.72	0.724		
1.000	-3.350	5.512	15.23	0.1249	857.7	101.8	6.77	0.733		
SL	BLDBLK	PS	PT	IT	BETAM	VREL	MREL	VABS	MABS	
1	0.880	21.735	26.533	658.0	65.14	1561.9	1.278	662.0	0.542	
2	0.880	21.725	27.093	654.9	63.63	1549.6	1.275	693.7	0.571	
3	0.881	21.718	27.532	652.0	62.41	1536.0	1.269	716.6	0.592	
4	0.882	21.690	28.390	647.4	60.08	1510.6	1.259	758.9	0.632	
5	0.883	21.633	28.940	643.7	58.23	1481.9	1.242	785.5	0.658	
6	0.883	21.542	29.280	640.8	56.59	1450.6	1.221	804.0	0.677	
7	0.884	21.412	29.380	638.3	55.15	1415.7	1.196	814.1	0.688	
8	0.885	21.240	29.350	636.2	53.70	1378.7	1.168	821.4	0.696	
9	0.886	21.011	29.280	634.4	52.03	1341.3	1.139	830.4	0.705	
10	0.886	20.727	29.160	633.0	50.12	1302.7	1.109	840.7	0.716	
11	0.886	20.403	28.920	631.6	48.06	1260.9	1.076	848.5	0.724	
12	0.887	20.199	28.770	631.5	46.86	1240.5	1.059	854.0	0.729	
13	0.887	19.894	28.564	632.3	45.44	1222.4	1.044	863.7	0.738	

STA 22.000 MASS AVERAGED PROPERTIES

PI = 28.776 TT = 640.23 GAMMA = 1.4007 PI-RAT = 1.958 TT-RAT = 1.234
 RCU = 649.0 VM = 793.0 CZ = 786.0 MM = 0.668 MABS = 0.673 MREL = 1.184

STATOR		I=24		STA = 23.000		AFLOW = 118.18		D+C=0.		TE STATOR	
WTF	61.365	OPTX=DPP	Z	WTIP=300	OP1Y=BETM	I TYPE=3	INBR=4	ABC=0.	ABH=0.	MM	
PSIC		R	PHI	CURV	VM	CU	ALPHAM				
O.	-2.567	8.500	O.	O.	651.5	O.	O.	O.	O.	0.533	
0.050	-2.581	8.358	0.49	0.0662	678.4	O.	O.	O.	O.	0.557	
0.100	-2.595	8.219	0.91	0.0089	696.0	O.	O.	O.	O.	0.574	
0.200	-2.622	7.948	1.67	0.0121	738.2	O.	O.	O.	O.	0.614	
0.300	-2.648	7.684	2.43	0.0178	764.1	O.	O.	O.	O.	0.639	
0.400	-2.674	7.420	3.25	0.0261	782.7	O.	O.	O.	O.	0.657	
0.500	-2.700	7.152	4.17	0.0372	792.7	O.	O.	O.	O.	0.668	
0.600	-2.727	6.876	5.21	0.0498	798.8	O.	O.	O.	O.	0.675	
0.700	-2.756	6.589	6.42	0.0648	807.7	O.	O.	O.	O.	0.684	
0.800	-2.785	6.292	8.00	0.0932	827.7	O.	O.	O.	O.	0.703	
0.900	-2.816	5.977	9.87	0.1246	827.4	O.	O.	O.	O.	0.704	
0.950	-2.832	5.809	10.79	0.1318	827.3	O.	O.	O.	O.	0.704	
1.000	-2.850	5.631	11.52	0.1267	815.2	O.	O.	O.	O.	0.692	
SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS		
1	0.940	21.526	26.109	658.0	66.52	1635.4	1.337	651.5	0.533		
2	0.940	21.527	26.581	654.9	65.30	1623.4	1.334	678.4	0.557		
3	0.940	21.523	26.910	652.0	64.36	1608.7	1.327	696.0	0.574		
4	0.940	21.507	27.731	647.4	62.24	1585.0	1.318	738.2	0.614		
5	0.940	21.478	28.269	643.7	60.60	1556.5	1.301	764.1	0.639		
6	0.940	21.425	28.638	640.8	59.13	1525.5	1.281	782.7	0.657		
7	0.940	21.337	28.780	638.3	57.87	1490.5	1.256	792.7	0.668		
8	0.940	21.204	28.769	636.2	56.64	1452.8	1.227	798.8	0.675		
9	0.940	21.019	28.749	634.4	55.22	1415.8	1.199	807.7	0.684		
10	0.940	20.734	28.851	633.0	53.30	1384.9	1.177	827.7	0.703		
11	0.940	20.293	28.252	631.6	51.89	1340.6	1.141	827.4	0.704		
12	0.940	20.013	27.861	631.5	51.10	1317.3	1.121	827.3	0.704		
13	0.940	19.713	27.159	632.3	50.64	1285.4	1.091	815.2	0.692		

STIA 23.000 MASS AVERAGED PROPERTIES
 PT= 28.163 TT= 640.23 GAMMA= 1.4002 PT-RAT= 1.916 TT-RAT= 1.234
 RCU= 0. VM= 777.4 C2= 773.3 MM= 0.654 MABS= 0.654 MREL= 1.244

AVERAGE	BLADE SPEED	ACC PT	ACC TT	EFFICIENCY	AXIAL
PCT IMM RAD	IN OUT	RATIC	RATIO	AD. POLY	VEL R
0. 8.500		1.7766	1.2686	0.665	0.691
4.7 8.340		1.8088	1.2626	0.703	1.210
9.3 8.188		1.8311	1.2570	0.734	1.167
18.0 7.893		1.8870	1.2481	0.802	0.819
26.5 7.608		1.9236	1.2410	0.853	0.866
35.0 7.322		1.9487	1.2354	0.893	0.902
43.8 7.028		1.9584	1.2306	0.919	0.926
52.9 6.722		1.9576	1.2266	0.935	0.941
62.5 6.397		1.9563	1.2231	0.948	0.953
73.0 6.046		1.9632	1.2204	0.965	0.968
84.8 5.649		1.9224	1.2177	0.944	0.949
91.6 5.419		1.8958	1.2175	0.923	0.929
100.0 5.137		1.8480	1.2190	0.876	0.887

STA= 24.000 MTTIP=313 AFLOW= 116.57 D+C=0. D+H=0. FREE
 WTF= 61.365 OPTX=DPP OPTY=FREE ITYPE=O INBR=O ABC=0. ABH=0.
 PSIC 2 R
 0. -2.000 8.500 0. 0. 668.7 0. 0. 0.548
 0.050 -2.000 8.362 0.32 0.0039 695.2 0. 0. 0.572
 0.100 -2.000 8.226 0.63 0.0079 712.5 0. 0. 0.589
 0.200 -2.000 7.964 1.18 0.0157 754.5 0. 0. 0.628
 0.300 -2.000 7.707 1.67 0.0231 781.5 0. 0. 0.655
 0.400 -2.000 7.452 2.15 0.0309 801.4 0. 0. 0.675
 0.500 -2.000 7.194 2.63 0.0398 812.9 0. 0. 0.687
 0.600 -2.000 6.929 3.12 0.0506 820.4 0. 0. 0.695
 0.700 -2.000 6.656 3.62 0.0640 830.7 0. 0. 0.705
 0.800 -2.000 6.374 4.11 0.0789 850.3 0. 0. 0.725
 0.900 -2.000 6.098 4.59 0.0994 846.8 0. 0. 0.722
 0.950 -2.000 5.922 4.85 0.1149 844.5 0. 0. 0.720
 1.000 -2.000 5.757 5.28 0.1260 831.2 0. 0. 0.707

SL	BLDBLK	PS	PT	TT	BETAM	VREL	MREL	VABS	MABS
1	0.950	21.297	26.109	658.0	65.97	1642.3	1.345	668.7	0.548
2	0.950	21.294	26.109	654.9	64.77	1631.1	1.342	695.2	0.572
3	0.950	21.286	26.910	652.0	63.86	1617.1	1.336	712.5	0.589
4	0.950	21.254	27.731	647.4	61.77	1595.1	1.328	754.5	0.628
5	0.950	21.197	28.269	643.7	60.12	1568.6	1.314	781.5	0.655
6	0.950	21.113	28.638	640.8	58.64	1540.1	1.296	801.4	0.675
7	0.950	20.995	28.780	638.3	57.37	1507.5	1.273	812.9	0.687
8	0.950	20.837	28.769	636.2	56.14	1472.5	1.247	820.4	0.695
9	0.950	20.624	28.749	634.4	54.73	1438.6	1.222	830.7	0.705
10	0.950	20.339	28.851	633.0	52.91	1410.0	1.202	850.3	0.725
11	0.950	19.959	28.252	631.6	51.71	1366.6	1.165	846.8	0.722
12	0.950	19.722	27.861	631.5	51.06	1343.6	1.146	844.5	0.720
13	0.950	19.453	27.159	632.3	50.71	1312.6	1.117	831.2	0.707

STA= 24.000 MASS AVERAGED PROPERTIES
 PT= 28.163 TT= 640.23 GAMMA= 1.4002 PT-RAT= 1.916 TT-RAT= 1.234
 RCU= 0. VM= 796.6 CZ= 795.5 MM= 0.671 MABS= 0.671 MREL= 1.262

EXIT	I=26	STA= 25.000	AFLOW= 116 27	D+C=O.	FREE
WTF=	61.365	MTIP=326	OPTY-FREE	INBR-O	D+H=O.
PSIC	Z R	OPTX=DPP	ITYPE=O	ABC=O.	ABII=O.
	PHI CURV	VM	CU	ALPHAM	MM
0	-1.270 8.500	0. 0.	685.8	0.	0. 0.562
0.050	-1.270 8.365	0.19 0.0025	711.4	0.	0. 0.586
0.100	-1.270 8.232	0.36 0.0048	728.0	0.	0. 0.602
0.200	-1.270 7.975	0.66 0.0091	768.2	0.	0. 0.641
0.300	-1.270 7.723	0.91 0.0135	793.2	0.	0. 0.665
0.400	-1.270 7.472	1.13 0.0179	810.9	0.	0. 0.683
0.500	-1.270 7.218	1.32 0.0227	819.5	0.	0. 0.693
0.600	-1.270 6.956	1.48 0.0279	823.0	0.	0. 0.697
0.700	-1.270 6.686	1.58 0.0334	827.8	0.	0. 0.703
0.800	-1.270 6.407	1.62 0.0397	839.6	0.	0. 0.715
0.900	-1.270 6.112	1.54 0.0463	824.5	0.	0. 0.701
0.950	-1.270 5.956	1.42 0.0493	814.0	0.	0. 0.691
1.000	-1.270 5.790	0.00 0.1263	794.2	0.	0. 0.673

SL	BLDBLK	PS	PT	TT	RETAM	VREL	MREL	MABS	VAWS
1	0.956	21.067	26.109	658.0	65.43	1649.3	1.353	685.8	0.562
2	0.956	21.065	26.581	654.9	64.27	1638.6	1.350	711.4	0.586
3	0.956	21.060	26.910	652.0	63.38	1625.0	1.345	728.0	0.602
4	0.956	21.040	27.731	647.4	61.37	1603.3	1.337	768.2	0.641
5	0.956	21.005	28.269	643.7	59.80	1576.9	1.323	793.2	0.665
6	0.956	20.954	28.638	640.8	58.41	1548.0	1.304	810.9	0.683
7	0.956	20.884	28.780	638.3	57.24	1514.6	1.280	819.5	0.693
8	0.956	20.792	28.769	636.2	56.16	1477.9	1.252	823.0	0.697
9	0.956	20.674	28.749	634.4	54.95	1441.3	1.223	827.8	0.703
10	0.956	20.526	28.851	633.0	53.40	1408.3	1.199	839.6	0.715
11	0.956	20.342	28.252	631.6	52.61	1357.7	1.155	824.5	0.701
12	0.956	20.235	27.861	631.5	52.25	1329.4	1.129	814.0	0.691
13	0.956	20.052	27.159	632.3	52.15	1294.2	1.096	794.2	0.673

STA 25.000 MASS AVERAGED PROPERTIES
 PT = 28.163 TI = 640.23 GAMMA = 1.4002 PT-RAT = 1.916 TI-RAT = 1.234
 RCU = 0. VM = 797.9 CZ = 797.7 MM = 0.672 MABS = 0.672 MREL = 1.264

FREE
 STA= 26 000 AFLOW= 116.28 D+C=0.
 MTIP=339 INBR=0 ABC=0. ABH=0.
 OPTY=FREE ITYPE=0 CU ALPHAM MM
 I=27 D+H=0.
 WTF= 61.365 OPTX=DPP PHI CURV VM
 PSIC Z R
 0. -0.350 8.500 0. 0. 703.4 0.
 0.050 -0.350 8.367 0.11 -0.0000 728.1 0.
 0.100 -0.350 8.237 0.21 -0.0000 743.9 0.
 0.200 -0.350 7.982 0.38 -0.0000 781.9 0.
 0.300 -0.350 7.733 0.49 -0.0000 804.2 0.
 0.400 -0.350 7.484 0.57 -0.0000 818.5 0.
 0.500 -0.350 7.231 0.62 -0.0000 822.9 0.
 0.600 -0.350 6.970 0.61 -0.0000 821.1 0.
 0.700 -0.350 6.700 0.55 -0.0000 819.1 0.
 0.800 -0.350 6.419 0.40 -0.0000 822.5 0.
 0.900 -0.350 6.121 0.12 -0.0000 736.0 0.
 0.950 -0.350 5.961 -0.10 -0.0000 778.3 0.
 1.000 -0.350 5.791 0. 0. 745.2 0.
 0. 0. 0. 0. 0. 0. 0.
 SL BLDLWK PS PT TT BETAM VREL MREL VABS MABS
 1 0.956 20.824 26.109 658.0 64.88 1656.7 1.361 703.4 0.578
 2 0.956 20.824 26.581 654.9 63.75 1646.3 1.359 728.1 0.601
 3 0.956 20.824 26.910 652.0 62.90 1632.8 1.353 743.9 0.617
 4 0.956 20.824 27.731 647.4 60.97 1611.1 1.346 781.9 0.653
 5 0.956 20.824 28.269 643.7 59.49 1584.0 1.330 804.2 0.675
 6 0.956 20.824 28.638 640.8 58.21 1553.8 1.310 818.5 0.690
 7 0.956 20.824 28.780 638.3 57.18 1518.4 1.284 822.9 0.696
 8 0.956 20.824 28.769 636.2 55.28 1478.9 1.252 821.1 0.695
 9 0.956 20.823 28.749 634.4 55.28 1438.4 1.220 819.1 0.695
 10 0.956 20.823 28.851 633.0 54.02 1399.9 1.189 822.5 0.699
 11 0.956 20.822 28.252 631.6 53.61 1341.8 1.137 796.0 0.675
 12 0.956 20.822 27.861 631.5 53.51 1308.6 1.107 778.3 0.659
 13 0.956 20.822 27.159 632.3 53.90 1264.7 1.066 745.2 0.623

STA 26,000 MASS AVERAGED PROPERTIES

PT= 28.163 TT= 640.23 GAMMA= 1.4002 PT-RAT= 1.916 TT-RAT= 1.234
 RCU= O. VM= 796.1 CZ= 796.0 MM= 0.670 MABS= 0.670 MREL= 1.264

Phase III Propotor

BLADE FORCES

THE FORCE CALCULATIONS ARE 'PER BLADE ROW'.
TO FIND THE FORCE ON A SINGLE BLADE, DIVIDE BY 'NB'.

THE FORCES ARE THAT OF THE AIR ON THE BLADES.
POSITIVE AXIAL IS AFT; POSITIVE TANGENTIAL IS IN ROTATION DIRECTION.
THE COLUMNS HEADED BY F-TAN*, F-AXL*, AND F-RAD* ARE THE TANGENTIAL,
AXIAL, AND RADIAL FORCES PER INCH OF CHANGE IN R-AVG.

SL	R-AVG (IN.)	H-AVG (IN.)	F-TAN*	F-AXL*	F-RAD*
		(LB/IN)	(LB/IN)	(LB/IN)	(LB/IN)
1	8.500	8.	-288.3	-492.4	-36.8
2	8.316	8.184	-292.6	-493.9	-32.7
3	8.137	8.363	-297.6	-495.6	-27.0
4	7.779	8.721	-299.1	-491.5	-21.7
5	7.416	1.084	-296.7	-388.1	-16.9
6	7.038	1.462	-298.3	-367.9	-8.7
7	6.637	1.863	-281.0	-339.2	-6.6
8	6.262	2.298	-270.3	-301.8	-17.2
9	5.724	2.776	-260.1	-259.4	-26.4
10	5.184	3.316	-244.6	-206.9	-31.1
11	4.524	3.976	-214.5	-134.4	-29.1
12	4.104	4.396	-187.0	-82.6	-29.1
13	3.547	4.953	-165.3	-46.1	-28.8
NET TORQUE = -7884.4 IN-LB					
NET TAN. FORCE = -1263.1 LB					
NET AXIAL FORCE = -1311.3 LB					
NET RADIAL FORCE = -189.2 LB					

2. STREAMSURFACE BLADE COORDINATES

Figure 54 shows the stacked Phase III rotor streamsurface sections. Each page of the following tabulation gives the coordinates for one of these sections. The streamline designation for these sections corresponds to the calculation streamlines of the circumferential average flow calculation. Streamline 1 is at the casing and streamline 13 is at the hub. Also given in the tabulations are coordinates for the section meanline, the meanline angle, and the section thickness at each point. Streamsurface section chord, camber angle, and stagger angle are also given. All dimensions in this tabulation are in inches or degrees.

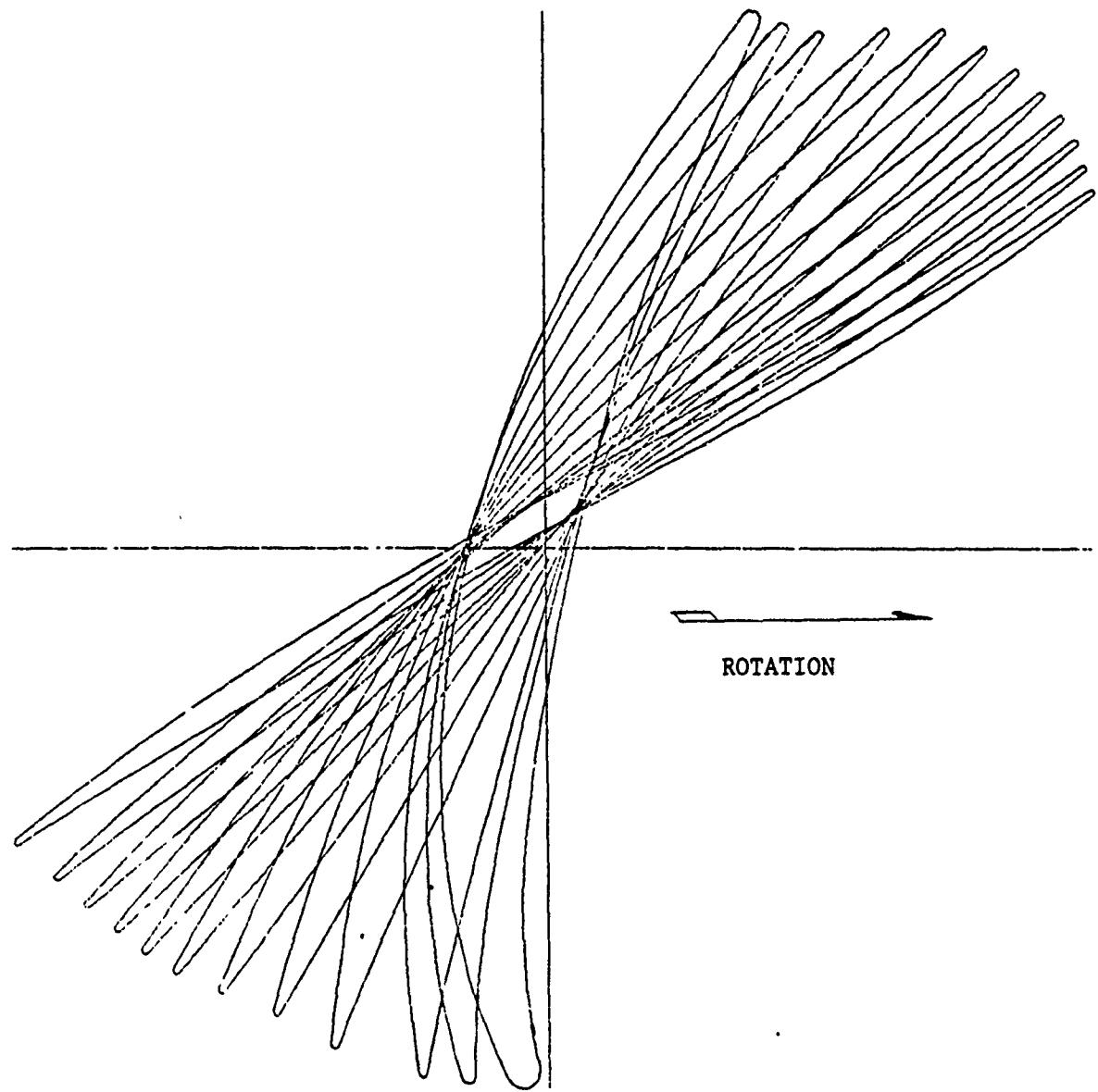


Figure 54. Stacked Phase III Rotor Streamsurface Sections

MERIDIONAL AIRFOIL GEOMETRY

MEANLINE INPUT DATA - STREAMLINE 1

PT	Z	R	THETA	B*	T(Z)	PHI	X	B.M.	T(M)
1	-1 11520	8 .50000	0 .21035	-54 .985	0 .01884	0.	-1 11520	.54 .985	0 .01884
2	-1 06430	8 .50000	0 .20172	-55 .546	0 .02331	0.	-1 06430	.55 .546	0 .02331
3	-0 96250	8 .50000	0 .18390	-56 .645	0 .03219	0.	-0 96250	.56 .645	0 .03219
4	-0 .86080	8 .50000	0 .16532	-57 .795	0 .04090	0.	-0 86080	.57 .795	0 .04090
5	-0 .75900	8 .50000	0 .14590	-58 .943	0 .04930	0.	-0 .75900	.58 .943	0 .04930
6	0 .64700	8 .50000	0 .12339	-60 .414	0 .05808	0.	-0 .64700	.60 .414	0 .05808
7	-0 .52490	8 .50000	0 .09722	-62 .004	0 .06692	0.	-0 .52490	.62 .004	0 .06692
8	-0 .40280	8 .50000	0 .06944	-63 .192	0 .07484	0.	-0 .40280	.63 .192	0 .07484
9	-0 .28060	8 .50000	0 .04062	-63 .660	0 .08168	0.	-0 .28060	.63 .660	0 .08168
10	-0 .15850	8 .50000	0 .01177	-63 .265	0 .08732	0.	-0 .15850	.63 .265	0 .08732
11	-0 .03640	8 .50000	-0 .01619	-62 .262	0 .09163	0.	-0 .03640	.62 .262	0 .09163
12	0 .08570	8 .50000	-0 .04283	-61 .050	0 .09454	0.	0 .08570	.61 .050	0 .09454
13	0 .20790	8 .50000	-0 .06828	-60 .131	0 .09600	0.	0 .20790	.60 .131	0 .09600
14	0 .33000	8 .50000	-0 .09297	-59 .418	0 .09529	0.	0 .33000	.59 .478	0 .09529
15	0 .45210	8 .50000	-0 .11705	-58 .874	0 .08866	0.	0 .45210	.58 .874	0 .08866
16	0 .57430	8 .50000	-0 .14058	-58 .322	0 .07609	0.	0 .57430	.58 .332	0 .07609
17	0 .69640	8 .50000	-0 .16358	-57 .586	0 .05867	0.	0 .69640	.57 .586	0 .05867
18	0 .81850	8 .50000	-0 .18563	-56 .089	0 .03786	0.	0 .81850	.56 .089	0 .03786
19	0 .92030	8 .50000	-0 .20289	-54 .322	0 .01923	0.	0 .92030	.54 .322	0 .01923

MEANLINE INPUT DATA - STREAMLINE 3

PT	Z	R	THETA	B*	T(Z)	PHI	X	B.M.	T(M)
1	-1 .19130	8 .14010	0 .21626	-53 .628	0 .01946	0 .775	-1 .19136	.53 .626	0 .01946
2	-1 .13570	8 .14090	0 .20690	-54 .205	0 .02421	0 .768	1 .13575	.54 .202	0 .02421
3	-1 .02470	8 .14240	0 .18759	-55 .324	0 .03365	0 .727	-1 .02474	.55 .322	0 .03365
4	-0 .91370	8 .14370	0 .16748	-56 .417	0 .04289	0 .612	0 .91373	.56 .415	0 .04289
5	-0 .80270	8 .14480	0 .14652	-57 .498	0 .05180	0 .428	0 .80273	.57 .497	0 .05180
6	-0 .68050	8 .14550	0 .12243	-58 .678	0 .06106	0 .215	-0 .68053	.58 .677	0 .06106
7	-0 .54730	8 .14570	0 .09490	-59 .850	0 .07035	-0 .018	0 .54733	.59 .850	0 .07035
8	-0 .41410	8 .14540	0 .06621	-60 .678	0 .07861	0 .291	-0 .41413	.60 .678	0 .07861
9	-0 .28080	8 .14440	0 .03700	-60 .631	0 .08567	-0 .561	-0 .28082	.60 .630	0 .08567
10	-0 .14760	8 .14270	0 .00854	-59 .418	0 .09138	-0 .751	0 .14761	.59 .416	0 .09139
11	-0 .01440	8 .14080	-0 .01826	-57 .769	0 .09563	-0 .788	-0 .01440	.57 .766	0 .09564
12	0 .11890	8 .13910	-0 .04349	-56 .335	0 .09833	-0 .679	0 .11891	.56 .333	0 .09833
13	0 .25210	8 .13760	-0 .06759	-55 .371	0 .09943	0 .517	0 .25212	.55 .369	0 .09943
14	0 .38530	8 .13660	-0 .09102	-54 .746	0 .09739	-	0 .38532	.54 .745	0 .09739
15	0 .51860	8 .13590	-0 .11395	-54 .187	0 .08948	-0 .321	0 .51862	.54 .187	0 .08948
16	0 .65180	8 .13520	-0 .13639	-53 .547	0 .07625	-0 .355	0 .65183	.53 .546	0 .07625
17	0 .78500	8 .13420	-0 .15823	-52 .679	0 .05871	0 .371	0 .78503	.52 .678	0 .05871
18	0 .91830	8 .13360	-0 .17927	-51 .445	0 .03820	-0 .180	0 .91833	.51 .445	0 .03820
19	1 .02930	8 .13320	0 .19603	-50 .196	0 .01997	0 .995	1 .02933	.50 .196	0 .01997

MFRIDIONAL AIRFOIL GEOMETRY

MEANLINE INPUT DATA - STREAMLINE 4

PT	Z	R	THETA	B*	T(Z)	PHI	X	B.M	T(M)
1	-1.27080	7 76280	0 22306	-52 318	0 02006	1 773	1 27111	-52 305	0 02007
2	-1.21110	7 76470	0 21302	-52 825	0 02541	1 768	-1 21138	52 812	0 02542
3	-1.09170	7 76840	0 19739	-53 807	0 03603	1 748	-1 09192	53 794	0 03604
4	-0.97240	7.71200	0 17102	54 758	0 04642	1 700	0 97257	54 746	0 04643
5	-0.85310	7.77550	0 14890	-55 734	0 05643	1 605	-0 85322	55 724	0 05645
6	-0.72180	7.77900	0 12357	-56 927	0 06683	1 422	-0 72187	56 919	0 06684
7	-0.57860	7.78230	0 09464	-58 029	0 07724	1 138	-0 57864	58 024	0 07725
8	-0.43620	7 78470	0 06494	-58 295	0 08646	0 821	-0 43532	-58 293	0 08647
9	-0.29210	7 78640	0 03553	-57 363	0 09430	0 541	-0 29211	-57 362	0 09430
10	-0.14890	7.78740	0 00787	-55 272	0 10060	0 351	-0 14890	-55 272	0 10060
11	-0.06570	7 78810	-0 01766	-53 334	0 10522	0 239	-0 00570	-53 334	0 10522
12	0.13750	7 78860	-0 04185	-52 298	0 10808	0 203	0 13750	52 298	0 10808
13	0.26070	7 78910	-0 06537	-51 693	0 10911	0 249	0 28070	-51 692	0 10911
14	0.42400	7 78980	0 08842	-51 153	0 10620	0 327	0 42400	-51 152	0 10620
15	0.56220	7 79080	-0 11100	-50 512	0 09702	0 374	0 56721	-50 511	0 09702
16	0.71040	7 79160	-0 13305	-49 884	0 08223	0 359	0 71041	-49 883	0 08223
17	0.85360	7 79260	-0 15461	-49 177	0 06292	0 396	0 85361	-49 177	0 06292
18	0 99680	7 79350	-0 17551	-48 082	0 04050	0 693	0 99682	-48 080	0 04050
19	1 11620	7 79520	-0 19223	-46 919	0 02052	1 083	1 11623	-46 914	0 02052

MEANLINE INPUT DATA - STREAMLINE 5

PT	Z	R	THETA	B*	T(Z)	PHI	X	B.M	T(M)
1	-1 34560	7 36390	0 22853	-50 990	0 02034	3 152	-1 34730	-50 948	0 02036
2	-1 28220	7 36740	0 21782	-51 445	0 02670	3 196	-1 28380	-51 401	0 02673
3	-1 15540	7 37460	0 19590	-52 326	0 03931	3 268	-1 15680	-52 281	0 03935
4	-1 02860	7 38190	0 17330	-53 170	0 05165	3 299	-1 02979	-53 125	0 05170
5	-0 90180	7 38920	0 15003	-54 003	0 06352	3 267	-0 90278	-53 959	0 06359
6	-0 76240	7 39710	0 12360	-54 945	0 07882	3 149	-0 76316	-54 904	0 07590
7	-0 61020	7 40530	0 09388	-55 594	0 08809	2 932	-0 61074	-55 559	0 08817
8	-0 45810	7 41280	0 06403	-55 155	0 09891	2 659	-0 45846	-55 126	0 09898
9	-0 30600	7 41940	0 03529	-53 649	0 10R04	2 376	-0 30621	-53 626	0 10810
10	-0 15380	7 12540	C 00846	-51 537	0 11528	2 126	-0 15390	51 518	0 11533
11	-0 00170	7 43070	-0 01649	-49 841	0 12049	1 932	-0 00170	-49 825	0 12053
12	0 15040	7 43560	-0 04032	-48 936	0 12355	1 801	0 15048	48 922	0 12358
13	0 30260	7 44020	-0 06351	-48 233	0 12435	1 723	0 30275	-48 220	0 12438
14	0 45470	7 44480	-0 08613	-47 569	0 11988	1 687	0 45492	-47 556	0 11991
15	0 60680	7 44920	-0 10824	-46 953	0 10857	1 678	0 60708	46 940	0 10859
16	0 75900	7 45360	-0 12985	-46 273	0 09121	1 673	0 75935	46 260	0 09123
17	0 91110	7 45810	-0 15089	-45 465	0 06898	1 764	0 91152	-45 451	0 06900
18	1 06330	7 46330	-0 17129	-44 543	0 04343	2 126	1 06380	-44 524	0 04344
19	1 19000	7 46790	-0 18776	-43 715	0 02086	2 570	1 19061	-43 686	0 02087

PHASE III ROTOR
MERIDIONAL AIRFOIL GEOMETRY

NB 20

MEANLINE INPUT DATA - STREAMLINF 6

	R	THETA	R*	T(1)	T(2)	PHI	X	B,M	T(M)
P1	2	0.23267	-49.878	0.02048	4.875	-1.41947	-49.775	0.02052	
1	6.93800	0.22147	-50.314	0.02849	4.974	-1.35253	-50.208	0.02855	
2	6.94390	0.22147	-50.158	0.04438	5.152	-1.21840	-51.044	0.04449	
3	6.95570	0.19766	-51.948	0.05990	5.275	-1.08435	-51.830	0.06006	
4	6.96790	0.17350	-52.667	0.07480	5.313	-0.95027	-52.548	0.07500	
5	6.98040	0.14871	-53.327	0.09021	5.232	-0.80275	-53.212	0.09015	
6	6.99410	0.12080	-53.000	0.10550	5.031	-0.64180	-53.000	0.10576	
7	6.63980	0.08960	-53.336	0.11888	4.793	-0.48101	-51.659	0.11914	
8	-0.47960	0.02230	-51.757	0.13005	4.527	-0.32027	-49.340	0.13028	
9	-0.31940	0.03530	-49.428	0.13876	4.229	-0.15951	-47.629	0.13897	
10	-0.15910	0.04780	-47.707	0.14480	3.943	-0.00110	-46.479	0.14498	
11	0.00110	0.05920	-46.547	0.14806	3.714	-0.16176	-45.603	0.14822	
12	0.16140	0.06970	-45.663	0.14806	3.571	-0.32228	-44.798	0.14823	
13	0.32160	0.07990	-44.854	0.14809	3.483	-0.48279	-44.018	0.14119	
14	0.48180	0.08980	-44.071	0.14106	3.422	-0.64138	-43.256	0.12618	
15	0.64210	0.09940	-43.307	0.12637	3.096	-0.80386	-42.435	0.10505	
16	0.80230	0.10880	-42.486	0.10496	3.434	-0.96446	-41.547	0.07826	
17	0.96260	0.11860	-41.604	0.07819	3.599	-1.12501	-40.663	0.04781	
18	1.12280	0.12970	-40.732	0.04776	4.000	-1.25888	-39.929	0.02105	
19	1.25630	0.13860	-40.014	0.02102	4.448				

MEANLINE INPUT DATA - STREAMLINF 7

	R	THETA	R*	T(1)	T(2)	PHI	X	B,M	T(M)
P1	2	0.23549	-48.937	0.02013	6.882	-1.49130	-48.731	0.02021	
1	6.47660	0.22302	-49.363	0.03045	7.027	-1.42088	-49.150	0.03058	
2	6.48540	0.19759	-50.135	0.05092	7.284	-1.27998	-49.906	0.05116	
3	6.50300	0.17160	-50.703	0.07090	7.453	-1.13912	-50.465	0.07125	
4	6.52110	0.14529	-50.991	0.09004	7.514	-0.99R11	-50.749	0.09051	
5	6.53950	0.11630	-50.915	0.10978	7.478	-0.84299	-50.675	0.11034	
6	6.55980	0.08523	-50.123	0.12928	7.335	-0.67387	-49.891	0.12930	
7	6.58180	0.05566	-48.268	0.14622	7.096	-0.50473	-48.049	0.14685	
8	6.60310	0.02835	-45.972	0.16021	6.817	-0.33578	-45.769	0.16080	
9	6.62350	0.00305	-44.135	0.17092	6.539	-0.16684	-43.948	0.17146	
10	6.64320	0.02081	-42.754	0.17810	6.245	0.00191	-42.584	0.17859	
11	6.66200	0.04360	-41.671	0.18157	5.961	0.17057	-41.517	0.18200	
12	6.67990	0.06556	-40.756	0.18027	5.763	-0.33925	-40.613	0.18066	
13	6.69100	0.08680	-39.911	0.16999	5.660	-0.50778	-39.773	0.17033	
14	6.71370	0.10737	-39.071	0.15083	5.641	0.67640	-38.535	0.15112	
15	6.73030	0.12728	-38.233	0.12392	5.717	0.84492	-38.094	0.12416	
16	6.74680	0.14655	-37.385	0.09089	5.935	1.01353	-37.236	0.09107	
17	6.76370	0.16517	-36.502	0.05368	6.353	1.18225	-36.374	0.05380	
18	6.78190	0.18040	-35.740	0.02110	6.795	1.32298	-35.549	0.02115	
19	6.79770	0.18560							

MERIDIONAL AIRFOIL GROMFIRY

MEANLINE INPUT DATA - STREAMLINF 8

PT	Z	R	THETA	R*	T(7)	PHI	X	R.M.	T(M)
1	-1.53950	5.96840	0.23715	-48.024	0.01915	9.193	-1.56132	-47.655	0.01929
2	-1.46680	5.98050	0.22356	-48.299	0.03197	9.378	-1.48765	-47.916	0.03221
3	-1.32140	6.00490	0.19610	-48.744	0.05739	9.707	-1.34021	-48.334	0.05786
4	-1.17610	6.03010	0.16846	-48.891	0.08216	9.919	-1.9275	-48.463	0.08286
5	-1.03070	6.05580	0.14098	-48.635	0.10586	9.989	-1.04542	-48.200	0.10677
6	-0.87080	6.08400	0.11143	-47.837	0.13021	9.934	-0.98277	-47.406	0.13129
7	-0.69630	6.11440	0.08063	-46.248	0.15414	9.771	-0.70565	-45.810	0.15531
8	-0.52190	6.14410	0.05206	-43.840	0.17476	9.561	-0.52874	-43.440	0.17593
9	-0.34740	6.17320	0.02605	-41.350	0.19158	9.338	-0.35184	-40.972	0.19269
10	-0.17300	6.20150	0.00203	-39.623	0.20417	9.110	-0.17515	-39.266	0.20522
11	0.00150	6.22910	-0.02058	-38.348	0.21221	8.865	-0.00152	-38.014	0.21319
12	0.17590	6.25600	-0.04233	-37.208	0.21553	8.617	0.17797	-36.895	0.21642
13	0.35040	6.28200	-0.06305	-36.130	0.21207	8.414	0.35441	-35.836	0.21286
14	0.52480	6.30740	-0.08290	-35.101	0.19808	8.299	0.53068	-34.818	0.19877
15	0.69930	6.33280	-0.10195	-34.109	0.17424	8.312	0.70702	-33.828	0.17487
16	0.87370	6.35840	-0.12022	-33.101	0.14188	8.488	0.88331	-32.812	0.14234
17	1.04620	6.38470	-0.13772	-32.078	0.10280	8.818	1.05981	-31.772	0.10314
18	1.22260	6.41270	-0.15448	-31.094	0.05917	9.247	1.23640	-30.764	0.05937
19	1.36800	6.43680	-0.16791	-30.289	0.02109	9.642	1.38380	-29.936	0.02117

MEANLINE INPUT DATA - STREAMLINF 9

PT	Z	R	THETA	R*	T(7)	PHI	X	B.M.	T(M)
1	-1.57750	5.39540	0.23554	-46.985	0.01860	12.228	-1.61594	-46.328	0.01883
2	-1.50250	5.41220	0.22067	-46.948	0.03390	12.414	-1.53917	-46.271	0.03433
3	-1.35250	5.44590	0.19116	-46.805	0.06423	12.729	-1.38548	-46.093	0.06508
4	-1.20250	5.48000	0.16207	-46.152	0.12374	12.873	-1.21665	-45.724	0.09499
5	-1.05250	5.51430	0.13371	-45.665	0.12191	12.841	-1.07778	-44.939	0.12348
6	-0.88760	5.55180	0.10401	-43.906	0.15074	12.735	-0.90RG9	-43.193	0.15253
7	-0.70760	5.59230	0.07434	-41.162	0.17891	12.611	-0.72449	-40.470	0.18079
8	-0.52760	5.63230	0.04754	-38.694	0.20297	12.465	-0.53079	-38.029	0.20484
9	-0.34760	5.67190	0.02294	-36.727	0.22227	12.291	-0.35551	-36.072	0.22409
10	-0.16760	5.71080	0.00010	-34.972	0.23633	12.099	-0.17135	-34.370	0.23805
11	0.01240	5.74900	-0.02121	-33.320	0.24479	11.913	0.01267	-32.750	0.24638
12	0.19240	5.78670	-0.04112	-31.790	0.24743	11.764	0.19658	-31.248	0.24887
13	0.37240	5.82400	-0.05984	-30.484	0.24096	11.680	0.38041	-29.963	0.24224
14	0.55230	5.86110	-0.07755	-29.326	0.22309	11.677	0.56411	-28.817	0.22419
15	0.73230	5.89830	-0.09435	-28.182	0.19475	11.782	0.74794	-27.677	0.19566
16	0.91230	5.93610	-0.11026	-27.042	0.15740	12.060	0.93190	-26.528	0.15812
17	1.09230	5.97510	-0.12531	-25.885	0.11299	12.463	1.1610	-25.152	0.11349
18	1.27230	6.01570	-0.13949	-24.657	0.06381	12.837	1.30058	-24.112	0.06409
19	1.42230	6.05050	-0.15062	-23.580	0.02102	13.104	1.45451	-23.030	0.02111

MERIDIONAL AIRROLL GEOMETRY

MEANLINE INPUT DATA - STIRFAMLINE 10

	R	THETA	B*	T(Z)	FHL	X	B.M.	T(M)
PT 1	-1.55230	4.73400	0.22877	-45.415	0.02077	16.114	-44.267	0.02119
2	-1.47620	4.75660	0.21254	45.263	0.03828	16.269	-44.092	0.03906
3	-1.32400	4.80180	0.18064	-44.756	0.07238	16.522	-43.549	0.07449
4	-1.17170	4.84710	0.14988	-43.647	0.10670	16.613	-42.224	0.10884
5	-1.01950	4.89230	0.12094	-41.830	0.13880	16.549	-40.629	0.1437
6	-0.85210	4.94190	0.09174	-39.326	0.17153	16.434	-38.160	0.17436
7	-0.66940	4.99580	0.06313	-36.459	0.20329	16.302	-36.342	0.20618
8	-0.48680	5.04880	0.03749	-33.966	0.23012	16.110	-30.50627	0.23294
9	-0.30410	5.10120	0.01422	-31.808	0.25127	15.941	-30.31619	0.25395
10	-0.12140	5.15320	-0.06999	-29.732	0.26619	15.885	-28.12622	0.26868
11	-0.06120	5.20510	-0.02633	-27.767	0.27446	15.886	-26.06363	0.27672
12	0.24390	5.25710	-0.04401	-25.962	0.27563	15.925	-25.090	0.27774
13	0.42660	5.30930	-0.06023	-24.326	0.26584	16.025	-23.484	0.26758
14	0.60920	5.36200	-0.07516	-22.787	0.24421	16.202	-21.970	0.24565
15	0.79190	5.41540	-0.08887	-21.253	0.21179	16.486	-20.453	0.21292
16	0.97460	5.47020	-0.10140	-19.668	0.17015	16.932	-18.877	0.17097
17	1.15720	5.52650	-0.11273	-17.956	0.12130	17.399	-17.184	0.12182
18	1.33990	5.58440	-0.12279	-16.027	0.06760	17.595	-15.313	0.06784
19	1.49210	5.63350	-0.13014	-14.281	0.02102	17.603	-13.638	0.02108

MEANLINE INPUT DATA - STIRFAMLINE 11

	R	THETA	B*	T(Z)	FHL	X	B.M.	T(M)
PT 1	-1.49560	3.90490	0.22116	-42.968	0.02535	21.672	-1.60872	-40.880
2	-1.42020	3.93560	0.20338	-42.581	0.04322	21.849	-1.52754	-40.462
3	-1.26960	3.99690	0.16904	-41.589	0.07865	22.120	-1.36511	-39.427
4	-1.11900	4.05820	0.13669	-39.987	0.11319	22.156	-1.20250	-37.839
5	-0.96830	4.11920	0.10696	-37.712	0.14624	21.971	-1.03988	-35.643
6	-0.80270	4.18570	0.07764	-34.858	0.18022	21.715	-0.86147	-32.906
7	-0.62190	4.25730	0.04953	-31.675	0.21366	21.471	-0.66703	-29.864
8	-0.44110	4.32790	0.02511	-28.538	0.24252	21.247	-0.47290	-26.878
9	-0.26040	4.39780	0.00398	-25.509	0.26611	21.105	-0.27312	-23.997
10	-0.07960	4.46740	-0.01424	-22.669	0.28384	21.109	-0.08534	-21.288
11	0.10110	4.53730	-0.02998	-20.202	0.29530	21.254	0.1044	-18.929
12	0.28190	4.60800	-0.04367	-17.989	0.30020	21.553	0.30261	-16.805
13	0.46270	4.68000	-0.05545	-15.625	0.29575	22.041	0.49731	-14.533
14	0.64340	4.75420	-0.06529	-13.181	0.27610	22.696	0.69270	-12.192
15	0.82420	4.83120	-0.07330	-10.811	0.24210	23.455	0.88922	-9.936
16	1.00490	4.91130	-0.07958	-8.326	0.19567	24.203	1.08677	-7.603
17	1.18570	4.99370	-0.08405	-5.604	0.13943	24.657	1.28542	-5.076
18	1.33650	5.07670	-0.08666	-2.666	0.07654	24.468	1.48429	-2.427
19	1.51710	5.14590	-0.08737	-0.092	0.02162	23.990	1.61946	-0.084

MFRIDIONAL AIRFOIL GEOMETRY

MFANLINF INPUT DATA - STREAMLINE 12

PT	Z	R	THETA	R*	T(Z)	PHI	X	R.M	T(M)
1	-1 47580	3.36160	0 22120	-41 143	0 03326	25 961	-1 63314	-38 150	0 03473
2	-1 40200	3.39840	0 20236	40 501	0 05440	26 022	-1 55103	-37 507	0 05675
3	-1 25460	0 15660	-39 029	0 09624	26 054	-1 38696	-36 064	0 10015	
4	-1 10720	3.54320	0 13360	-37 132	0 13669	25 825	-1 22701	-34 277	0 14169
5	-0 95970	3.61350	0 10366	-34 788	0 17483	25 399	-1 05943	-32 111	0 18031
6	-0 79750	3.68970	0 07438	-31 968	0 21313	25 053	-0 88016	-29 483	0 21870
7	-0 62060	3.77210	0 04653	-28 845	0 24978	24 934	-0 68501	-26 539	0 25470
8	-0 44370	3.85420	0 02257	-25 764	0 27874	24 923	-0 48695	-23 639	0 28354
9	-0 26670	3.93640	0 00209	-22 765	0 30026	25 014	-0 29472	-20 821	0 30436
10	-0 08980	4.01930	-0 01527	-19 884	0 31326	25 242	-0 09934	-18 115	0 31661
11	0 08710	4.10320	-0 02988	-17 229	0 31732	25 572	0 09649	-15 628	0 31995
12	0 26400	4.18860	-0 04210	-14 718	0 31134	26 046	0 29296	-13 279	0 31330
13	0 44100	4.27610	-0 05208	-12 077	0 29474	26 761	0 49054	-10 816	0 29606
14	0 61790	4.36670	-0 06985	-9 416	0 26819	27 611	0 68941	-8 360	0 26896
15	0 79480	4.46120	-0 06556	-6 724	0 23274	28 463	0 88983	5 917	0 23310
16	0 97180	4.55920	-0 06903	3 165	0 18978	29 304	1 09200	-2 761	0 18985
17	1 14870	4.65960	-0 06965	1 438	0 14102	29 776	1 29545	1 248	0 14103
18	1 32560	4.76030	0 06709	6 393	0 08838	29 280	1 49892	5 582	0 08851
19	1 47310	4.84420	-0 06251	10 587	0 04306	28 357	1 66731	9 341	0 04322

MFANLINF INPUT DATA - STREAMLINE 13

PT	Z	R	THETA	R*	T(Z)	PHI	X	R.M	T(M)
1	-1 45600	2.65330	0 23827	-36 479	0 05041	31 199	-1 67260	-32 312	0 05299
2	-1 38530	2.69470	0 21684	-36 105	0 06966	30 603	-1 59021	-32 119	0 07302
3	-1 24380	2.77680	0 18170	-35 204	0 10782	29 513	-1 42675	-31 549	0 11245
4	-1 10230	2.85610	0 14701	-33 975	0 14492	28 724	-1 26482	-30 581	0 15045
5	-0 96080	2.93290	0 11503	-32 310	0 18023	28 302	-1 10383	-29 110	0 18631
6	-0 80510	3.01620	0 08337	-29 910	0 21622	28 181	-0 92714	-26 888	0 22248
7	-0 63530	3.10760	0 05341	-26 755	0 25119	28 431	-0 73434	-23 910	0 25717
8	-0 46550	3.20060	0 02816	-23 484	0 28074	29 100	-0 54070	-20 788	0 28617
9	-0 29570	3.29700	0 00708	-20 497	0 30408	30 104	0 34545	-17 921	0 30888
10	-0 12590	3.39790	-0 01061	-17 993	0 32058	31 281	-0 14799	-15 514	0 32479
11	0 04390	3.50340	-0 02551	-15 680	0 32978	32 389	0 05191	13 335	0 33329
12	0 21370	3.61310	-0 03782	-13 161	0 33130	33 335	0 25409	11 054	0 33392
13	0 38350	3.72660	-0 04739	-10 054	0 32241	34 186	0 45835	-8 344	0 32397
14	0 55330	3.84380	-0 05392	-6 069	0 30253	35 003	0 66462	4 977	0 30309
15	0 72310	3.96430	-0 06559	-0 993	0 27265	35 786	0 87294	-0 805	0 27266
16	0 89290	4.08830	-0 05510	5 240	0 23419	36 379	1 08311	4 223	0 23453
17	1 06270	4.21400	-0 04884	12 253	0 18905	36 561	1 29439	9 866	0 19058
18	1 23250	4.33860	-0 03760	19 299	0 13944	36 136	1 50533	15 791	0 14217
19	1 37400	4.44150	-0 02448	24 952	0 09632	35 488	1 67045	20 750	0 09935

MERIDIONAL AIRFOIL GEOMETRY - SURFACE INF 1

MEANLINE DATA

PT	PCT X	X	Y	R.M.	T (M)
1	0.	-1.11520	1.78797	-54.985	0.01884
2	0.02500	-1.06431	1.71457	55.545	0.02331
3	0.05000	-1.01343	1.63963	56.645	0.02776
4	0.07500	-0.96254	1.56312	56.645	0.03219
5	0.10000	-0.91165	1.48500	-57.196	0.03658
6	0.12500	-0.86076	1.4520	-57.756	0.04090
7	0.15000	-0.80988	1.32362	-58.333	0.04515
8	0.17500	-0.75899	1.24014	58.943	0.04930
9	0.20000	-0.70810	1.15455	-59.596	0.05336
10	0.23000	-0.64704	1.04876	-60.414	0.05808
11	0.26000	-0.58597	0.93937	-61.235	0.06260
12	0.29000	-0.52491	0.82630	-62.004	0.06692
13	0.32000	-0.46384	0.70975	-62.671	0.07101
14	0.35000	-0.40278	0.59018	-63.192	0.07484
15	0.38000	-0.34171	0.46835	-63.531	0.07840
16	0.41000	-0.28065	0.34526	-63.660	0.08168
17	0.44000	-0.21958	0.22209	-63.560	0.08466
18	0.47000	-0.15852	0.10001	-63.265	0.08732
19	0.50000	-0.09745	-0.02012	-62.819	0.08965
20	0.53000	-0.03639	-0.13766	-62.262	0.09163
21	0.56000	-0.02468	-0.25231	-61.647	0.09327
22	0.59000	-0.08574	-0.36405	-61.049	0.09454
23	0.62000	-0.14681	-0.47326	-60.544	0.09544
24	0.65000	-0.20787	-0.58045	-60.131	0.09600
25	0.68000	-0.26894	-0.68603	-59.788	0.09616
26	0.71000	-0.33000	-0.79025	-59.478	0.09529
27	0.74000	-0.39107	-0.89319	-59.170	0.09279
28	0.77000	-0.45213	-0.99490	-58.874	0.08866
29	0.80000	-0.51320	-1.09548	-58.602	0.08305
30	0.83000	-0.57426	-1.19501	-58.332	0.07609
31	0.86000	-0.63533	-1.29344	-58.023	0.06791
32	0.89000	-0.89000	-1.39048	-57.586	0.05867
33	0.92000	-0.92000	-1.48553	-56.940	0.04858
34	0.95000	-0.81853	-1.57790	-56.089	0.03786
35	0.97500	-0.86941	-1.65244	-55.248	0.02861
36	1.00000	0.92030	-1.72456	-54.322	0.01923

PT	PCT X	X	Y	R.M.	T (M)
1	1	-1.11520	1.78797	-54.985	0.01884
2	1	-1.06431	1.71457	55.545	0.02331
3	1	-1.01343	1.63963	56.645	0.02776
4	1	-0.96254	1.56312	56.645	0.03219
5	1	-0.91165	1.48500	-57.196	0.03658
6	1	-0.86076	1.4520	-57.756	0.04090
7	1	-0.80988	1.32362	-58.333	0.04515
8	1	-0.75899	1.24014	58.943	0.04930
9	1	-0.70810	1.15455	-59.596	0.05336
10	1	-0.64704	1.04876	-60.414	0.05808
11	1	-0.58597	0.93937	-61.235	0.06260
12	1	-0.52491	0.82630	-62.004	0.06692
13	1	-0.32000	0.46384	0.70975	-62.671
14	1	-0.35000	0.40278	-0.59018	
15	1	-0.38000	0.34171	-0.46835	
16	1	-0.41000	0.28065	-0.34526	
17	1	-0.44000	0.21958	-0.22209	
18	1	-0.47000	0.15852	-0.10001	
19	1	-0.50000	0.09745	-0.02012	
20	1	-0.53000	0.03639	-0.13766	
21	1	-0.56000	0.02468	-0.25231	
22	1	-0.59000	0.08574	-0.36405	
23	1	-0.62000	0.14681	-0.47326	
24	1	-0.65000	0.20787	-0.58045	
25	1	-0.68000	0.26894	-0.68603	
26	1	-0.71000	0.33000	-0.79025	
27	1	-0.74000	0.39107	-0.89319	
28	1	-0.77000	0.45213	-0.99490	
29	1	-0.80000	0.51320	-1.09548	
30	1	-0.83000	0.57426	-1.19501	
31	1	-0.86000	0.63533	-1.29344	
32	1	-0.89000	0.89000	-1.39048	
33	1	-0.92000	0.92000	-1.48553	
34	1	-0.95000	0.81853	-1.57790	
35	1	-0.97500	0.86941	-1.65244	
36	1	-1.00000	0.92030	-1.72456	

SURFACE COORDINATES

PT	PCT X	X	Y	R.M.	T (M)
1	1	-1.11520	1.78797	-54.985	0.01884
2	1	-1.06431	1.71457	55.545	0.02331
3	1	-1.01343	1.63963	56.645	0.02776
4	1	-0.96254	1.56312	56.645	0.03219
5	1	-0.91165	1.48500	-57.196	0.03658
6	1	-0.86076	1.4520	-57.756	0.04090
7	1	-0.80988	1.32362	-58.333	0.04515
8	1	-0.75899	1.24014	58.943	0.04930
9	1	-0.70810	1.15455	-59.596	0.05336
10	1	-0.64704	1.04876	-60.414	0.05808
11	1	-0.58597	0.93937	-61.235	0.06260
12	1	-0.52491	0.82630	-62.004	0.06692
13	1	-0.32000	0.46384	0.70975	-62.671
14	1	-0.35000	0.40278	-0.59018	
15	1	-0.38000	0.34171	-0.46835	
16	1	-0.41000	0.28065	-0.34526	
17	1	-0.44000	0.21958	-0.22209	
18	1	-0.47000	0.15852	-0.10001	
19	1	-0.50000	0.09745	-0.02012	
20	1	-0.53000	0.03639	-0.13766	
21	1	-0.56000	0.02468	-0.25231	
22	1	-0.59000	0.08574	-0.36405	
23	1	-0.62000	0.14681	-0.47326	
24	1	-0.65000	0.20787	-0.58045	
25	1	-0.68000	0.26894	-0.68603	
26	1	-0.71000	0.33000	-0.79025	
27	1	-0.74000	0.39107	-0.89319	
28	1	-0.77000	0.45213	-0.99490	
29	1	-0.80000	0.51320	-1.09548	
30	1	-0.83000	0.57426	-1.19501	
31	1	-0.86000	0.63533	-1.29344	
32	1	-0.89000	0.89000	-1.39048	
33	1	-0.92000	0.92000	-1.48553	
34	1	-0.95000	0.81853	-1.57790	
35	1	-0.97500	0.86941	-1.65244	
36	1	-1.00000	0.92030	-1.72456	

STAGGER

PT	PCT X	X	Y	R.M.	T (M)
1	1	-1.11520	1.78797	-54.985	0.01884
2	1	-1.06431	1.71457	55.545	0.02331
3	1	-1.01343	1.63963	56.645	0.02776
4	1	-0.96254	1.56312	56.645	0.03219
5	1	-0.91165	1.48500	-57.196	0.03658
6	1	-0.86076	1.4520	-57.756	0.04090
7	1	-0.80988	1.32362	-58.333	0.04515
8	1	-0.75899	1.24014	58.943	0.04930
9	1	-0.70810	1.15455	-59.596	0.05336
10	1	-0.64704	1.04876	-60.414	0.05808
11	1	-0.58597	0.93937	-61.235	0.06260
12	1	-0.52491	0.82630	-62.004	0.06692
13	1	-0.32000	0.46384	0.70975	-62.671
14	1	-0.35000	0.40278	-0.59018	
15	1	-0.38000	0.34171	-0.46835	
16	1	-0.41000	0.28065	-0.34526	
17	1	-0.44000	0.21958	-0.22209	
18	1	-0.47000	0.15852	-0.10001	
19	1	-0.50000	0.09745	-0.02012	
20	1	-0.53000	0.03639	-0.13766	
21	1	-0.56000	0.02468	-0.25231	
22	1	-0.59000	0.08574	-0.36405	
23	1	-0.62000	0.14681	-0.47326	
24	1	-0.65000	0.20787	-0.58045	
25	1	-0.68000	0.26894	-0.68603	
26	1	-0.71000	0.33000	-0.79025	
27	1	-0.74000	0.39107	-0.89319	
28	1	-0.77000	0.45213	-0.99490	
29	1	-0.80000	0.51320	-1.09548	
30	1	-0.83000	0.57426	-1.19501	
31	1	-0.86000	0.63533	-1.29344	
32	1	-0.89000	0.89000	-1.39048	
33	1	-0.92000	0.92000	-1.48553	
34	1	-0.95000	0.81853	-1.57790	
35	1	-0.97500	0.86941	-1.65244	
36	1	-1.00000	0.92030	-1.72456	

PHASE III ROTOR

MFRIDIONAL AIRFOIL GEOMETRY - SURFAMI INF. 3

NB 20

MEANLINE DATA				SURFACE COORDINATES											
PT	PCT X	X	Y	B.M	T(M)	r'	x'	y'	z'	xS	yS	zP	xP	yP	
1	0	19136	1 76137	-53 626	0 01946	1	-1 19136	1 76137	-1 19136	1 19136	1 19136	1 19136	1 19136	1 19136	
2	0.02500	-1 13584	1 68519	-54 201	0 02420	2	-1 19536	1 75447	-1 18743	1 76310	1 76310	1 76310	1 76310	1 76310	
3	0.05000	-1.08032	1 60740	-54 766	0 02894	3	-1 19363	1 74763	-1 17756	1 75944	1 75944	1 75944	1 75944	1 75944	
4	0.07500	-1.02481	1 52798	-55.321	0.03365	4	-1 14565	1 67811	-1 12602	1 69227	1 69227	1 69227	1 69227	1 69227	
5	0 10000	0 96929	1 44691	-55 870	0 03830	5	-1 09214	1 59905	-1 06850	1 61575	1 61575	1 61575	1 61575	1 61575	
6	0 12500	0 91377	1 36416	-56 415	0 04289	6	-1 03854	1 51841	-1 01097	1 53756	1 53756	1 53756	1 53756	1 53756	
7	0 15000	0 85825	1 27969	-56 956	0 04740	7	-1 08514	1 42617	-1 09534	1 45766	1 45766	1 45766	1 45766	1 45766	
8	0 17500	0 80274	1 19345	-57 497	0 05180	8	-0 93164	1 35230	-0 89591	1 37602	1 37602	1 37602	1 37602	1 37602	
9	0 20000	0 74722	1 10540	-58 038	0 05609	9	-0 87812	1 25676	-0 83839	1 29261	1 29261	1 29261	1 29261	1 29261	
10	0 23000	0 68060	0 99728	-58 677	0 06106	10	-0 82458	1 17953	0 78099	1 20737	1 20737	1 20737	1 20737	1 20737	
11	0 26000	0 61398	0 88646	-59 286	0 06582	11	-0 77101	1 09055	-0 72343	1 12024	1 12024	1 12024	1 12024	1 12024	
12	0 29000	0 54736	0 706303	-59 849	0 07035	12	-0 70658	0 98141	-0 65452	1 01315	1 01315	1 01315	1 01315	1 01315	
13	0 32000	0 48074	0 65716	-60 337	0 07462	13	-0 64227	0 86966	-0 58556	0 90327	0 90327	0 90327	0 90327	0 90327	
14	0 35000	0 41412	0 53930	-60 678	0 07861	14	-0 57777	0 75536	-0 51694	0 79070	0 79070	0 79070	0 79070	0 79070	
15	0 38000	0 34750	0 42030	-60 797	0 08230	15	-0 51316	0 63870	-0 44832	0 67563	0 67563	0 67563	0 67563	0 67563	
16	0 41000	0 28087	0 30138	-60 631	0 08567	16	-0 44839	0 52005	0 37985	0 55855	0 55855	0 55855	0 55855	0 55855	
17	0 44000	0 21425	0 18407	-60 137	0 08870	17	-0 38342	0 40027	0 31158	0 44038	0 44038	0 44038	0 44038	0 44038	
18	0 47000	0 14763	0 09664	-59 416	0 09139	18	-0 31820	0 28038	-0 24354	0 32239	0 32239	0 32239	0 32239	0 32239	
19	0 50000	0 08101	0 04129	-58 597	0 09370	19	-0 25272	0 16198	0 17579	0 20615	0 20615	0 20615	0 20615	0 20615	
20	0 53000	0 01439	0 14865	-57 766	0 09564	20	-0 18697	0 04639	-0 10830	0 09289	0 09289	0 09289	0 09289	0 09289	
21	0 56000	0 05223	0 25773	-56 998	0 09717	21	-0 12100	-0 06570	-0 04102	0 01687	0 01687	0 01687	0 01687	0 01687	
22	0 59000	0 11885	0 35398	-56 333	0 09833	22	-0 05484	0 17416	0 02606	0 12315	0 12315	0 12315	0 12315	0 12315	
23	0 62000	0 18547	0 45295	-55 794	0 09913	23	-0 01148	0 28038	-0 24354	0 32239	0 32239	0 32239	0 32239	0 32239	
24	0 65000	0 25209	0 55015	-55 370	0 09943	24	-0 07793	0 38124	0 38124	0 32673	0 32673	0 32673	0 32673	0 32673	
25	0 68000	0 31871	0 64599	-55 032	0 09901	25	-0 04448	0 43082	0 22646	0 42509	0 42509	0 42509	0 42509	0 42509	
26	0 71000	0 38533	0 74074	-54 745	0 09739	26	-0 21118	-0 57841	0 29300	0 52190	0 52190	0 52190	0 52190	0 52190	
27	0 74000	0 45195	0 83450	-54 469	0 09449	27	-0 27814	-0 67436	0 35928	0 61762	0 61762	0 61762	0 61762	0 61762	
28	0 77000	0 51857	0 92732	-54 187	0 08949	28	-0 34557	-0 76884	0 42510	0 71263	0 71263	0 71263	0 71263	0 71263	
29	0 80000	0 58519	-1 01914	-53 885	0 08347	29	-0 41363	-0 86187	0 49028	0 80714	0 80714	0 80714	0 80714	0 80714	
30	0 83000	0 65181	1 10990	-53 546	0 07625	30	-0 48229	0 95350	0 55486	0 90114	0 90114	0 90114	0 90114	0 90114	
31	0 86000	0 71843	-1 1946	-53 152	0 06794	31	-0 55148	-1 04374	0 61891	-0 49454	-0 49454	-0 49454	-0 49454	-0 49454	
32	0 89000	0 78506	1 28762	-52 678	0 05871	32	-0 62115	-1 13256	0 68248	1 08725	1 08725	1 08725	1 08725	1 08725	
33	0 92000	0 85168	-1 37413	-52 107	0 04873	33	-0 69125	-1 21983	0 74562	-1 17909	-1 17909	-1 17909	-1 17909	-1 17909	
34	0 95000	0 91830	1 45874	-51 446	0 03821	34	-0 76171	-1 30542	0 80840	1 26982	1 26982	1 26982	1 26982	1 26982	
35	0 97500	0 97381	-1 52766	-50 839	0 02915	35	-0 83245	-1 38910	0 87090	-1 35917	-1 35917	-1 35917	-1 35917	-1 35917	
36	1 00000	1 02933	1 59506	-50 196	0 01997	36	-0 90336	-1 47065	0 91324	1 44684	1 44684	1 44684	1 44684	1 44684	
37						37	0 96251	1 53666	0 95111	-1 51846	-1 51846	-1 51846	-1 51846	-1 51846	
38						38	1 01487	-1 59112	1 03103	-1 59068	-1 59068	-1 59068	-1 59068	-1 59068	
39						39	1 02120	-1 59740	1 01309	-1 58770	-1 58770	-1 58770	-1 58770	-1 58770	
40						40	1 02933	1 59506	1 02933	-1 59165	-1 59165	-1 59165	-1 59165	-1 59165	

MFRIDIONAL AIRFOIL GEOMETRY - STREAMLINE 4

MEANL. INF. DATA

PT	PCT X	X	Y	B.M.	T (M)	PT	X _S	Y _S	X _P	Y _P
1	0	-1.27111	1.73523	-52.305	0.02007	1	-1.27111	1.73523	-1	27111
2	0.02500	-1.21143	1.65728	-52.811	0.02541	2	-1.27508	1.72801	-1	26298
3	0.05000	-1.15174	1.57790	-53.307	0.03074	3	-1.27314	1.72100	-1	25082
4	0.07500	-1.09206	1.49709	-53.793	0.03603	4	-1.22155	1.64960	-1	20130
5	0.10000	-1.03237	1.41485	-54.271	0.04126	5	-1.16407	1.56872	-1	13942
6	0.12500	-0.97269	1.33115	-54.745	0.04642	6	-1.10659	1.48645	-1	10752
7	0.15000	-0.91301	1.24596	-55.222	0.05149	7	-1.04912	1.40280	-1	01563
8	0.17500	-0.85332	1.15922	-55.723	0.05644	8	-0.99165	1.31775	-0	95374
9	0.20000	-0.79364	1.07077	-56.259	0.06125	9	-0.93415	1.23128	-0	89186
10	0.23000	-0.72202	0.96219	-56.918	0.06683	10	-0.87664	1.14333	-0	83001
11	0.26000	-0.65040	0.85091	-57.537	0.07217	11	-0.81911	1.05376	-0	76817
12	0.29000	-0.57878	0.73720	-58.023	0.07724	12	-0.75002	0.94395	-0	69402
13	0.32000	-0.50716	0.62179	-58.290	0.08201	13	-0.68085	0.83154	-0	61195
14	0.35000	-0.43554	0.50575	-58.293	0.08645	14	-0.61154	0.71674	-0	54602
15	0.38000	-0.36392	0.39038	-57.995	0.09056	15	-0.54204	0.60023	-0	47228
16	0.41000	-0.29230	0.27704	-57.364	0.09429	16	-0.47231	0.48304	-0	39876
17	0.44000	-0.22068	0.16715	-56.401	0.09764	17	-0.40231	0.36638	-0	32552
18	0.47000	-0.14906	0.06159	-55.274	0.10060	18	-0.33200	0.25162	-0	30247
19	0.50000	-0.07744	-0.03969	-54.210	0.10313	19	-0.26134	0.14013	-0	19416
20	0.53000	-0.00582	-0.13738	-53.335	0.10522	20	-0.19040	0.03293	-0	10772
21	0.56000	0.06580	-0.23245	-52.720	0.10685	21	-0.11927	-0.06985	-0	00953
22	0.59000	0.13742	-0.32578	-52.298	0.10808	22	-0.0492	-0.16880	0	03638
23	0.62000	0.20904	-0.41788	-51.971	0.10892	23	0.02329	-0.26481	0	25260
24	0.65000	0.28056	-0.50899	-51.692	0.10911	24	0.09467	-0.35883	0	18018
25	0.68000	0.35228	-0.59923	-51.430	0.10834	25	0.16615	-0.45143	0	25194
26	0.71000	0.42390	-0.68861	-51.152	0.10621	26	0.23785	-0.54281	0	38433
27	0.74000	0.49552	-0.77704	-50.839	0.10240	27	0.3093	-0.63300	-0	47517
28	0.77000	0.56714	-0.86447	-50.511	0.09703	28	0.38255	-0.72191	0	10597
29	0.80000	0.63876	-0.95090	-50.194	0.09026	29	0.45582	-0.80938	0	20009
30	0.83000	0.71038	-1.03637	-49.883	0.08223	30	0.52970	-0.89532	0	53522
31	0.86000	0.78201	-1.12089	-49.561	0.07306	31	0.60409	-0.97979	0	74183
32	0.89000	0.85363	-1.20439	-49.177	0.06292	32	0.67894	-1.06286	-0	46526
33	0.92000	0.92525	-1.28660	-48.681	0.05199	33	0.75420	-1.14459	0	65530
34	0.95000	0.99687	-1.36724	-48.079	0.04049	34	0.82982	-1.22495	0	53522
35	0.97500	1.05655	-1.43306	-47.515	0.03062	35	0.90572	-1.30376	0	74471
36	1.00000	1.11623	1.49755	-46.914	0.02062	36	0.98180	-1.38076	0	83362
						37	1.04526	-1.44340	-0	18382
						38	1.10124	-1.49747	-1	29447
						39	1.10799	-1.50047	1	11717
						40	1.11623	-1.49755	1	11623

MERIDIONAL AIRFOIL GEOMETRY - STRAMI INF 5

PI	MFI INF DATA				SURFACE COORDINATES			
	PCT X	X	Y	B.M	I(M)	XS	YS	YP
1	0	-1 34730	1 69080	-50 948	0 02036	1 -1 34730	1 69080	1 34730
2	0 02500	1 28385	1 61195	-51 401	0 02672	2 -1 35116	1 68336	-1 33909
3	0 05000	-1 22040	1 53183	-51 845	0 03305	3 -1 34905	1 67629	-1 33273
4	0 07500	-1 15695	1 45044	-52 280	0 03933	4 -1 29429	1 60362	-1 27341
5	0 10000	-1 09350	1 36777	-52 706	0 04555	5 -1 23340	1 52162	-1 20741
6	0 12500	-1 03006	1 28383	-53 123	0 05168	6 -1 17251	1 43841	-1 14140
7	0 15000	-0 96661	1 19862	-53 525	0 05769	7 -1 11162	1 35397	-1 07539
8	0 17500	-0 90316	1 11210	-53 956	0 06355	8 -1 05073	1 26832	-1 00939
9	0 20000	-0 83971	1 02421	-54 393	0 06926	9 -0 98981	1 18147	-0 94341
10	0 23000	-0 76358	0 91687	-54 902	0 07586	10 -0 92886	1 09340	-0 87747
11	0 26000	-0 68744	0 80762	-55 328	0 08216	11 -0 86787	1 00405	-0 81156
12	0 29000	-0 61130	0 69699	-55 558	0 08813	12 -0 79461	0 89506	-0 73254
13	0 32000	-0 53517	0 58600	-55 491	0 09373	13 -0 72123	0 78426	-0 65366
14	0 35000	0 45903	0 47590	-55 130	0 09894	14 -0 64764	0 67207	-0 57496
15	0 38000	-0 38289	0 36783	-54 497	0 10373	15 -0 57379	0 55945	-0 49655
16	0 41000	-0 30675	0 26272	-53 633	0 10807	16 -0 49962	0 44762	-0 41844
17	0 44000	-0 23062	0 16121	-52 597	0 11193	17 -0 42512	0 33771	-0 34067
18	0 47000	-0 15448	0 06355	-51 526	0 11530	18 -0 35027	0 23057	-0 26324
19	0 50000	-0 07834	0 03060	-50 577	0 11818	19 -0 27508	0 12722	-0 18616
20	0 53000	-0 00221	0 12194	-49 829	0 12052	20 -0 19962	0 02768	-0 10935
21	0 56000	0 07393	-0 21125	-49 308	0 12228	21 -0 12393	0 06813	-0 03270
22	0 59000	0 15007	-0 29917	-48 924	0 12358	22 -0 04825	0 16081	-0 04384
23	0 62000	0 22620	-0 38598	-48 570	0 12442	23 -0 02757	0 25111	-0 12029
24	0 65000	0 30234	0 47172	-48 222	0 12428	24 -0 10349	0 33977	-0 17138
25	0 68000	0 37848	-0 55614	-47 883	0 12297	25 -0 17956	0 42715	-0 25857
26	0 71000	0 45462	-0 64016	-47 558	0 11992	26 -0 25596	0 51316	-0 34481
27	0 74000	0 53075	-0 72247	-47 249	0 11510	27 -0 33287	0 59767	-0 42409
28	0 77000	0 60689	0 80489	-46 941	0 10861	28 -0 41037	0 68063	-0 59970
29	0 80000	0 68303	-0 88591	-46 615	0 10062	29 -0 48849	0 76204	-0 57301
30	0 83000	0 75916	-0 96598	-46 261	0 09125	30 -0 56721	0 84197	-0 64557
31	0 86000	0 83530	-1 04502	-45 873	0 08065	31 -0 64646	0 92047	-0 71959
32	0 89000	0 91144	-1 12294	-45 452	0 06901	32 -0 72620	0 99753	-0 79213
33	0 92000	0 98758	-1 19969	-45 001	0 05654	33 -0 80636	0 07309	-0 86425
34	0 95000	1 06371	-1 27521	-44 524	0 04346	34 -0 88685	0 14715	-0 93603
35	0 97500	1 12716	-1 33716	-44 110	0 03224	35 -0 96758	0 21968	-0 10757
36	1 00000	1 19061	-1 39822	-43 686	0 02887	36 -1 04848	0 29070	-1 07895
						37 -1 11594	-1 34873	-1 13838
						38 -1 17540	-1 39905	-1 19016
						39 -1 18245	-1 40166	-1 19368
						40 -1 19061	-1 39822	-1 19061

MFRIDIONAL AIRFOIL GEOMETRY - SURFAMI INF 6

MEAN LINE DATA

PT	PCT X	X	Y	Z	B.M.	T (M)	PT	xS	yS	zS	XP	YP
1	0	-1.41947	1.62753	-49.775	0.02052	1	-1.41947	1.62753	-1.41947	1	62753	
2	0.02500	-1.35251	1.54776	50.208	0.02856	2	-1.42324	1.61994	-1.41123	1	62997	
3	0.05000	-1.28555	1.46676	-50.631	0.03654	3	-1.42100	1.61283	1.40471	1	62658	
4	0.07500	-1.38455	51.043	0.04447	4	-1.36348	1.53R62	1.34154	1	55689		
5	0.10000	-1.15164	1.30113	51.444	0.05230	5	-1.29958	1.45517	-1.27143	1	47R35	
6	0.12500	-1.08468	1.21654	-51.828	0.06002	6	-1.23588	1.37057	-1.20131	1	39853	
7	0.15000	-1.01772	1.13080	-52.194	0.06758	7	-1.17209	1.28483	-1.13119	1	31743	
8	0.17500	-0.95076	1.04394	-52.546	0.07495	8	-1.10827	1.19799	-1.06109	1	23509	
9	0.20000	-0.88380	0.95599	-52.879	0.08211	9	-1.04442	1.11008	-0.99102	1	15151	
10	0.23000	-0.80345	0.89116	-53.210	0.09038	10	-0.98051	1.02115	-0.92101	1	06673	
11	0.26000	-0.72310	0.74130	-53.381	0.09825	11	-0.91654	0.93122	-0.85107	0	98077	
12	0.29000	-0.64275	0.63334	-53.234	0.10568	12	-0.83964	0.82209	-0.76726	0	87622	
13	0.32000	-0.56240	0.52682	-52.635	0.11263	13	-0.76253	0.71200	-0.68367	0	77060	
14	0.35000	-0.48205	0.42331	-51.673	0.11906	14	-0.68508	0.60171	-0.60042	0	66496	
15	0.38000	-0.40170	0.32372	-50.511	0.12443	15	-0.60716	0.49264	-0.51764	0	56100	
16	0.41000	-0.32135	0.22821	-49.354	0.13022	16	-0.52875	0.38639	-0.43535	0	46022	
17	0.44000	-0.24100	0.13624	-48.398	0.13489	17	-0.44991	0.28399	-0.35344	0	36344	
18	0.47000	-0.16065	0.04698	-47.639	0.13891	18	-0.37075	0.18580	-0.27195	0	27062	
19	0.50000	-0.08030	-0.04014	-47.013	0.14228	19	-0.29143	0.09146	-0.19057	0	18102	
20	0.53000	0.00005	0.12553	-46.485	0.14495	20	0.21197	0.00018	-0.10933	0	09371	
21	0.56000	-0.08040	-0.20948	-46.027	0.14689	21	-0.13234	-0.08R65	-0.02826	0	00836	
22	0.59000	0.16075	-0.29215	-45.608	0.14821	22	-0.05251	-0.17544	-0.05261	0	07563	
23	0.62000	0.24110	-0.37364	-45.202	0.14886	23	0.02755	-0.26047	0.13226	0	15848	
24	0.65000	0.32145	-0.45400	-44.802	0.14825	24	0.10780	-0.34399	0.21371	0	24031	
25	0.68000	0.40180	0.53325	44.408	0.14573	25	0.18829	-0.42609	0.29392	0	32120	
26	0.71000	0.48216	-0.61142	-44.021	0.14123	26	0.26922	-0.50659	0.37369	-0.40140	0	
27	0.74000	0.56251	0.68856	-43.642	0.13479	27	0.35R81	-0.58530	0.45279	-0.48119	0	
28	0.77000	0.64286	-0.76458	-43.258	0.12654	28	0.43308	-0.66220	0.53123	-0.56065	0	
29	0.80000	0.72321	-0.83977	-42.857	0.11658	29	0.51599	-0.73733	0.60902	-0.63979	0	
30	0.83000	0.80356	0.91378	-42.436	0.10509	30	0.59950	-0.81076	0.68621	-0.71860	0	
31	0.86000	0.88391	-0.98668	-41.996	0.09226	31	0.68356	-0.88249	0.76285	-0.79704	0	
32	0.89000	0.96426	-1.05845	-41.549	0.07829	32	0.76810	-0.95256	0.83901	-0.87500	0	
33	0.92000	1.04611	-1.12911	-41.104	0.06340	33	0.85304	-1.02097	0.91477	-0.95240	0	
34	0.95000	1.12496	1.19867	-40.664	0.04782	34	0.93829	-1.08775	0.99022	-1.02916	0	
35	0.97500	1.19192	-1.25582	-40.297	0.03451	35	1.02377	-1.15299	1.06545	-1.10522	0	
36	1.00000	1.25888	-1.31223	-39.929	0.02105	36	1.10938	-1.21681	1.14054	-1.18053	0	
37						37	1.18076	-1.26898	1.20308	-1.24266	0	
38						38	1.24355	-1.31418	1.25811	-1.29682	0	
39						39	1.25087	-1.31628	1.26147	-1.30377	0	
40						40	1.25R88	-1.31223	1.25R88	-1.31223	0	

PIVOT III ROTOR

MFRIDIONAL AIRFOIL GEOMETRY - STREAMLINF 7

NB 20

MEAN INF DATA

PT	PCT X	X	Y	Z	B.M.	T(M)	P	X5	YS	XP	YP
1	0.	-1 49130	1 54450		-48.731	0.02021	1	-1 49130	1 54450	-1 49130	1.54450
2	0 02500	-1 42094	1 46373		-49 149	0.03057	2	-1 49491	1 53692	-1 48320	1.54707
3	0 05000	-1 35059	1 38178		-49 545	0.04089	3	-1 49362	1 52992	-1 47668	1.54389
4	0 07500	-1 28023	1 29874		-49 905	0.05113	4	-1 43251	1 45373	-1 40938	1.47372
5	0 10000	-1 20987	1 21470		-50 214	0.06125	5	-1 36614	1 36852	-1 33503	1.39505
6	0 12500	-1 13952	1 12982		-50 463	0.07120	6	-1 29979	1 28227	-1 26067	1.31520
7	0 15000	-1 06916	1 04429		-50 643	0.08094	7	-1 23341	1 19510	-1 18634	1.23429
8	0 17500	-0 99880	0 95832		-50 748	0.09042	8	-1 16697	1 10715	-1 11206	1.15248
9	0 20000	-0 92845	0 87215		-50 773	0 09961	9	-1 10045	1 01862	-1 03787	1.06995
10	0 23000	-0 84402	0 76846		-50 677	0 11022	10	-1 03381	0 92472	-0 96379	0.98693
11	0 26000	-0 75959	0 66623		-50 409	0 12079	11	-0 96703	0 84066	-0 88986	0.91365
12	0 29000	-0 67516	0 56497		-49 901	0 12977	12	-0 88665	0 73394	-0 81139	0.80379
13	0 32000	-0 59073	0 46603		-49 099	0 13858	13	-0 80594	0 62790	-0 71324	0.70456
14	0 35000	-0 50630	0 37026		-48 069	0 14670	14	-0 72479	0 52318	-0 62553	0.60676
15	0 38000	-0 42188	0 27812		-46 921	0 15408	15	-0 64311	0 42066	-0 53836	0.51140
16	0 41000	-0 33745	0 18962		-45 790	0 16067	16	-0 56087	0 32125	-0 45173	0.41928
17	0 44000	-0 25302	0 10434		-44 804	0 16645	17	-0 47815	0 22550	-0 36560	0.33074
18	0 47000	-0 16859	0 02174		-43 964	0 17137	18	-0 39503	0 13360	-0 27986	0.24563
19	0 50000	-0 08416	0 05864		-43 235	0 17541	19	-0 31166	0 04530	-0 19437	0.16339
20	0 53000	0 00027	-0 13712		-42 596	0 17854	20	-0 22807	0 03993	-0 10911	0.08341
21	0 56000	0 08470	-0 21397		-42 032	0 18073	21	-0 14424	0 12254	-0 02409	0.00526
22	0 59000	0 18090	-0 57919		-41 525	0 18199	22	-0 06015	0 20283	-0 06069	0.07141
23	0 62000	0 25355	-0 36353		-41 059	0 18222	23	-0 02419	0 1360	-0 04520	0.14685
24	0 65000	0 33798	-0 43651		-40 619	0 18070	24	-0 10880	0 35751	-0 22945	0.22126
25	0 68000	0 42241	-0 50838		-40. 195	0 17676	25	-0 19371	0 43223	-0 31339	0.29483
26	0 71000	0 50684	-0 59684		-39 778	0 17041	26	-0 27916	-0 50509	-0 39680	0.36793
27	0 74000	0 59127	-0 64895		-39 359	0 16186	27	-0 47945	-0 57588	-0 4087	0.4087
28	0 77000	0 67569	-0 72553		-38 939	0 15122	28	-0 45232	-0 61467	-0 51370	0.51370
29	0 80000	0 76012	-0 78540		-38 518	0 13860	29	-0 53994	-0 71153	-0 64259	0.58638
30	0 83000	0 84455	-0 85209		-38 096	0 12422	30	-0 62817	-0 77650	-0 7321	0.65988
31	0 86000	0 92898	-0 91778		-37 670	0 10832	31	-0 71696	-0 83962	-0 80328	0.73118
32	0 89000	1 01341	-0 98246		-37 237	0 09111	32	-0 80623	-0 90097	-0 88287	0.80321
33	0 92000	1 09784	-1 04612		-36 791	0 07284	33	-0 96065	-0 96065	-0 96208	0.87491
34	0 95000	1 18226	-1 10874		-36 334	0 05379	34	-0 98584	-1 01873	-1 04097	0.94619
35	0 97500	1 25262	-1 16012		-35 944	0 03755	35	-1 07602	-1 07528	-1 01695	1.01695
36	1 00000	1 32298	-1 21076		-35 549	0 02115	36	-1 16633	-1 13040	-1 08707	1.08707
							37	-1 24160	-1 17532	-1 26364	-1.14491
							38	-1 30768	-1 21405	-1 32115	-1.19523
							39	-1 31521	-1 21550	-1 32497	-1.20202
							40	-1 32298	-1 21076	-1.32298	1.21076

PHASE III ROTOR

NB 20

MERIDIONAL AIRFOIL GEOMETRY - STREAMLINE 8

MEANLINE DATA						SURFACE COORDINATES					
PT	PC1	X	Y	B.M.	T(M)	PT	X5	YS	XP	YP	
1	0.	1.56132	1.44158	-47.655	0.01929	1	1.56132	1.44158	-1	56132	1.44158
2	0.02500	-1.48769	1.36041	-47.916	0.03220	2	-1.56466	1.43423	-1	55362	1.44421
3	0.05000	1.41406	1.27854	-48.150	0.04506	3	-1.56238	1.42758	-1	54728	1.44134
4	0.07500	1.34043	1.19605	-48.333	0.05782	4	-1.49964	1.34962	-1	47574	1.37120
5	0.10000	1.26681	1.11314	-48.443	0.07041	5	-1.43085	1.26350	-1	39728	1.29357
6	0.12500	-1.19318	1.03003	-48.464	0.08279	6	-1.36203	1.17683	-1	31884	1.21527
7	0.15000	-1.11955	0.94701	-48.385	0.09488	7	-1.29315	1.08978	-1	24046	1.13649
8	0.17500	-1.04592	0.86436	-48.203	0.10664	8	-1.22416	1.00259	-1	16219	1.05748
9	0.20000	-0.97229	0.78240	-47.913	0.11803	9	-1.15502	0.91550	-1	08408	0.97851
10	0.23000	-0.88394	0.68537	-47.414	0.13112	10	-1.08567	0.82882	-1	00617	0.89990
11	0.26000	-0.79559	0.59032	-46.736	0.14350	11	-1.01609	0.74284	-0	92850	0.82195
12	0.29000	-0.70723	0.49782	-45.848	0.15511	12	-0.93221	0.64101	-0	83567	0.72974
13	0.32000	-0.61898	0.40851	-44.731	0.16588	13	-0.84784	0.54114	-0	74334	0.63949
14	0.35000	-0.53053	0.32287	-43.466	0.17574	14	-0.76288	0.44380	-0	65159	0.55184
15	0.38000	-0.44217	0.24100	-42.175	0.18463	15	-0.67725	0.34950	-0	56051	0.46743
16	0.41000	-0.35382	0.16263	-40.996	0.19252	16	-0.59098	0.25909	-0	47008	0.38664
17	0.44000	-0.26547	0.08716	-40.044	0.19937	17	-0.50416	0.17258	-0	38019	0.30941
18	0.47000	-0.17711	0.01394	-39.281	0.20510	18	-0.41697	0.08997	-0	29067	0.23528
19	0.50000	-0.08876	-0.05747	-38.624	0.20968	19	-0.32960	0.01085	-0	20133	0.16348
20	0.53000	-0.00041	-0.12731	-38.027	0.21312	20	-0.24204	0.06544	-0	11219	0.09332
21	0.56000	0.08795	-0.19569	-37.457	0.21543	21	-0.15420	-0.13938	-0	02332	0.02443
22	0.59000	0.17630	-0.26271	-36.905	0.21642	22	-0.06605	-0.21125	-0	04337	0.04337
23	0.62000	-0.26465	-0.32841	-36.369	0.21580	23	0.02244	-0.2120	0	15345	0.11019
24	0.65000	0.35301	-0.39285	-35.844	0.21293	24	0.11132	-0.34924	0	24128	0.17618
25	0.68000	0.44136	-0.45608	-35.329	0.20725	25	0.20067	-0.41529	0	32863	0.24153
26	0.71000	0.52971	-0.51812	-34.824	0.19887	26	0.29066	-0.47916	0	41535	0.30655
27	0.74000	0.61807	0.57901	-34.327	0.18804	27	0.38144	-0.54062	0	50128	0.37153
28	0.77000	0.70642	-0.63879	-33.832	0.17491	28	0.47293	-0.5974	0	58650	0.43649
29	0.80000	0.79477	-0.69744	-33.327	0.15961	29	0.56505	-0.65666	0	67109	0.50137
30	0.83000	0.88313	-0.75498	-32.813	0.14238	30	0.65773	-0.71143	0	75511	0.56614
31	0.86000	0.97148	-0.81138	-32.292	0.12347	31	0.75092	-0.76413	0	83062	0.63076
32	0.89000	1.05983	-0.86666	-31.772	0.10314	32	0.84455	-0.81481	0	92170	0.69515
33	0.92000	1.14819	0.92084	-31.263	0.08165	33	0.93850	-0.86357	1	10446	0.75920
34	0.95000	1.23654	-0.97395	-30.763	0.05934	34	1.03268	-0.91050	1	08699	0.82282
35	0.97500	1.31017	-1.01742	-30.349	0.04034	35	1.12700	-0.95574	1	16937	0.88594
36	1.00000	1.38380	-1.06017	-29.936	0.02117	36	1.22136	-0.99945	1	25171	0.94846
37						37	1.2998	-1.03493	1	32036	1.00001
38						38	1.1680	-1.05512	1	38057	1.04473
39						39	1.37649	-1.06573	1	38499	1.05121
40						40	1.38380	-1.06017	1	38380	1.06017

MERIDIONAL AIRFOIL GEOMETRY - SURFACE INF 9

MF AND INF DATA				SURFACE COORDINATES			
PT	PCI X	Y	Z	PT	XS	YS	ZP
1	0	-1.61594	1	30460	-46 328	0 01883	-1 61594
2	0 02500	-1.53918	1	22427	-46 271	0 03432	-1 61906
3	0 05000	-1.46242	1	14412	-46 197	0 04475	-1 61674
4	0 07500	-1.38566	1	06422	-46 093	0 06504	-1 55158
5	0 10000	-1 30889	0	98467	-45 943	0 08011	-1 48037
6	0 12500	-1 23213	0	90562	-45 725	0 09489	-1 40309
7	0 15000	-1 15537	0	82730	-45 411	0 10932	-1 33768
8	0 17500	-1 07861	0	75002	-44 946	0 12333	-1 26611
9	0 20000	-1 00185	0	67424	-44 277	0 13686	-1 19430
10	0 23000	-0 90974	0	58601	-43 207	0 15236	-1 12217
11	0 26000	-0 81762	0	50141	-41 889	0 16697	-1 04962
12	0 29000	-0 72551	0	42080	-40 489	0 18060	-0 96189
13	0 32000	-0 63340	0	34397	-39 195	0 19319	-0 87336
14	0 35000	-0 54128	0	27041	-38 046	0 20467	-0 78414
15	0 38000	-0 44917	0	19965	-37 031	0 21494	-0 69444
16	0 41000	-0 35706	0	13133	-36 107	0 22395	-0 60435
17	0 44000	-0 26494	0	06522	-35 230	0 23165	-0 51389
18	0 47000	-0 17283	0	00119	-34 383	0 23796	-0 42304
19	0 50000	-0 08072	0	06088	-33 561	0 24283	-0 33176
20	0 53000	-0 01140	0	12106	-32 761	0 24634	-0 24002
21	0 56000	-0 10351	0	17945	-31 986	0 24850	-0 14784
22	0 59000	-0 19562	0	23616	-31 255	0 24888	-0 05526
23	0 62000	-0 28774	0	29132	-30 585	0 24693	-0 03769
24	0 65000	-0 37985	0	34509	-29 967	0 24228	-0 13106
25	0 68000	-0 47196	0	39757	-29 384	0 23466	-0 22492
26	0 71000	-0 56408	0	44884	-28 817	0 22420	-0 03934
27	0 74000	-0 65619	0	49892	-28 247	0 21112	-0 41439
28	0 77000	-0 74830	0	54782	-27 675	0 19560	-0 51004
29	0 80000	-0 84042	0	59554	-27 100	0 17780	-0 60623
30	0 83000	-0 93253	0	64210	-26 524	0 15797	-0 70288
31	0 86000	-0 102464	0	68749	-25 941	0 13639	-0 79934
32	0 89000	-1 14676	0	73172	-25 348	0 11333	-0 89726
33	0 92000	-1 20887	0	77476	-24 737	0 08908	-0 99481
34	0 95000	-1 30098	-C	81659	-24 109	0 06397	-1 09250
35	0 97500	-1 37774	0	85051	-23 574	0 04263	-1 19023
36	1 00000	-1 45451	0	88158	-23 030	0 02111	-1 28792
					37	0 07227	-0 87005
					38	1 44020	-0 89043
					39	1 44789	-0 84003
					40	1 45451	0 88358

MEANLINF DATA

PT	PCT X	X	Y	B+M	T (M)	XS	YS	XP	YP
1	0	-1.61706	1.12360	-44.267	0.02119	-1.61706	1.12360	-1.61706	1.12360
2	0.02500	-1.53770	1.04648	-44.092	0.03909	-1.62031	1.11518	-1.60880	1.12707
3	0.05000	-1.45834	0.96988	-43.870	0.05690	-1.61748	1.10803	-1.60151	1.12443
4	0.07500	-1.37899	0.89399	-43.548	0.07453	-1.55130	1.03244	-1.52410	1.06051
5	0.10000	-1.29963	0.81914	-43.075	0.09187	-1.47506	0.94937	-1.43863	0.99039
6	0.12500	-1.22027	0.74574	-42.428	0.10883	-1.40466	0.86698	-1.35331	0.92100
7	0.15000	-1.14092	0.67421	-41.603	0.12535	-1.33100	0.78559	-1.26826	0.85269
8	0.17500	-1.06156	0.60491	-40.630	0.14135	-1.25699	0.70557	-1.18356	0.78591
9	0.20000	-0.98220	0.53809	-39.550	0.15675	-1.18253	0.62734	-1.09930	0.72103
10	0.23000	-0.88697	0.46135	-38.162	0.17433	-1.10758	0.55128	-1.01554	0.65855
11	0.26000	-0.79175	0.38841	-36.732	0.19083	-1.03211	0.47766	-0.93230	0.59852
12	0.29000	-0.69652	0.31914	-35.343	0.20616	-0.94083	0.39281	-0.83312	0.52989
13	0.32000	-0.60129	0.25321	-34.069	0.22024	-0.84881	0.31194	-0.73468	0.46488
14	0.35000	-0.50606	0.19022	-32.908	0.23226	-0.75615	0.23506	-0.63689	0.40322
15	0.38000	-0.41083	0.14987	-31.834	0.24424	-0.66298	0.16199	-0.53960	0.34443
16	0.41000	-0.31561	0.07193	-30.804	0.25400	-0.56935	0.09243	-0.44278	0.28801
17	0.44000	-0.22038	0.01629	-29.781	0.26220	-0.47525	0.02612	-0.34642	0.23362
18	0.47000	-0.12515	-0.03709	-28.770	0.26874	-0.38064	0.03716	-0.25057	0.18101
19	0.50000	-0.02992	-0.08831	-27.787	0.27356	-0.28549	0.09749	-0.15526	0.13008
20	0.53000	0.06531	-0.13749	-26.841	0.27676	-0.18982	0.15487	0.06648	0.08059
21	0.56000	0.16053	-0.18473	-25.935	0.27831	-0.09369	0.20912	0.03384	0.03226
22	0.59000	0.25576	-0.23015	-25.071	0.27759	0.00283	0.26096	0.12779	0.01402
23	0.62000	0.35099	-0.27386	-24.250	0.27399	0.09668	-0.30987	0.22139	-0.05959
24	0.65000	0.44622	-0.31597	-23.463	0.26736	0.19695	-0.35587	0.31458	0.10443
25	0.68000	0.54145	-0.35655	-22.700	0.25772	0.29472	0.39777	0.40726	-0.14896
26	0.71000	0.63667	-0.39565	-21.946	0.24522	0.39299	0.43860	0.49144	-0.19335
27	0.74000	0.73190	-0.43330	-21.191	0.23001	0.49172	0.47543	0.59117	-0.23768
28	0.77000	0.82713	-0.46949	-20.426	0.21231	0.59085	0.50938	0.68250	-0.28193
29	0.80000	0.92236	-0.50423	-19.651	0.19233	0.69033	0.54053	0.77348	-0.32607
30	0.83000	0.95000	0.10759	-0.53749	-18.853	0.17031	0.79008	0.56897	0.86418
31	0.86000	0.97500	1.47786	-0.67608	-14.480	0.04441	0.89002	0.59479	0.95470
32	1.00000	1.55721	0.69595	-13.638	0.02108	0.99007	0.61808	1.04510	-0.45691
33						0.99007	0.61808	1.04510	-0.45691
34						0.99007	0.61808	1.04510	-0.45691
35						0.99007	0.61808	1.04510	-0.45691
36						0.99007	0.61808	1.04510	-0.45691

SURFACE COORDINATES									
PT	PCT X	X	Y	Z	U	V	W	U'	V'
1	0	-1.61706	1.12360	0.02119	-1.61706	1.12360	0.02119	-1.61706	1.12360
2	0.02500	-1.53770	1.04648	-44.092	0.03909	-1.62031	1.11518	-1.60880	1.12707
3	0.05000	-1.45834	0.96988	-43.870	0.05690	-1.61748	1.10803	-1.60151	1.12443
4	0.07500	-1.37899	0.89399	-43.548	0.07453	-1.55130	1.03244	-1.52410	1.06051
5	0.10000	-1.29963	0.81914	-43.075	0.09187	-1.47506	0.94937	-1.43863	0.99039
6	0.12500	-1.22027	0.74574	-42.428	0.10883	-1.40466	0.86698	-1.35331	0.92100
7	0.15000	-1.14092	0.67421	-41.603	0.12535	-1.33100	0.78559	-1.26826	0.85269
8	0.17500	-1.06156	0.60491	-40.630	0.14135	-1.25699	0.70557	-1.18356	0.78591
9	0.20000	-0.98220	0.53809	-39.550	0.15675	-1.18253	0.62734	-1.09930	0.72103
10	0.23000	-0.88697	0.46135	-38.162	0.17433	-1.10758	0.55128	-1.01554	0.65855
11	0.26000	-0.79175	0.38841	-36.732	0.19083	-1.03211	0.47766	-0.93230	0.59852
12	0.29000	-0.69652	0.31914	-35.343	0.20616	-0.94083	0.39281	-0.83312	0.52989
13	0.32000	-0.60129	0.25321	-34.069	0.22024	-0.84881	0.31194	-0.73468	0.46488
14	0.35000	-0.50606	0.19022	-32.908	0.23226	-0.75615	0.23506	-0.63689	0.40322
15	0.38000	-0.41083	0.14987	-31.834	0.24424	-0.66298	0.16199	-0.53960	0.34443
16	0.41000	-0.31561	0.07193	-30.804	0.25400	-0.56935	0.09243	-0.44278	0.28801
17	0.44000	-0.22038	0.01629	-29.781	0.26220	-0.47525	0.02612	-0.34642	0.23362
18	0.47000	-0.12515	-0.03709	-28.770	0.26874	-0.38064	0.03716	-0.25057	0.18101
19	0.50000	-0.02992	-0.08831	-27.787	0.27356	-0.28549	0.09749	-0.15526	0.13008
20	0.53000	0.06531	-0.13749	-26.841	0.27676	-0.18982	0.15487	0.06648	0.08059
21	0.56000	0.16053	-0.18473	-25.935	0.27831	-0.09369	0.20912	0.03384	0.03226
22	0.59000	0.25576	-0.23015	-25.071	0.27759	0.00283	0.26096	0.12779	0.01402
23	0.62000	0.35099	-0.27386	-24.250	0.27399	0.09668	-0.30987	0.22139	-0.05959
24	0.65000	0.44622	-0.31597	-23.463	0.26736	0.19695	-0.35587	0.31458	0.10443
25	0.68000	0.54145	-0.35655	-22.700	0.25772	0.29472	0.39777	0.40726	-0.14896
26	0.71000	0.63667	-0.39565	-21.946	0.24522	0.39299	0.43860	0.49144	-0.19335
27	0.74000	0.73190	-0.43330	-21.191	0.23001	0.49172	0.47543	0.59117	-0.23768
28	0.77000	0.82713	-0.46949	-20.426	0.21231	0.59085	0.50938	0.68250	-0.28193
29	0.80000	0.92236	-0.50423	-19.651	0.19233	0.69033	0.54053	0.77348	-0.32607
30	0.83000	0.95000	1.01759	-0.53749	-18.853	0.17031	0.79008	0.56897	0.86418
31	0.86000	0.97500	1.47786	-0.67608	-14.480	0.04441	0.89002	0.59479	0.95470
32	1.00000	1.55721	0.69595	-13.638	0.02108	0.99007	0.61808	1.04510	-0.45691

MFRIDIONAL AIRFOIL GEOMFIRY STREAMLINE 11

MEAN INF DATA

PT	PCT	X	Y	B.M.	T(M)	P1	X _S	Y _S	P2	X _P	Y _P
1	0.	-1.60872	0.91169	-40.880	0.02619	1	-1.60872	0.91169	-1.60872	0.91169	0.91169
2	0.02500	-1.52727	0.84170	-40.460	0.04472	2	-1.61214	0.90100	-1.59882	0.91660	0.91660
3	0.05000	-1.44581	0.77279	-39.990	0.06315	3	-1.60815	0.89240	-1.58957	0.91391	0.91391
4	0.07500	-1.36436	0.70512	-39.421	0.08140	4	-1.54178	0.82468	-1.51276	0.85871	0.85871
5	0.10000	-1.28291	0.63878	-38.705	0.09934	5	-1.46611	0.74859	-1.42552	0.79698	0.79698
6	0.12500	-1.20145	0.57469	-37.827	0.11689	6	-1.39020	0.67368	-1.33852	0.73656	0.73656
7	0.15000	-1.12000	0.51260	-36.785	0.13396	7	-1.31396	0.60022	-1.25185	0.67774	0.67774
8	0.17500	-1.03854	0.45296	-35.623	0.15049	8	-1.23729	0.52853	-1.16561	0.62086	0.62086
9	0.20000	-0.95709	0.39589	-34.394	0.16646	9	-1.16010	0.45896	-1.07989	0.56624	0.56624
10	0.23000	-0.85934	0.33086	-32.873	0.18478	10	-1.08237	0.39179	-0.99471	0.51412	0.51412
11	0.26000	-0.76160	0.26952	-31.341	0.20206	11	-1.00410	0.32721	-0.91007	0.46457	0.46457
12	0.29000	-0.66385	0.21176	-29.815	0.21823	12	-0.90949	0.25326	-0.80920	0.40845	0.40845
13	0.32000	-0.56611	0.15745	-28.302	0.23318	13	-0.81415	0.18323	-0.70905	0.35581	0.35581
14	0.35000	-0.46836	0.10645	-26.809	0.24684	14	-0.71810	0.11709	-0.60960	0.30643	0.30643
15	0.38000	-0.37061	0.05861	-25.342	0.25915	15	-0.62135	0.05480	-0.51083	0.26010	0.26010
16	0.41000	-0.27287	0.01381	-23.906	0.27002	16	-0.52403	0.03371	-0.41269	0.21660	0.21660
17	0.44000	-0.17512	-0.02809	-22.509	0.27938	17	-0.42608	0.05849	-0.31515	0.17572	0.17572
18	0.47000	-0.07738	-0.06726	-21.184	0.28719	18	-0.32758	0.10961	-0.21816	0.13724	0.13724
19	0.50000	0.02037	-0.10393	-19.956	0.29340	19	-0.22860	0.15714	-0.12165	0.10096	0.10096
20	0.53000	0.13832	-0.11811	-18.820	0.29803	20	-0.12927	0.20115	-0.07225	0.06663	0.06663
21	0.56000	0.21586	-0.17061	-17.752	0.30103	21	-0.02970	0.24182	-0.07044	0.03396	0.03396
22	0.59000	0.31360	-0.20091	-16.682	0.30218	22	0.07004	0.27936	0.16618	0.00273	0.00273
23	0.62000	0.41135	-0.22916	-15.557	0.30102	23	0.16997	0.31376	0.26175	-0.02726	-0.02726
24	0.65000	0.50909	-0.25531	-14.391	0.29652	24	0.27023	0.34564	0.35697	0.05618	0.05618
25	0.68000	0.60684	-0.27933	-13.213	0.28791	25	0.37096	0.37416	0.45171	0.08417	0.08417
26	0.71000	0.70458	-0.30123	-12.053	0.27547	26	0.47225	0.39892	0.54594	0.11171	0.11171
27	0.74000	0.80233	-0.32110	-10.929	0.25960	27	0.57394	0.41947	0.63974	0.17918	0.17918
28	0.77000	0.90008	-0.33898	-9.812	0.24050	28	0.67582	0.43593	0.73335	0.16653	0.16653
29	0.80000	0.99782	-0.35489	-8.671	0.21142	29	0.77772	0.44854	0.82694	0.19365	0.19365
30	0.83000	1.09557	0.36878	-7.496	0.19369	30	0.87958	0.45748	0.92057	0.22049	0.22049
31	0.86000	1.19331	-0.38059	-6.279	0.16672	31	0.98136	0.46285	1.01429	0.24693	0.24693
32	0.89000	1.29106	-0.39027	-5.023	0.13784	32	1.08293	0.46480	1.10820	0.27276	0.27276
33	0.92000	1.38880	-0.39776	-3.731	0.10741	33	1.18420	0.46345	1.20243	0.29774	0.29774
34	0.95000	1.48655	-0.40300	-2.396	0.07581	34	1.28502	0.45892	1.29709	0.32162	0.32162
35	0.97500	1.56800	-0.40559	-1.249	0.04886	35	1.38531	0.45135	1.39230	0.34417	0.34417
36	1.00000	1.64946	-0.41654	-0.084	0.02162	36	1.48496	0.44087	1.48813	0.36512	0.36512
						37	1.56747	0.43001	1.56853	0.38117	0.38117
						38	1.63874	0.41911	1.63885	0.39392	0.39392
						39	1.64572	0.41553	1.64607	0.39784	0.39784
						40	1.64946	0.40654	1.64946	0.40654	0.40654

MERIDIONAL AIRFOIL GEOMETRY - STREAMLINE 12

MEANLINE DATA				SURFACE COORDINATES			
PT	PCT X	X	Y	B.M.	T (M)	P1	XP
1	0.	-1.63314	0.79810	-38.150	0.03473	1	-1.63314
2	0.02500	-1.55063	0.73402	-37.504	0.05686	2	-1.63706
3	0.05000	-1.46811	0.67148	-3.112	0.07884	3	-1.63137
4	0.07500	-1.38560	0.61056	-36.050	0.10050	4	-1.56793
5	0.10000	-1.30309	0.55141	-35.199	0.12169	5	-1.49173
6	0.12500	-1.22058	0.49421	-34.247	0.14227	6	-1.41518
7	0.15000	-1.13807	0.43912	-33.192	0.16213	7	-1.33816
8	0.17500	-1.05556	0.38629	-32.056	0.18119	8	-1.26061
9	0.20000	-0.97305	0.33580	-30.864	0.19938	9	-1.18245
10	0.23000	-0.87404	0.27833	-29.391	0.21993	10	-1.10364
11	0.26000	-0.77502	0.22424	-27.898	0.23892	11	-1.02419
12	0.29000	-0.67601	0.17346	-26.404	0.25620	12	-0.92800
13	0.32000	-0.57700	0.12589	-24.924	0.27162	13	-0.83092
14	0.35000	-0.47798	0.08140	-23.464	0.28505	14	-0.73297
15	0.38000	-0.37897	0.03989	-22.026	0.29640	15	-0.63423
16	0.41000	-0.27996	0.00125	-20.611	0.30559	16	-0.53473
17	0.44000	-0.18094	-0.03462	-19.223	0.31255	17	-0.43455
18	0.47000	-0.08193	-0.06784	-17.884	0.31728	18	-0.33374
19	0.50000	0.01708	-0.09857	-16.608	0.31973	19	-0.23240
20	0.53000	0.11610	-0.12696	-15.391	0.31975	20	-0.13065
21	0.56000	0.21511	-0.15312	-14.211	0.31718	21	-0.02861
22	0.59000	0.31412	-0.17711	-13.021	0.31195	22	-0.07367
23	0.62000	0.41314	-0.19889	-11.791	0.30403	23	0.07618
24	0.65000	0.51215	-0.21844	-10.544	0.29356	24	0.27598
25	0.68000	0.61116	-0.23576	-9.309	0.28071	25	0.38208
26	0.71000	0.71018	-0.25093	-8.114	0.26562	26	0.48529
27	0.74000	0.80919	-0.26401	-6.940	0.24847	27	0.58846
28	0.77000	0.90820	-0.27498	-5.668	0.22944	28	0.69143
29	0.80000	1.00722	-0.28355	-4.197	0.20873	29	0.79418
30	0.83000	1.10623	-0.28938	-2.503	0.18659	30	0.89887
31	0.86000	1.20524	-0.29209	0.599	0.16322	31	0.99958
32	0.89000	1.30426	-0.29138	1.431	0.13883	32	1.10216
33	0.92000	1.40327	-0.28711	3.514	0.11359	33	1.20439
34	0.95000	1.50228	-0.27918	5.656	0.08762	34	1.30599
35	0.97500	1.58479	-0.26968	7.490	0.06553	35	1.40675
36	1.00000	1.66731	-0.25747	9.341	0.04322	36	1.50660
						37	1.58907
						38	1.64990
						39	1.66258
						40	1.66731

MERIDIONAL AIRFOIL GEOMETRY - STREAMLINE 13

MF ANL INF DATA

PT	PC1 X	X	Y	B.M.	T (M)	XS	YS	XP	YP
1	0.	-1.67260	0.69954	-32.312	0.05299	1	-1.67260	-1.67260	0.69954
2	0.02500	-1.58879	0.64673	-32.115	0.07337	2	-1.67634	0.67705	0.71254
3	0.05000	-1.50493	0.54337	-31.863	0.09366	3	-1.65591	0.66077	0.71016
4	0.07500	-1.42117	0.54260	-31.524	0.11378	4	-1.60829	0.61566	0.67780
5	0.10000	-1.33736	0.49162	-31.073	0.13361	5	-1.52970	0.55459	0.63414
6	0.12500	-1.25355	0.44167	-30.495	0.15303	6	-1.45091	0.49410	0.59109
7	0.15000	-1.16974	0.39299	-29.777	0.17192	7	-1.37184	0.43440	0.54884
8	0.17500	-1.08592	0.34585	-28.913	0.19014	8	-1.29238	0.37574	0.50760
9	0.20000	-1.00211	0.30049	-27.904	0.20760	9	-1.21243	0.31838	0.46760
10	0.23000	-0.90154	0.24874	-26.518	0.22739	10	-1.13189	0.26262	0.42907
11	0.26000	-0.80097	0.20021	-24.979	0.24580	11	-1.05069	0.20876	0.39222
12	0.29000	-0.70039	0.15506	-23.359	0.26269	12	-0.95230	0.14701	0.35047
13	0.32000	-0.59982	0.11331	-21.729	0.27797	13	-0.85286	0.08881	0.31161
14	0.35000	-0.49924	0.07485	-20.145	0.29155	14	-0.75247	0.03448	0.27565
15	0.38000	-0.39867	0.03944	-18.656	0.30336	15	-0.65127	0.01580	0.24243
16	0.41000	-0.29810	0.00683	-17.302	0.31336	16	-0.54945	0.06201	0.21170
17	0.44000	-0.19752	-0.02331	-16.050	0.32149	17	-0.44719	-0.10427	-0.35015
18	0.47000	-0.09695	-0.05122	-14.946	0.32767	18	-0.34470	-0.14276	-0.25150
19	0.50000	0.00362	-0.07704	-13.855	0.33191	19	-0.24205	-0.17777	-0.15300
20	0.53000	0.10420	-0.10084	-12.766	0.33431	20	-0.13921	-0.2051	0.10708
21	0.56000	0.20477	-0.12260	-11.636	0.33465	21	-0.03612	-0.23817	0.04337
22	0.59000	0.30534	-0.14222	-10.423	0.33246	22	0.06726	-0.26386	0.14113
23	0.62000	0.40592	-0.15953	-9.092	0.32759	23	0.17102	-0.28648	0.23852
24	0.65000	0.50649	-0.17433	-7.621	0.32002	24	0.27527	-0.30574	0.33542
25	0.68000	0.60707	-0.18635	-5.989	0.30992	25	0.38003	-0.32127	0.43180
26	0.71000	0.70764	-0.19533	-4.182	0.29751	26	0.49527	-0.33293	0.52771
27	0.74000	0.80821	-0.20096	-2.190	0.28300	27	0.59090	-0.34047	0.62323
28	0.77000	0.90879	-0.20291	0.004	0.26663	28	0.69679	0.34369	0.71849
29	0.80000	1.00936	-0.20086	2.372	0.24864	29	0.80281	0.34236	0.81362
30	0.83000	1.10993	-0.19448	4.917	0.22924	30	0.90378	0.33623	0.90879
31	0.86000	1.21051	-0.18346	7.598	0.20862	31	1.01450	-0.32507	1.00422
32	0.89000	1.31108	-0.16758	10.356	0.18691	32	1.11976	-0.30967	1.10111
33	0.92000	1.41165	-0.14666	13.150	0.16420	33	1.22430	-0.28685	1.19671
34	0.95000	1.51223	-0.12052	15.987	0.14051	34	1.32788	-0.25951	1.29428
35	0.97500	1.59604	-0.09460	18.380	0.12009	35	1.43033	-0.22660	1.39298
36	1.00000	1.67985	-0.06480	20.750	0.09935	36	1.53158	-0.18806	1.49288
						37	1.64497	-0.15158	1.57111
						38	1.65183	-0.13408	1.61525
						39	1.67707	-0.10938	1.65254
						40	1.67985	-0.06480	1.67485

3.0 PLANE SECTION BLADE COORDINATES

Figure 55 shows the stacked Phase III rotor plane sections. The following tabulation gives the coordinates for these sections. These sections are spaced one half inch apart, beginning at the tip height of 8.5 inches and progressing inward to 2.5 inches. These are the same section locations as given for the base line rotor in Reference 1. Also included in the tabulation are coordinates for the section meanline, the meanline angle, and the section percent thickness at each point. Plane section chord, camber angle, and stagger angle are also given. These coordinates are intended to represent the blade under hot running conditions and do not include any corrections for blade untwist, meanline deformation, centrifugal growth or thermal growth.

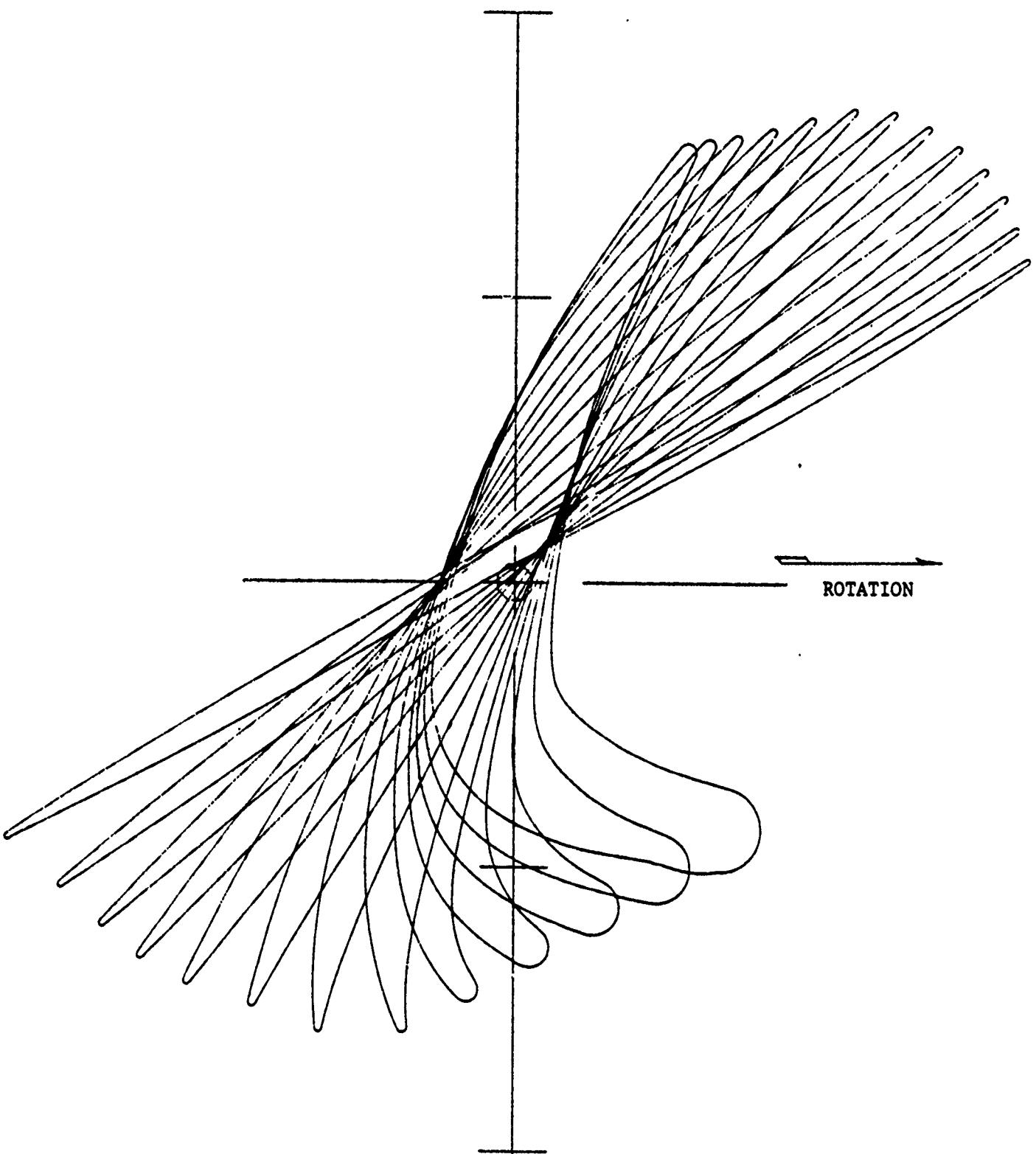


Figure 55. Stacked Phase III Rotor Plane Sections

PLATE IIII POUTER

•7PC•

COORD SYSTEM ORIGIN Z -7.04880 R O.
 SECTION NO 1 SECTION AA RHO 8.50000
 MU 0. ETA 0.

PI	ALPHA	ZETA	INPUT DATA		UPSILON
			MFANI. INF	THICKNESS	
1	-1.07809	57.235	0.01008	1.78986	
2	1.03248	57.651	0.02332	1.71825	
3	1.93913	58.482	0.03216	1.56349	
4	0.84594	59.336	0.04087	1.41322	
5	0.75024	60.296	0.04927	1.24869	
6	0.64325	61.519	0.05807	1.05649	
7	0.52468	62.775	0.06692	0.83168	
8	0.40438	63.611	0.07485	0.59296	
9	0.28275	63.811	0.08169	0.34613	
10	0.16062	63.276	0.08732	0.10310	
11	0.03858	62.305	0.09165	-0.13756	
12	0.0281	61.326	0.09466	-0.36406	
13	0.20319	60.835	0.09634	-0.58141	
14	0.32191	60.792	0.09608	-0.79403	
15	0.43863	60.961	0.09007	1.00373	
16	0.55211	61.383	0.07801	-1.21153	
17	0.66472	61.820	0.06075	-1.41859	
18	0.77327	61.539	0.03947	-1.62104	
19	0.86174	60.630	0.01983	-1.78123	

MFANI INF COORDINATES WITH ORIGIN AT SECTION AXIS

PI	R/C	AI	T/C	ALPHA	UPSILON	ZETA
1	0	0.00464		-1.07809	1.78986	57.434
2	0.0250	0.00581		-1.02959	1.71368	57.686
3	0.0500	0.00695		-0.98109	1.63630	58.159
4	0.0750	0.00808		-0.93260	1.55749	58.620
5	0.1000	0.00920		-0.88410	1.47729	59.056
6	0.1250	0.01029		-0.83561	1.39573	59.464
7	0.1500	0.01134		-0.78711	1.31281	59.906
8	0.1750	0.01237		-0.73862	1.22827	60.414
9	0.2000	0.01336		-0.69012	1.14191	60.962
10	0.2310	0.01451		-0.63193	1.03557	61.560
11	0.2600	0.01560		-0.57373	0.92615	62.141
12	0.2900	0.01662		-0.51554	0.81386	62.876
13	0.3210	0.01759		-0.45734	0.69908	63.326
14	0.3500	0.01850		-0.39915	0.58240	63.639
15	0.3800	0.01933		-0.34095	0.46448	63.901
16	0.4110	0.02010		-0.28276	0.34614	63.799
17	0.4410	0.02080		-0.22456	0.22882	63.640

PART III PCTOR

*70C.

STAGE 3. ROTOR
COORD SYSTEM ORIGIN Z -7.0480 R 0.
SECTION NO 1 SECTION AA

NB 20

RHO 8.5000

MIDLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	PCT AL	T/C	ALPHA	UPSILON	ZETA
18	0 4700	0.02143	0.16637	0.11152	63.323.
19	0 5000	0.02198	-0.10817	-0.00324	62.891.
20	0 5300	0.02247	0.04998	-0.11579	62.413.
21	0 5600	0.02288	0.00822	-0.22599	61.915.
22	0 5900	0.02321	0.06541	0.33399	61.456
23	0 6200	0.02347	0.12461	-0.44007	61.074
24	0 6500	0.02365	0.18280	-0.54487	60.866
25	0 6800	0.02378	0.24100	-0.64914	60.816
26	0 7100	0.02374	0.29919	0.75333	60.822
27	0 7400	0.02336	0.35739	-0.85765	60.870
28	0 7700	0.02258	0.41558	-0.96222	60.942
29	0 8000	0.02141	0.47378	-1.06717	61.055
30	0 8300	0.01985	0.53197	1.17285	61.246
31	0 8600	0.01793	0.59017	1.27978	61.599
32	0 8900	0.01565	0.64836	-1.38799	61.842
33	0 9200	0.01305	0.70656	1.49701	61.908
34	0 9500	0.01016	0.76475	-1.60536	61.533
35	0 9750	0.00756	0.81325	1.69387	61.071
36	1 0000	0.00488	0.86174	-1.78123	60.911
CHORD					
4 0679					
STAGGER					
61.499					
CAMBER					
-3.477					

SURFACE COORDINATES WITH ORIGIN AT SECTION. AXIS

PT	T/C	ALPHA	UPPER UPSILON	LOWER ALPHA	UPSTAIRS
1	0.00464	-1.07809	1.78986	-1.07809	1.78986
2	0.00464	-1.08240	1.78341	-1.07032	1.79105
3	0.00464	-1.08116	1.77669	-1.06484	1.78714
4	0.00581	-1.03956	1.70737	-1.01962	1.71999
5	0.00695	-0.99310	1.62884	-0.96909	1.64375
6	0.00808	-0.94662	1.54694	-0.91657	1.56605
7	0.00920	0.90013	1.46768	-0.86807	1.48690
8	0.01029	-0.85361	1.38512	-0.81761	1.40635
9	0.01134	0.80705	1.30125	-0.76717	1.32436
10	0.01237	0.76047	1.21587	-0.71676	1.24068
11	0.01346	0.71396	1.12873	-0.66538	1.15504
12	0.01451	-0.65787	1.02157	-0.60598	1.04956
13	0.01560	0.60179	0.91142	-0.54567	0.94087
14	0.01662	-0.54560	0.79846	-0.48547	0.82926

PLATE III ROTOR

COORDINATE ORIGIN Z -7.04880 R O.
 SECTION NO 1 SECTION AA
 SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	V/C	UPPER	LOWER	ALPHA	UPSILON
		ALPHA	UPPER	ALPHA	UPSILON
15	0.01759	-0.48928	0.68304	-0.42540	0.71513
16	0.01850	-0.43282	0.56571	-0.36547	0.59008
17	0.01943	-0.37620	0.44713	-0.30571	0.48162
18	0.02010	-0.31940	0.32811	-0.24611	0.36417
19	0.02080	-0.26243	0.20946	-0.18669	0.24699
20	0.02143	-0.20527	0.09178	-0.12746	0.13107
21	0.02198	-0.14793	-0.02360	-0.06841	0.01711
22	0.02247	-0.09044	-0.13693	-0.00752	0.03465
23	0.02288	-0.03279	-0.24787	0.04923	-0.20411
24	0.02321	0.02498	-0.35653	0.10784	-0.31145
25	0.02347	0.08287	-0.46314	0.16634	-0.41701
26	0.02365	0.14082	-0.56826	0.22478	-0.52147
27	0.02378	0.19881	0.67270	0.28318	-0.62558
28	0.02374	0.25707	-0.77685	0.34131	-0.72981
29	0.02336	0.31592	-0.88076	0.39886	-0.83453
30	0.02258	0.37547	-0.98451	0.45569	-0.93993
31	0.02144	0.43571	-1.08622	0.51184	-1.04611
32	0.01985	0.49659	-1.19223	0.56735	-1.15347
33	0.01793	0.55812	-1.29711	0.62221	-1.26246
34	0.01565	0.62033	-1.40299	0.67640	-1.37298
35	0.01305	0.68317	-1.50949	0.72994	-1.48452
36	0.01016	0.74661	-1.61520	0.78290	-1.59552
37	0.00756	0.79979	-1.70131	0.82570	-1.68644
38	0.00488	0.84777	-1.77766	0.86562	-1.76550
39	0.00188	0.85342	-1.78204	0.86691	-1.77424
40	0.00188	0.86174	-1.78123	0.86174	-1.78123
LE RAD	0.00969	CENTER AT ALPHA	-1.07286	UPSILON	1.78170
TF RAD	0.01050	CENTER AT ALPHA	0.85665	UPSILON	-1.77205

PHASE III ROTOR

•711.

	STAGE	3	ROTOR	MU	NP	20
COORD SYSTEM ORIGIN Z	-7.04880	R O	MU	O.	FIA	O.
SECTION NO 1	SECTION AA			RHO	B .5000	
CHORD A 0679	STAGFR 61.489			CAMFR -3 477		
AREA O 284298	SURFACE ARC LENGTH	8.1544R				
SECTION C.G. SURFACE SECTION C.G. B. ADF AXIS CLACKING AXIS (RADIAL)	ALPHA -0.03327 -0.03869 -0.03869 0.00210	UPSILON -0.12177 -0.10763 -0.10763 O.				

PHASE III ROTOR

CHORD SYSTEM ORIGIN 2 -7.04880 R. O.
SECTION NO 2 SECTION RR MU O. E1A 0.
RHO 8.0000

PI PI' AI T/C ALPHA MEANLINE INPUT DATA

PI	PI'	AI	T/C	ALPHA	THICKNESS	UPSILON
1	1	1.18293	55.367	0.01973	1.75033	
2	1	1.13124	55.814	0.02453	1.67495	
3	1	1.02672	56.653	0.03413	1.51853	
4	-0	0.92044	57.438	0.04362	1.35458	
5	0	0.81248	58.183	0.05287	1.18301	
6	0	0.69176	58.995	0.06257	0.98539	
7	0	0.55821	59.771	0.07238	0.75935	
8	0	0.42287	60.099	0.08112	0.52484	
9	0	0.28624	59.468	0.08854	0.28915	
10	0	0.14902	57.744	0.09443	0.06356	
11	0	0.01175	55.983	0.09873	-0.14649	
12	0	0.12520	54.887	0.10141	-0.34492	
13	0	0.26124	54.389	0.10245	-0.53653	
14	0	0.39620	54.257	0.10013	-0.72448	
15	0	0.52969	54.272	0.09190	-0.90987	
16	0	0.66115	54.368	0.07835	-1.09292	
17	0	0.79029	54.384	0.06047	-1.27324	
18	0	0.91705	54.094	0.03956	-1.44934	
19	0	1.02058	53.629	0.02081	-1.59154	

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	PI'	AI	T/C	ALPHA	UPSILON	ZETA*
1	0	0.00493	-1.18293	1.75033	55.367	
2	0	0.0250	0.00621	-1.12784	1.66996	55.815
3	0	0.0500	0.00748	-1.07275	1.58406	56.313
4	0	0.0750	0.00873	-1.01767	1.50475	56.727
5	0	0.1000	0.00997	-0.96258	1.42017	57.122
6	0	0.1250	0.01118	-0.90749	1.33428	57.531
7	0	0.1500	0.01237	-0.85240	1.24704	57.920
8	0	0.1750	0.01352	-0.79732	1.15853	58.277
9	0	0.2000	0.01464	-0.74223	1.06878	58.643
10	0	0.2300	0.01593	-0.67612	0.95931	59.106
11	0	0.2600	0.01716	-0.61002	0.84789	59.527
12	0	0.2900	0.01833	-0.54391	0.73477	59.861
13	0	0.3200	0.01942	-0.47781	0.62038	60.065
14	0	0.3500	0.02043	-0.41170	0.50541	60.110
15	0	0.3800	0.02136	-0.34560	0.39076	59.903
16	0	0.4100	0.02220	-0.27949	0.27772	59.399
17	0	0.4400	0.02295	-0.21339	0.16751	58.663

PLATE IIII RODOR

COORD SYSTEM ORIGIN 2 -7.04400 R O.
 SECTION NO 2 SECTION RP
 SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

RI	I/C	STAGE	1	ROTOR	MR	20
		SECTION NO	2	SECTION RP	RHO	FIA G.
15	0.01942	-0.51149	0.60099	-0.44413	0.63977	
16	0.02043	-0.44715	0.48504	-0.37625	0.52579	
17	0.02136	-0.38259	0.36932	-0.30861	0.41219	
18	0.02220	-0.31774	0.25511	-0.24125	0.30014	
19	0.02295	-0.25262	0.14362	-0.17416	0.19139	
20	0.02361	-0.18724	0.03558	-0.10732	0.04601	
21	0.02417	-0.12166	-0.06853	-0.04069	-0.01556	
22	0.02464	0.05597	-0.16913	-0.02583	-0.11400	
23	0.02502	0.09882	-0.26690	0.09225	-0.21004	
24	0.02531	0.07568	-0.36253	0.15860	-0.30435	
25	0.02551	0.14161	-0.45665	0.22488	-0.39756	
26	0.02559	0.20767	-0.54968	0.29102	-0.49010	
27	0.02551	0.27398	-0.64193	0.35692	-0.58240	
28	0.02514	0.34071	-0.73353	0.42241	-0.67474	
29	0.02440	0.40803	-0.82449	0.48730	-0.76741	
30	0.02328	0.47595	-0.91497	0.55159	-0.86053	
31	0.02181	0.54442	-1.00514	0.61533	-0.95418	
32	0.02003	0.61341	-1.09511	0.67855	-1.04839	
33	0.01793	0.68290	-1.18494	0.74127	-1.14315	
34	0.01557	0.75287	-1.27451	0.80351	-1.23821	
35	0.01297	0.82322	-1.36358	0.86537	-1.33327	
36	0.01017	0.89389	-1.45206	0.92691	-1.42821	
37	0.00772	0.95299	-1.52526	0.97798	-1.50712	
38	0.00520	1.00558	-1.58966	1.02326	-1.57668	
39	0.00520	1.01203	-1.59348	1.02497	-1.58405	
40	0.00520	1.02058	-1.59154	1.02058	-1.59154	
41	0.00520					
IF RADI	0.01013	CENTER AT ALPHA	-1.17717	UPSIION	1.74200	
IF RADI	0.01099	CFNTER AT ALPHA	1.01407	UPSIION	-1.58263	

PILOT III ROTOR

ZPC.

	STAGE	3.	ROTOR	WT:	20
CRANK SYSTEM ORIGIN Z	-7.04880	R O.	MU O.	ETA	O
SECTION NO 2	SECTION RR		RIED	8.0000	
CURRDI 4 X129	SIARGFR 56.601		CAMBR		
ARFA O 292919	SURFACF ARC LENGTH	8.03668	1.738		
SECTION C.G.	ALPHIA	UPSILON			
SURFACF SECTION C.G.	-0.03382	-0.06230			
BLANT AXIS	-0.03611	-0.05345			
CLACKING AXIS (RADIAL)	-0.03811	-0.05345			
	-0.00210	O.			

RIVAGE III ROTOR

•7PC•

STAGE 3. ROTOR
COPUP SYSTEM ORIGIN Z -7.04000 R 0.
SECTION NO 3 SECTION CC
MEANLINE INPUT DATA

PT	ALPHA	ZETA*	THICKNESS	UPSILON
1	1.28692	53.725	0.02050	1.71011
2	1.23010	54.079	0.02612	1.63218
3	-1.11510	54.726	0.03748	1.47150
4	-0.99843	55.307	0.04886	1.30456
5	-0.88005	55.887	0.06006	1.13165
6	0.74805	56.600	0.07196	0.93406
7	0.60198	57.036	0.08410	0.71034
8	-0.45414	56.422	0.09500	0.48440
9	0.30497	54.692	0.10428	0.26627
10	0.15487	52.303	0.11167	0.06337
11	0.00448	50.467	0.11760	0.12437
12	0.14591	49.625	0.12015	0.30355
13	0.29610	49.136	0.12106	0.47851
14	0.44561	48.788	0.11701	0.65023
15	0.59419	48.560	0.10625	0.81929
16	0.74166	48.415	0.08957	0.98595
17	0.88757	48.284	0.06815	-1.15027
18	1.03189	47.954	0.04351	-1.31164
19	1.15068	47.482	0.02171	-1.44268

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	P/C	AI	T/C	ALPHA	UPSILON	ZETA*
1	0	0.00014		1.28692	1.71011	53.725
2	0.0250	0.00666		-1.22598	1.62650	54.095
3	0.0500	0.00817		1.16504	1.54177	54.452
4	0.0750	0.00968		-1.10410	1.45593	54.806
5	0.1000	0.01117		-1.04316	1.36899	55.125
6	0.1250	0.01265		0.98222	1.28110	55.399
7	0.1500	0.01410		-0.92128	1.19230	55.686
8	0.1750	0.01553		0.86034	1.10248	56.006
9	0.2000	0.01692		-0.79940	1.01155	56.331
10	0.2300	0.01853		-0.72627	0.90097	56.708
11	0.2600	0.02007		0.65314	0.78505	56.940
12	0.2900	0.02153		-0.58002	0.67551	56.981
13	0.3200	0.02291		-0.50689	0.56438	56.756
14	0.3500	0.02418		-0.43376	0.45382	56.230
15	0.3800	0.02535		0.36563	0.34598	55.444
16	0.4100	0.02641		0.28750	0.24176	54.398
17	0.4400	0.02735		-0.21438	0.14173	53.255

PHASE III PULL

•7PC.

CORD SYSTEM ORIGIN 2 7.04880 R O.
SECTION NO. 3 SECTION CC RWD 7.5000

STAGE 3. ENDOR MU 0. EIA 0.

MF AND INF COORDINATES WITH ORIGIN AT SECTION AXIS

IT	IT	AI	I/C	ALPHA	UPSILON	ZETA
18	0	4700	0.02816	-0.14125	0.04581	52.097
19	0	5000	0.02886	0.06812	-0.04639	51.104
20	0	5300	0.02942	0.09501	-0.13585	50.409
21	0	5600	0.02985	0.07814	-0.22347	49.921
22	0	5900	0.03017	0.15127	-0.30984	49.591
23	0	6200	0.03037	0.24439	-0.39536	49.346
24	0	6500	0.03037	0.29752	-0.48048	49.125
25	0	6800	0.03067	0.37065	-0.56439	48.942
26	0	7100	0.02938	0.44378	0.64813	48.807
27	0	7400	0.02827	0.51691	-0.73152	48.696
28	0	7700	0.02676	0.58004	0.81459	48.584
29	0	8000	0.02487	0.63116	-0.89734	48.492
30	0	8300	0.02265	0.73629	0.97990	48.448
31	0	8600	0.02011	0.80942	-1.06236	48.409
32	0	8900	0.01730	0.88255	-1.14463	48.318
33	0	9200	0.01427	0.95568	-1.22659	48.199
34	0	9500	0.01106	1.02880	-1.30821	48.082
35	0	9700	0.00828	1.08974	1.37590	47.990
36	1	00000	0.00545	1.15058	-1.44288	47.482
					STAGGER 52.292	
					CAMBR 3.4854	

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

IT	IT	ALPHA	UPPER UPSILON	LOWER UPSILON	WPSILON
1	0	00514	1.28692	1.71011	-1.28692
2	0	00514	-1.29117	1.70281	-1.27858
3	0	00514	-1.28937	1.69561	-1.27236
4	0	00616	-1.23673	1.61872	1.21524
5	0	00717	-1.17829	1.53231	-1.15180
6	0	00968	-1.11986	1.44482	-1.08635
7	0	01117	-1.06143	1.35626	-1.02490
8	0	01265	-1.00297	1.26678	-0.96147
9	0	01410	-0.94449	1.17645	-0.88007
10	0	01553	-0.88600	1.08518	-0.83469
11	0	01692	-0.82746	0.99286	-0.77135
12	0	01851	0.75714	0.88070	-0.69544
13	0	02007	0.68666	0.76723	-0.61967
14	0	02153	0.61600	0.65315	-0.54404

PIVOT III ROTOR

7PC

STAGE 3. 80° Q
CLOUD SYSTEM ORIGIN 7 -7.04880 R O MU O. FIA O.
SECTION NO 3 SECTION CC RAD 7.5000

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

R1	T/C	ALPHA	UPPER UPSILON	LOWER ALPHIA	UPSILON
15	0.02291	-0.54506	0.53936	-0.46871	0.58941
16	0.02418	-0.47362	0.42704	-0.39370	0.40061
17	0.02535	-0.40224	0.31732	-0.31903	0.37463
18	0.02641	-0.33029	0.21113	-0.24472	0.27239
19	0.02735	-0.25804	0.10913	-0.17071	0.17433
20	0.02816	-0.18553	0.01133	-0.09696	0.06029
21	0.02886	-0.11268	-0.08250	-0.02336	-0.01029
22	0.02942	-0.04017	0.17322	0.05019	-0.09849
23	0.02985	0.03262	-0.26177	0.12365	-0.18518
24	0.03017	0.10549	-0.34880	0.19704	-0.27087
25	0.03037	0.17848	-0.43479	0.27031	-0.35592
26	0.03037	0.25176	0.51978	0.34329	-0.44057
27	0.031947	0.32547	-0.60374	0.41583	-0.52104
28	0.02928	0.39972	-0.68669	0.48784	-0.60957
29	0.03077	0.47459	0.76871	0.555923	-0.69433
30	0.02676	0.55005	-0.84986	0.63002	-0.77931
31	0.02487	0.62604	0.93019	0.70028	-0.86449
32	0.02265	0.70251	-1.00954	0.77007	-0.94995
33	0.02011	0.77945	-1.08896	0.83939	-1.03176
34	0.01730	0.85680	-1.16756	0.90030	-1.12171
35	0.01427	0.93448	-1.24554	0.97687	-1.20764
36	0.01106	1.01241	-1.32293	1.04520	-1.29349
37	0.00828	1.07751	-1.38696	1.10198	-1.36484
38	0.00545	1.13486	1.44268	1.15188	-1.42712
39	0.00545	1.14201	1.44588	1.15443	-1.43461
40	0.00145	1.15068	1.44288	1.15068	-1.44288
41	RAD 0.01055	CENTER AT ALPHA	-1.28068	UPSILON	1.70160
42	RAD 0.01155	CENTER AT ALPHA	1.14288	UPSILON	1.43136

PHASE III ROTOR

•ZPG•

	STAGE	3. ROTOR	NR	20
COORD SYSTEM ORIGIN	7 -7.04480 R O.	MU	O.	FTA
SECTION NO	3 SECTION CC	RND	7 500M	
CENTER 1.0954	STAGGER 52.7497	CAMFR		
AREA 0 335802	SURFACE ARC LENGTH	6.243		
SECTION C.G. SURFACE SECTION C.G. RADIAL AXIS STACKING AXIS (RADIAL)	ALPHA UPSILON	-0.03082 0.03288 -0.04129 -0.02354 -0.04129 -0.02354 -0.00210 O.		

PLATE III ROTOR

•7IV.

CORD SYSTEM ORIGIN 2 -7.04880 R O.
 SECTION NO 4 SECTION DD
 RND 7.0000

MF ANL. INF INPUT DATA

PT	ALPHA	ZFTA*	THICKNESS	UPSILON
1	1.37691	52.302	0.02080	1.64201
2	1.31555	52.631	0.02793	1.56218
3	1.19150	53.255	0.04253	1.39807
4	-1.06587	53.832	0.05737	1.22813
5	0.93870	54.337	0.07218	1.05251
6	0.79704	54.720	0.08808	0.85352
7	0.64047	54.326	0.10444	0.63293
8	0.48222	52.256	0.11922	0.41956
9	0.32259	49.469	0.13184	0.22340
10	0.16192	47.440	0.14192	0.04257
11	-0.00073	46.093	0.14906	0.12872
12	0.16090	45.089	0.15313	0.29360
13	0.32253	44.254	0.15360	0.45335
14	0.48404	43.538	0.14650	0.60678
15	0.64532	42.919	0.13138	0.76033
16	0.80597	42.342	0.10922	0.90826
17	0.96608	41.780	0.08146	1.05272
18	1.12538	41.260	0.04984	1.19371
19	1.25746	40.862	0.02193	1.30895

MF ANL. INF COORDINATES WITH ORIGIN AT SECTION AXIS

PT	PCT AL	T/C	ALPHA	UPSILON	ZFTA*
1	0	0.00526	-1.37691	1.64201	52.302
2	0.0250	0.00719	-1.31105	1.55629	52.630
3	0.05(X)	0.00915	-1.24519	1.46955	52.954
4	0.0750	0.01111	-1.17933	1.38177	53.282
5	0.1000	0.01308	-1.11347	1.29296	53.601
6	0.1250	0.01505	1.04761	1.20312	53.904
7	0.1500	0.01699	-0.98175	1.11233	54.177
8	0.1750	0.01891	-0.91590	1.02058	54.416
9	0.2000	0.02079	0.85004	0.92829	54.608
10	0.2300	0.02298	0.77101	0.81670	54.756
11	0.26(X)	0.02508	-0.69197	0.70504	54.600
12	0.29(X)	0.02709	-0.61294	0.59479	54.057
13	0.32(X)	0.02897	-0.53391	0.48746	53.127
14	0.35(X)	0.03073	-0.45488	0.38454	51.770
15	0.39(X)	0.03233	0.37585	0.28671	50.375
16	0.41(X)	0.03378	0.29682	0.19344	49.092
17	0.44(X)	0.03507	-0.21779	0.10399	48.023

PLATE III RODOR

•7PC.

COORD SYSTEM ORIGIN 7 -7.04880 R 0 MU 0 ETA 0.

SECTION NO 4 SECTION DD RAD 7.0000

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	I/C	STAGF	3	POLAR	MU	RAD	20
				UPPER UPSILON	LOWER ALPHA		UPSILON
15	0.02997	-0.57976	0.45307	-0.48807	0.52185		
16	0.03073	-0.50262	0.34693	-0.41714	0.42215		
17	0.03213	-0.42511	0.24592	-0.32659	0.32750		
18	0.03378	-0.34732	0.14968	-0.24632	0.23720		
19	0.03507	-0.26936	0.05760	-0.16622	0.15039		
20	0.03618	-0.19126	-0.03117	-0.08623	0.06605		
21	0.03711	-0.11301	-0.11743	-0.06644	-0.01646		
22	0.03785	-0.03451	-0.20153	-0.07312	-0.09745		
23	0.03840	0.04422	-0.28376	0.15245	-0.17719		
24	0.03878	0.12314	-0.16434	0.23159	-0.25586		
25	0.03896	0.20231	0.44342	0.31048	-0.33367		
26	0.03876	0.28197	0.52088	0.38889	-0.41096		
27	0.03807	0.36230	0.59660	0.46662	-0.48800		
28	0.02686	0.44330	0.67054	0.54368	-0.56487		
29	0.03518	0.52491	-0.74306	0.62013	0.64160		
30	0.03033	0.60709	-0.61397	0.69602	-0.71827		
31	0.03044	0.68981	-0.88351	0.71136	-0.79491		
32	0.02746	0.77303	-0.87174	0.84620	0.87143		
33	0.02414	0.85667	-1.01868	0.92062	0.94777		
34	0.02052	0.94065	-1.08441	0.99470	-1.02387		
35	0.01665	1.02489	-1.14903	1.06852	-1.09971		
36	0.01258	1.10932	-1.21272	1.14216	-1.17533		
37	0.00909	1.17978	-1.26525	1.20342	-1.23817		
38	0.00554	1.24144	-1.31075	1.25635	-1.29285		
39	0.00554	1.24906	-1.31304	1.26031	-1.30013		
40	0.00554	1.25746	-1.30895	1.25746	-1.30895		
	IF RAD 0.01077	CENTER AT ALPHA	-1.37033	UPSILON	1.63349		
	IF RAD 0.01188	CENTER AT ALPHA	1.24847	UPSILON	-1.30117		

PHASE III output

714

	STAGE	A	ROTOR	RP	20
COORD SYSTEM ORIGIN	2	7 OABRO R O.	WU	RIA	O.
SECTION NO	4	SECTION ID	RIID	7.0000	
1 HUB 3 DIA/R		SHAPE 4R 244	CARRIER		
AREA O A10556		SURFACE ARC LENGTH	7.97157		
SECTION C G SHRINKING SURFACE SECTION C G. RADIAL AXIS SHRINKING AXIS (RADIAL)		ALPHA -0.03095 -0.05116 0.05116 -0.00210	POSITION 0.03460 0.02121 0.02121 0		

MAIN III ROTOR

CORN SYSTEM ORIGIN Z -7 OABBO R O.
 SECTION NO 5 SECTION FF
 RHO 6.5000

MEANLINE INPUT DATA

R1	ALPHA	ZETA*	THICKNESS	UPSILON
1	1.45376	51.274	0.02071	1.55213
2	1.38873	51.624	0.03016	1.47038
3	1.25732	52.244	0.04963	1.30241
4	1.12445	52.631	0.06946	1.12922
5	0.98991	52.627	0.08924	0.95276
6	0.84015	52.052	0.11042	0.75831
7	0.67497	50.547	0.13217	0.55146
8	0.50800	48.013	0.15175	0.35692
9	0.33977	45.176	0.16846	0.17915
10	-0.17048	42.905	0.18175	0.01586
11	0.00045	41.181	0.19124	-0.13718
12	0.17012	39.813	0.19665	-0.28265
13	0.34127	38.652	0.19641	-0.42228
14	0.51260	37.606	0.18622	-0.55679
15	0.68425	36.627	0.16631	-0.68671
16	0.85594	35.660	0.13767	-0.81225
17	1.02784	34.682	0.10174	-0.93352
18	1.19817	33.660	0.06023	1.05039
19	1.34343	32.767	0.02276	-1.14459

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

R1	RCT	AI	1/C	ALPHA	UPSILON	ZETA*
1	0	0.00533	-1.45376	1.55213	51.274	
2	0	0.0250	0.00795	-1.38383	1.46418	51.694
3	0	0.0500	0.01061	-1.31390	1.37517	51.908
4	0	0.0750	0.01329	-1.24397	1.28514	52.328
5	0	0.1000	0.01598	-1.17405	1.19414	52.572
6	0	0.1250	0.01865	-1.10412	1.10255	52.679
7	0	0.1500	0.02131	-1.03419	1.01080	52.678
8	0	0.1750	0.02392	-0.96426	0.91922	52.577
9	0	0.2000	0.02648	0.89433	0.82817	52.342
10	0	0.2300	0.02946	0.81041	0.72030	51.861
11	0	0.2600	0.03233	-0.72649	0.61473	51.127
12	0	0.2900	0.03505	-0.64258	0.51241	50.116
13	0	0.3200	0.03760	0.55866	0.41409	48.877
14	0	0.3500	0.03997	0.47475	0.32034	47.430
15	0	0.3800	0.04214	-0.39083	0.23126	46.000
16	0	0.4100	0.04409	-0.30691	0.14638	44.669
17	0	0.4400	0.04582	-0.22300	0.06514	43.503

FRANCIS FLOW ROTOR

78%

CHORD SURFACE ORIGIN 2 -7.04880 R O.
 SECTION NO 5 SECTION FE MU O. EIA O.
 6.5000

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	P/C	AI	T/C	ALPHA	UPSILON	7FIA.
18	0.47442	0.04730	-0.13908	-0.01312	42.528	
19	0.50000	0.04855	-0.05517	-0.08992	41.670	
20	0.53000	0.04954	0.02875	-0.16260	40.911	
21	0.56000	0.05027	0.11266	-0.23443	40.224	
22	0.59000	0.05072	0.19658	-0.30462	39.603	
23	0.62000	0.05082	0.28050	-0.37334	39.034	
24	0.65000	0.05036	0.36441	-0.44074	38.513	
25	0.68000	0.04922	0.44833	-0.50691	38.006	
26	0.71000	0.04746	0.53224	-0.57190	37.501	
27	0.74000	0.04512	0.61616	-0.63572	37.016	
28	0.77000	0.04221	0.70008	-0.69847	36.557	
29	0.80000	0.03877	0.78399	-0.76017	36.099	
30	0.83000	0.03484	0.86791	-0.82084	35.628	
31	0.86000	0.03047	0.95182	-0.88045	35.143	
32	0.89000	0.02572	1.03574	-0.93898	34.643	
33	0.92000	0.02063	1.11965	-0.99643	34.154	
34	0.95000	0.01526	1.20357	-1.05286	33.693	
35	0.97500	0.01060	1.27350	-1.09913	33.273	
36	1.00000	0.00586	1.34343	-1.14459	32.767	
PIHRI				STAGGER	CAMFR	
1	0.9554			43.952	18.508	

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	P/C		UPPER	UPSILON	LOWER	UPSILON
1	0.00533	-1.45376	1.55213	-1.45376	1.55213	
2	0.00533	-1.45780	1.54453	-1.44537	1.55437	
3	0.01533	-1.45576	1.53729	-1.43885	1.55080	
4	0.00795	-1.39596	1.45460	-1.37171	1.47376	
5	0.01061	-1.33014	1.36249	-1.29767	1.38786	
6	0.01329	-1.26441	1.26937	-1.22354	1.30092	
7	0.01598	-1.19869	1.17527	-1.14940	1.21300	
8	0.01865	-1.13294	1.08058	-1.07529	1.12453	
9	0.02131	-1.06710	0.98574	-1.01127	1.03590	
10	0.02392	-1.00116	0.89098	-0.92735	0.94746	
11	0.02648	0.93505	0.79674	0.85360	0.85360	
12	0.02946	0.85543	0.68415	0.76539	0.75565	
13	0.03233	-0.77539	0.57532	-0.67760	0.65415	
14	0.03505	-0.69482	0.46875	-0.59043	0.55697	

PHASE III PRIORITY

7PC.

STAGE 3. ROTOR NR 20

CHORD SYSTEM ORIGIN Z -7.04880 R O. NR 0.

SECTION NO 5 SECTION EE NR 0.

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	R/C	UPPER ALPHA	UPPER UPSILON	LOWER ALPHA	LOWER UPSILON
15	0 03760	-0.61369	0.36605	-0.50363	0.46214
16	0 03997	-0.53193	0.26781	-0.41756	0.37288
17	0 04214	-0.44972	0.17439	-0.33194	0.28413
18	0 04409	-0.36714	0.08546	-0.24669	0.20130
19	0 04682	-0.28427	0.0058	-0.16172	0.12770
20	0 04730	-0.20120	0.08084	-0.07695	0.05461
21	0 04855	-0.11787	0.15937	0.00753	0.01847
22	0 04954	-0.03428	-0.23533	0.09177	-0.08987
23	0 05027	0.04960	-0.30909	0.17573	-0.15987
24	0 05072	0.13376	-0.38054	0.25940	-0.22869
25	0 05082	0.21832	0.45003	0.34267	-0.29665
26	0 05036	0.30350	0.51729	0.42533	-0.36419
27	0 04922	0.38945	-0.58226	0.50721	-0.43156
28	0 04746	0.47611	-0.64505	0.58838	-0.49874
29	0 04512	0.56339	-0.70571	0.66893	-0.56573
30	0 04221	0.65123	-0.76434	0.74892	-0.63259
31	0 03877	0.73961	-0.82104	0.82637	-0.69931
32	0 03484	0.82848	0.87586	0.90734	-0.76582
33	0 03047	0.91774	-0.92886	0.98590	-0.83204
34	0 02572	1.00733	-0.98009	1.06414	-0.89787
35	0 02063	1.09715	-1.02960	1.14216	-0.96325
36	0 01526	1.18713	-1.07753	1.22001	-1.02820
37	0 01050	1.26220	1.11635	1.28480	-1.08191
38	0 00586	1.32707	-1.14908	1.34076	-1.12784
39	0 00146	1.33528	-1.15016	1.34518	-1.13503
40	0 0X586	1.34343	-1.14459	1.34343	-1.14459
41	RAD 0 01083	CENTER AT ALPHA	-1.44700	UPSILON	1.54366
42	RAD 0 01272	CENTER AT ALPHA	1.33274	UPSILON	-1.13770

PILOT III ROTOR

•TOP•

	STAGE	3	ROTOR	MP.	20
CENTER SYMM ORIGIN	2	-7.0440	R O.	0	0
SECTION NO	5	SECTION FE			
CHORD		SIAGFR		R140	6.5000
3 RMS		43.952		CAMFR	
AREA	0 506647	SURFACE ARC LENGTH	7.86397	IR.508	
SECTION C.G.		ALPHA	UPSION		
SURFACE SECTION C.G.		-0.03211	-0.03877		
RADIAL AXIS		-0.06264	-0.02788		
WALKING AXIS (RADIAL)		0.06264	0.02788		
		-0.00210	0.		

PLATE III FLOOR

COORDINATE SYSTEM ORIGIN 7 -7.04000 R. O.
SECTION NO. 6 SECTION FF
RHO 6.00000

STAGE 3. ROTOR
PI PI AL T/C ALPHA UPSILON ZETA.
18 0 4700 0.05841 -0.13928 -0.01852 37.346
19 0 5000 0.05989 -0.05117 -0.04653 36.427
20 0 5300 0.06106 0.02693 -0.14856 35.526
21 0 5600 0.06190 0.12503 -0.21046 34.643
22 0 5900 0.06231 0.21313 0.27036 33.781
23 0 6200 0.06215 0.30123 -0.32837 32.949
24 0 6500 0.06128 0.38933 -0.38460 32.150
25 0 6800 0.05963 0.47743 -0.43913 31.362
26 0 7100 0.05726 0.56553 0.49201 30.580
27 0 7400 0.05421 0.65363 0.54326 29.792
28 0 7700 0.05051 0.74173 -0.59289 28.992
29 0 8000 0.04619 0.82983 -0.64089 28.169
30 0 8300 0.04132 0.91793 -0.68723 27.317
31 0 8600 0.03595 1.00603 -0.73188 26.430
32 0 9000 0.03013 1.09413 -0.77480 25.503
33 0 9300 0.02394 1.18224 -0.81590 24.494
34 0 9500 0.01741 1.27034 -0.85502 23.377
35 0 9750 0.01177 1.34375 0.88604 22.440
36 1.00000 0.00602 1.41717 -0.91570 21.562

CHORD
1 16.75
PI PI STAGGER
18 824
CAMFER
28.851

MFAN INFL COORDINATES WITH ORIGIN AT SECTION AXIS

PI	T/C	ALPHA	UPSILON	ZETA.
1	0.00528	-1.51952	1.44751	-1.51952
2	0.00528	-1.52311	1.44011	-1.51148
3	0.01528	-1.52128	1.43317	-1.450510
4	0.01871	-1.45879	1.34800	-1.43342
5	0.01217	-1.39047	1.25405	-1.35490
6	0.01564	-1.32217	1.15953	-1.27696
7	0.01911	-1.25382	1.06480	-1.19768
8	0.02255	-1.18535	0.97036	-1.11952
9	0.01504	-1.11670	0.87669	-1.04123
10	0.02027	-1.04183	0.78421	-0.96337
11	0.02611	0.97466	0.69329	0.85770
12	0.01627	-0.89517	0.58689	0.79209
13	0.01987	0.8104	0.48417	0.70042
14	0.01227	0.72617	0.38579	0.40559

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	T/C	ALPHA	UPSILON	ZETA.	LOWER ALPHA	UPPSILON
1	0.00528	-1.51952	1.44751	-1.51952	1.44751	1.44751
2	0.00528	-1.52311	1.44011	-1.51148	1.44011	1.44011
3	0.01528	-1.52128	1.43317	-1.450510	1.44655	1.44655
4	0.01871	-1.45879	1.34800	-1.43342	1.34800	1.34800
5	0.01217	-1.39047	1.25405	-1.35490	1.28300	1.28300
6	0.01564	-1.32217	1.15953	-1.27696	1.19665	1.19665
7	0.01911	-1.25382	1.06480	-1.19768	1.11017	1.11017
8	0.02255	-1.18535	0.97036	-1.11952	1.02412	1.02412
9	0.01504	-1.11670	0.87669	-1.04123	0.93808	0.93808
10	0.02027	-1.04183	0.78421	-0.96337	0.85516	0.85516
11	0.02611	0.97466	0.69329	0.85770	0.77311	0.77311
12	0.01627	-0.89517	0.58689	0.79209	0.67776	0.67776
13	0.01987	0.8104	0.48417	0.70042	0.58644	0.58644
14	0.01227	0.72617	0.38579	0.40559	0.40559	0.40559

PILOT III RUTTOR

COORD SYSTEM ORIGIN 7 -7.04880 R O:
 SECTION NO 6 SECTION FF
 RHO 6.00000

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS:

PT	PCT	AI	T/C	ALPHA	UPSILON	ZETA
18	0.4700	0.05841	-0.13928	-0.01852	37.346	
19	0.5000	0.05989	-0.06117	-0.04463	36.427	
20	0.5300	0.06106	0.03693	-0.14858	35.526	
21	0.5600	0.06190	0.12503	-0.21046	34.643	
22	0.5900	0.06231	0.21313	-0.27036	33.761	
23	0.6200	0.06215	0.30123	-0.32537	32.949	
24	0.6500	0.06128	0.38932	-0.38460	32.150	
25	0.6800	0.05963	0.47743	-0.43913	31.362	
26	0.7100	0.05726	0.56553	-0.49201	30.580	
27	0.7400	0.05421	0.65363	-0.54325	29.792	
28	0.7700	0.05051	0.74173	-0.59289	28.992	
29	0.8000	0.04619	0.82983	-0.64089	28.169	
30	0.8300	0.04132	0.91793	-0.68723	27.317	
31	0.8600	0.03595	1.00603	-0.73186	26.430	
32	0.8900	0.03013	1.09413	-0.77480	25.503	
33	0.9200	0.02394	1.18224	-0.81590	24.494	
34	0.9500	0.01741	1.27034	-0.85502	23.377	
35	0.9750	0.01177	1.34375	-0.88604	22.440	
36	1.0000	0.00602	1.41717	-0.91570	21.562	

CARRIER
 1.76765

STAGGER
 38.824
 28.851

SUPER ACF COORDINATES WITH ORIGIN AT SECTION AXIS:

PT	T/C	ALPHA	UPPER UPSILON	LOWER UPSILON	UPSILON
1	0.00528	-1.51952	1.44751	-1.51952	1.44751
2	0.00528	-1.52321	1.44011	-1.51148	1.44011
3	0.00528	-1.52128	1.43317	-1.50510	1.44655
4	0.00871	-1.45879	1.34800	-1.43342	1.36881
5	0.01217	-1.39047	1.25405	-1.35490	1.28300
6	0.01564	-1.32217	1.15953	-1.27636	1.19665
7	0.01911	-1.25382	1.06480	-1.19788	1.11017
8	0.02255	-1.18535	0.97036	-1.11952	1.02412
9	0.02504	-1.11670	0.87669	-1.04133	0.93898
10	0.02927	-1.04183	0.78421	-0.96337	0.85516
11	0.03211	0.97466	0.69329	0.88570	0.77313
12	0.03627	-0.89517	0.58689	-0.79289	0.67776
13	0.03987	0.81104	0.48417	0.70092	0.58644
14	0.04227	0.72617	0.38579	-0.3959	0.40999

PIASI III ROTOR

•7PC•

COORDINATE ORIGIN 2 -7.04880 R O.
 SECTION NO 6 SECTION FF
 RAD 6.0000

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

R1	I/C	ALPHA	UPPER UPSILON	LOWER ALPHA	UPSILON
15	0.04647	-0.64060	0.29229	-0.51896	0.41833
16	0.04942	-0.55432	0.20386	-0.42904	0.34171
17	0.05211	-0.46764	0.12048	-0.33951	0.26334
18	0.05451	-0.38073	0.04147	-0.25023	0.20020
19	0.05662	-0.29353	0.03385	-0.16122	0.13361
20	0.05841	-0.20606	-0.10604	-0.07249	0.06301
21	0.05949	-0.11820	-0.17545	0.01585	0.00618
22	0.06106	-0.02994	-0.24224	0.10380	-0.05493
23	0.06150	0.05871	0.30645	0.19135	-0.14448
24	0.06231	0.14783	-0.36797	0.27842	-0.17275
25	0.06215	0.23752	-0.42667	0.36494	-0.23007
26	0.05128	0.32787	-0.48239	0.45079	0.29611
27	0.05963	0.41893	-0.53511	0.53593	-0.34316
28	0.05726	0.51063	-0.58492	0.62043	-0.39910
29	0.05421	0.60287	-0.63192	0.70439	-0.45460
30	0.05051	0.69560	-0.67615	0.78787	-0.50963
31	0.04519	0.78873	-0.71764	0.87093	-0.56414
32	0.04132	0.88219	-0.75643	0.95367	-0.61804
33	0.03595	0.97588	-0.79255	1.03619	-0.67122
34	0.03013	1.06968	-0.82606	1.11859	-0.72354
35	0.02394	1.16353	-0.85696	1.20094	-0.77184
36	0.01741	1.25731	-0.88515	1.28336	-0.82490
37	0.01177	1.33529	-0.90651	1.35222	0.86554
38	0.00602	1.40195	-0.92354	1.41148	-0.89957
39	0.00602	1.41015	0.92288	1.41712	0.90578
40	0.00602	1.41717	-0.91570	1.41717	-0.91570
41	RAD 0.01051	CENTER AT ALPHA	-1.51282	UPSILON	1.43941
42	RAD 0.01302	CENTER AT ALPHA	1.40507	UPSILON	0.91090

PHASE 111 ROTOR

7PC

COORD SYSTEM ORIGIN	Z	0	X	Y	MU	N	FIA	O.
SECTION NO	6							RHO 6.0000
CIRCDP								CAMFR 28.851
	3.7616							
AREA	0.590286							
		SURFACE	ARC LENGTH					7.68397
SECTION C.G.		ALPHA	URSTON					
STREAM SURFACE		-0.03927	-0.01718					
BL AFF. AXIS		-0.07125	-0.01790					
TRACKING AXIS (RADIAL)		-0.07125	-0.01790					
		0.00210	0					

CLOUD SYSTEM ORIGIN Z -7.04880 R O
SECTION NO 7 SECTION GG MU O. FIA O.
RHO S.5(XX)

PT	ALPHA	ZETA+	THICKNESS	INPUT DATA	UPSILON
1	-1 56840	49.293	0.01886		1.32753
2	1.49634	49.264	0.03326		1.24356
3	-1.35082	48.979	0.06249		1.07497
4	1 20357	48.313	0.09183		0.90726
5	-1 05466	47.083	0.12074		0.74324
6	0 88936	44.678	0.15132		0.57199
7	0.70743	41.121	0.18225		0.40284
8	0 52434	38.011	0.20952		0.25199
9	0 34035	35.568	0.23205		0.11456
10	0 15562	33.209	0.24910		0.01186
11	0.02968	30.817	0.26013		-0.12785
12	0 21560	28.477	0.26456		-0.23385
13	0 40208	26.271	0.25837		0.33078
14	0 58902	24.019	0.24067		-0.41923
15	0 77666	21.452	0.21240		-0.49878
16	0 96502	18.332	0.17454		-0.56805
17	1 15395	14.373	0.12792		-0.62462
18	1 34358	9.329	0.07312		-0.66460
19	1.50177	4.166	0.02184		-0.68205

MFANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	PCT AL.	V/C	ALPHA	UPSILON	ZETA+
1	0	0.00514	1.56840	1.32753	49.293
2	0 0250	0.00932	1.49165	1.23809	49.358
3	0 0400	0.01352	1.41489	1.14893	49.186
4	0 0750	0.01772	1.33814	1.06037	48.972
5	0 1000	0.02190	1.26138	0.97262	48.659
6	0 1250	0.02604	1.18463	0.88603	48.211
7	0 1500	0.03012	1.10787	0.80100	47.610
8	0 1750	0.03412	1.03112	0.71800	46.834
9	0 2000	0.03802	0.95437	0.63757	45.794
10	0 2300	0.04255	0.86226	0.54545	44.143
11	0 2600	0.04687	0.77016	0.45879	42.350
12	0 2900	0.05094	0.67805	0.37745	40.550
13	0 3200	0.05474	0.58595	0.30092	38.932
14	0 3500	0.05821	0.49384	0.22834	37.576
15	0 3900	0.06135	0.40174	0.15905	36.334
16	0 4100	0.06412	0.30363	0.09277	35.158
17	0 4400	0.06651	0.21753	0.02928	33.996

PIAASI III ROTOR

470°C.

CORIN SYSTEM ORIGIN Z -7.0MRO R 0.
 SECTION NO 7 ROTOR
 SECTION GG

STAGE 3

RATOR

NR 20

MU O. TIA O.

RHO 5.5000

MANIFOLD COORDINATES WITH ORIGIN AT SECTION AXIS

R1	R2	AL	I/C	ALPHA	UPSILON	7FIA*
18	0 4700	0 06849	-0 12542	-0.03149	32.834	
19	0 5000	0 07005	-0.03332	-0.08960	31.662	
20	0 5300	0 07122	0.05879	-0.14511	30.481	
21	0 5600	0 07195	0 15089	-0.19807	29.329	
22	0 5900	0 07205	0 24300	-0.24865	28.223	
23	0 6200	0 07138	0.33510	-0.29698	27.154	
24	0 6500	0 06994	0.42721	0.34317	26.117	
25	0 6800	0 06774	0.51931	0 38728	25.047	
26	0 7100	0 06482	0.61142	0.42923	23.917	
27	0 7400	0 06121	0 70352	-0.46893	22.705	
28	0 7700	0 05696	0.79563	-0.50626	21.397	
29	0 8000	0 05210	0.88773	-0.54103	19.939	
30	0 8300	0 04665	0 97984	-0.57299	18.299	
31	0 8600	0 04065	1.07194	0 60184	16.437	
32	0 8900	0 03412	1 16405	-0.62722	14.325	
33	0 9200	0 02709	1.25615	-0.64865	11.773	
34	0 9500	0 01953	1.34826	0 66532	8.645	
35	0 9750	0 01284	1.42501	0.67520	6 123	
36	1 0000	0 00595	1.50177	0.68205	4.166	

CHORD 3 6694 STAGGER 33.207 CANTER 45.126

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

R1	R2	AL	UPPER	UPPER	LOWER	UPPER
1	0 00514	1.56840	1.32753	1.56840	1.32753	
2	0 00514	-1.57190	1.32038	-1.56083	1.32940	
3	0 00514	-1.56989	1 31385	-1.55465	1.32692	
4	0 00932	-1.50462	1.22695	-1.47867	1.24922	
5	0 01352	1.43367	1.13271	-1.39612	1.16514	
6	0 01712	-1.36267	1.03903	-1.31361	1.08172	
7	0 02190	-1.29155	0.94607	-1.23121	0.99916	
8	0 02604	-1.22025	0.85419	-1.14901	0.91786	
9	0 03012	-1.14869	0.76375	-1.06706	0.83826	
10	0 03412	-1.07678	0.67518	-0.98546	0.76083	
11	0 04002	-1.00438	0.59893	-0.96436	0.68621	
12	0 04295	-0.91663	0.48943	-0.80789	0.60147	
13	0 04687	0.82808	0.39525	-0.71773	0.52234	
14	0 05004	0.73881	0.30643	-0.61729	0.44846	

PIANOFORTE RULOR

STAGE 3 ROTOR
COORD SYSTEM ORIGIN Z -7.04800 R O.
SECTION NO 7 SECTION GG
SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI F/C ALPHAS UPPER UPSILON LOW R ALPHA UPSILON

15	0 05474	-0. 64905	0. 22280	-0. 52284	0. 37904
16	0 05821	-0. 55897	0. 14369	-0. 42871	0. 31299
17	0 06135	-0. 46843	0. 06839	-0. 33505	0. 24973
18	0 06412	-0. 37737	-0. 03341	-0. 24189	0. 18895
19	0 06651	-0. 28575	-0. 07148	-0. 14930	0. 13045
20	0 07449	-0. 19355	-0. 13707	-0. 05729	0. 07409
21	0 07005	-0. 10078	-0. 19899	0. 03414	0. 01978
22	0 07122	-0. 00750	0. 25772	0. 12507	-0. 03249
23	0 07195	0. 08623	-0. 31317	0. 21556	-0. 05298
24	0 07205	0. 18048	-0. 36513	0. 30551	-0. 13248
25	0 07138	0. 27533	0. 41351	0. 39487	-0. 18745
26	0 06794	0. 37072	-0. 15839	0. 48369	-0. 22795
27	0 06774	0. 46670	-0. 49987	0. 57193	-0. 27464
28	0 067482	0. 56324	0. 53794	0. 65963	-0. 32054
29	0 06121	0. 66018	-0. 57253	0. 74687	-0. 36534
30	0 06596	0. 75750	0. 60356	0. 83375	-0. 40896
31	0 05210	0. 85514	-0. 63089	0. 92033	-0. 45118
32	0 04665	0. 95296	0. 65426	1. 00674	-0. 49173
33	0 04065	1. 05084	0. 67337	1. 09305	0. 53031
34	0 03412	1. 14856	0. 68787	1. 17954	-0. 56656
35	0 02709	1. 24601	-0. 69731	1. 26629	-0. 59999
36	0 01953	1. 34287	-0. 70074	1. 35364	-0. 62990
37	0 01284	1. 42250	-0. 69862	1. 42752	-0. 65179
38	0 00515	1. 49004	0. 69392	1. 49199	-0. 66732
39	0 00595	1. 49718	0. 69098	1. 49910	-0. 68205
40	0 00995	1. 50177	-0. 68205	1. 50177	0. 68205

CE RAD	C 01006	CENTER AT ALPHIA	-1	56185	WPSILON	1.31980
CE RAD	C 011287	CENTER AT ALPHIA	-1	48693	WPSILON	0.68110
CE RAD	C 011287	CENTER AT ALPHIA	-1	48693	WPSILON	0.68110

PIALET III ROTOR

.700.

	STAGE	3. ROTOR	18	20
COORDINATE ORIGIN Z	-7.04880	R 0	MJ 0.	EIA 0
SECTION NO	7	SECTION GG	R140	5.5000
CHORD	3.6534	SLACKER	CAMBER	
		3.3 207	45.126	
AREA	0 640526	SURFACE ARC LENGTH	7 56495	
SECTION C.G.		ALPHA	UPSILON	
SURFACE SECTION C.G.		-0.02894	-0.01024	
BLADE AXIS		-0.06138	-0.02051	
STAKING AXIS (RADIAL)		0.06138	-0.02051	
		-0.00210	0	

PLATE III ROTOR

•7PC•

COORD SYSTEM ORIGIN 2 -7.04880 R O.
 SECTION NO 8 SECTION 100
 MEAN INPUT DATA

PI	ALPHA	THETA	THICKNESS	UPSILON
1	1 57695	47 508	0.01945	1.18826
2	1 49990	47 249	0.03628	1.10464
3	1 34494	46 574	0.07029	0.93898
4	1 18905	45 397	0.10399	0.77735
5	1 03278	43 473	0.13651	0.62380
6	-0 .86065	40 502	0.16993	0.46865
7	0 67770	37 121	0.20247	0.31756
8	0 48496	34 145	0.23004	0.18305
9	0 29729	31 318	0.25231	0.06238
10	0 10981	28 387	0.26927	-0.04517
11	0 07733	25 442	0.28102	-0.14000
12	0 26428	22 490	0.28776	-0.22776
13	0 45078	19 249	0.28627	-0.29343
14	0 63654	15 477	0.27192	-0.35093
15	0 62139	11 111	0.24329	-0.39364
16	1 00439	5 998	0.19996	-0.41993
17	1 18459	0 376	0.14156	0.42768
18	1 36003	-8 710	0.07962	-0.41423
19	1 50109	-17 691	0.02843	0.38218

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	R/C AL.	T/C	ALPHA	UPSILON	ZETA*
1	0	0.00563	-1.57695	1.18826	47.508
2	0 0250	0.01049	-1 50000	1 10475	47.200
3	0 0510	0.01538	1 42305	1 02202	46.929
4	0 0750	0.02027	-1 34610	0 94020	46.567
5	0 1000	0.02512	-1 26915	0 85957	46.079
6	0 1250	0.02990	-1 19220	0 78053	45.424
7	0 1500	0.03460	-1 11525	0 70353	44.581
8	0 1750	0.03918	1 03830	0 62903	43.524
9	0 2000	0.04362	-0 96134	0 55747	42.290
10	0 2300	0.04873	0 86900	0 47580	40.654
11	0 2600	0.05355	-0 77666	0 39888	38.945
12	0 2900	0.05805	0 68432	0 32639	37.328
13	0 3200	0.06221	-0 59196	0 25790	35.807
14	0 3500	0.06600	-0 49964	0 19305	34.369
15	0 3800	0.06943	0 40730	0 13151	32.982
16	0 4100	0.07248	0 31496	0 07318	31.571
17	0 4400	0.07516	-0 22262	0 01001	30.438

PLATE III MOTOR

•7PC.

STAGE

3.

MOTOR

COORD SYSTEM ORIGIN

Z -7.04880 R O.

SECTION NO 8

SECTION 1#1

MF ANI INE COORDINATES WITH ORIGIN AT SECTION AXIS

PT I/C 1/C ALPHA UPPERSILON ZETA

18	0 4700	0 07746	-0 13027	0 03405	28.682
19	0 5000	0 07940	0 03793	-0 08305	27.212
20	0 5300	0 08098	0 05441	-0 12904	25.740
21	0 5600	0 08225	0 14675	0 17211	24.263
22	0 5900	0 08312	0 23909	-0 21229	22.764
23	0 6200	0 08347	0 33143	-0 24961	21.222
24	0 6500	0 08311	0 42377	-0 28396	19.570
25	0 6800	0 08185	0 51611	-0 31521	17.789
26	0 7100	0 07958	0 60845	-0 34313	15.832
27	0 7400	0 07630	0 70080	-0 36750	13.704
28	0 7700	0 07195	0 79314	-0 38811	11.431
29	0 8000	0 06650	0 88548	-0 40478	8.989
30	0 8300	0 05995	0 97782	-0 41720	6.288
31	0 8600	0 05234	1 07016	-0 42500	3.325
32	0 8900	0 04374	1 16250	-0 42780	0.103
33	0 9200	0 03428	1 25484	0 42514	-3.509
34	0 9500	0 02441	1 34718	0 41607	-7.823
35	0 9750	0 01627	1 42413	-0 40261	-12.197
36	1 0000	0 00823	1 50109	-0 38218	-17.691

CHORD
3 4555STAGGER
27.031
CAMER
65.199

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PT I/C 1/C ALPHA UPPERSILON LOWER UPPILON

1	0 00563	-1 57695	1 18826	1 57695	1 18826
2	0 00563	-1 58034	1 18072	-1 56924	1 19099
3	0 00563	-1 57808	1 17405	-1 56271	1 18817
4	0 01049	-1 51330	1 09243	1 48670	1 11706
5	0 01538	-1 44246	1 00387	-1 40364	1 04017
6	0 02027	-1 37152	0 91613	-1 32067	0 96427
7	0 02512	-1 30041	0 82947	-1 23789	0 88967
8	0 02990	-1 22700	0 74427	-1 15510	0 81679
9	0 03460	-1 15720	0 66095	-1 07329	0 74610
10	0 03918	-1 08491	0 57994	-0 99168	0 67811
11	0 04362	-1 01206	0 50172	-0 91063	0 61323
12	0 04873	-1 92385	0 41193	-0 81415	0 53968
13	0 05375	-0 83482	0 32697	-0 71851	0 47083
14	0 05805	-0 74514	0 24664	-0 62350	0 40614

PHASE III ROTOR

•7PC.

COORD SYSTEM ORIGIN 7 -7.04480 R O.
 SECTION NO 8 SECTION IN
 SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

R	I/C	URDFR	ALPHA	UPSILON	LUMFR	ALPHIA	UPSILON
15	0.06221	-0.65486	0.17074	-0.52910	0.34507		
16	0.06600	-0.56401	0.09892	-0.43526	0.28717		
17	0.06943	-0.47260	0.03089	-0.34210	0.23213		
18	0.07248	-0.38052	-0.03350	0.24940	0.17987		
19	0.07516	-0.28781	0.0429	-0.15742	0.13031		
20	0.07746	-0.19451	-0.15147	-0.06604	0.04336		
21	0.07940	-0.10066	-0.20504	0.02490	0.03695		
22	0.08098	-0.00636	-0.25507	0.11517	-0.00301		
23	0.08225	0.08836	-0.30166	0.20514	-0.04256		
24	0.08312	0.18352	-0.34472	0.29466	-0.07987		
25	0.08347	0.27923	0.38404	0.38363	-0.11517		
26	0.08311	0.37567	-0.41926	0.47187	-0.14866		
27	0.08185	0.47291	-0.44985	0.55932	-0.18256		
28	0.07958	0.57094	0.47542	0.64597	-0.21085		
29	0.07630	0.66957	-0.49557	0.73202	-0.23943		
30	0.07195	0.76850	-0.50996	0.81778	-0.26625		
31	0.06650	0.86753	-0.51826	0.90343	-0.29129		
32	0.05995	0.96647	-0.52015	0.98916	-0.31424		
33	0.05234	1.06492	-0.51528	1.07540	-0.33472		
34	0.04374	1.16237	-0.50338	1.16264	-0.35227		
35	0.03428	1.25847	0.48427	1.25122	-0.36602		
36	0.02441	1.35292	-0.45785	1.34144	-0.37428		
37	0.01627	1.43007	-0.43008	1.41820	-0.37514		
38	0.00323	1.49241	-0.40224	1.48302	-0.37039		
39	0.00123	1.49941	-0.39582	1.49414	-0.37299		
40	0.00123	1.50109	-0.38218	1.50109	-0.38218		
41	RAD 0.01046	CENTER AT ALPHA	-1.56988	POSITION	1.16055		
42	RAD 0.01685	CENTER AT ALPHA	1.49498	POSITION	-0.38712		

PILOT III ROTOR

*INC.

	STAGE	3	ROTOR	MU	MU	20
CORD SYSTEM ORIGIN	2	-7.04880	R 0.	MU	0	FIA 0.
SECTION NO	8	SECTION	H4	RHO	5	(XXXX)
CHORD		STAGGER		CAMBER		
3.4555		27.031		65.199		
ARI A	0 678450	SURFACE	ARC LENGTH	7.27927		
SECTION C.G.		ALPHA	UPSILON			
SURFACE SECTION C.G.		-0.00510	-0.00283			
BLADE AXIS		0.01169	0.03865			
SPACING AXIS (RADIAL)		-0.01169	0.03865			
		-0.00210	0.			

PILOT III ROTOR

70C

COORD SYSTEM ORIGIN 2 -7.04880 R O.
SECTION NO 9 SECTION JU MU O. EIA O.

WB 20
RIO 4.5000

MFANI INF INPUT DATA

PT	ALPHA	ZETA*	THICKNESS	UPSILON
1	1 54614	45.513	0.02150	1.04166
2	.1 46696	45.250	0.02934	0.96146
3	1 30880	44.474	0.07431	0.80407
4	1 15028	42.932	0.10784	0.65334
5	0 99380	40.525	0.13959	0.51341
6	0 82178	37.430	0.17225	0.37452
7	0 63500	33.979	0.20490	0.24053
8	-0.44941	30.491	0.23424	0.12362
9	0.26514	26.940	0.25993	0.02272
10	-0.08216	23.398	0.28118	-0.06312
11	0.09902	20.053	0.29664	-0.13525
12	0 27828	16.779	0.30376	0.19545
13	0.45512	13.141	0.29726	0.24263
14	0.62877	9.249	0.27289	-0.27762
15	0.79835	4.959	0.23570	-0.29941
16	0.96269	-1.869	0.19328	-0.30511
17	1 12018	-13.461	0.14981	-0.28472
18	1 26985	-28.736	0.10590	-0.22589
19	1 38806	-41.867	0.07153	-0.13853

MFANI INF COORDINATES WITH ORIGIN AT SECTION AXIS

PT	PCT	AI	1/C	ALPHA	UPSILON	ZETA*
1	0	0.00680	-1.54614	1.04166	45.513	
2	0.0250	0.01203	-1 47279	0.96733	45.244	
3	0.0500	0.01720	1.39943	0.89371	44.941	
4	0.0750	0.02231	-1.32608	0.82102	44.515	
5	0.1000	0.02732	1.25272	0.74958	43.940	
6	0.1250	0.03223	-1.17937	0.67980	43.164	
7	0.1500	0.03703	1.10601	0.61213	42.190	
8	0.1750	0.04171	-1.03266	0.54690	41.074	
9	0.2000	0.04627	0.95930	0.48434	39.824	
10	0.2310	0.05157	0.87128	0.41293	38.261	
11	0.2610	0.05667	-0.78325	0.34547	36.653	
12	0.2910	0.06156	-0.69522	0.28186	35.032	
13	0.3210	0.06624	-0.60720	0.22201	33.407	
14	0.3510	0.07070	0.51917	0.16573	31.767	
15	0.3810	0.07492	0.43115	0.11296	30.112	
16	0.4110	0.07890	0.34312	0.06361	28.427	
17	0.4410	0.08259	-0.25509	0.01765	26.707	

PLATE III MOTOR

•71°C.

STAGF 3 ROTOR

COORDINATE SYSTEM ORIGIN 2 -7.04880 R O.

SECTION NO 9 SECTION JJ

MANIFOLD COORDINATES WITH ORIGIN AT SECTION AXIS

RT	R/C	AI	T/C	ALPHA	UPSILON	WNB	20
18	0 47910	0 08598	0 16707	0 02500	24 992		
19	0 56000	0 08901	-0 07904	-0 06447	23 313		
20	0 57000	0 09162	0 08898	-0 10092	21 682		
21	0 64700	0 09375	0 09701	0 13452	20 109		
22	0 59100	0 09529	0 18504	-0 16540	18 544		
23	0 62400	0 09603	0 27306	-0 19357	16 931		
24	0 65000	0 09575	0 36109	-0 21896	15 227		
25	0 68000	0 09415	0 44912	-0 24142	13 376		
26	0 71000	0 09103	0 53714	-0 26079	11 430		
27	0 74000	0 08650	0 62517	-0 27702	9 463		
28	0 77000	0 08082	0 71319	-0 29008	7 369		
29	0 80000	0 07430	0 80122	-0 29967	5 020		
30	0 83000	0 06725	0 88925	-0 30510	1 805		
31	0 86000	0 05987	0 97727	-0 30454	-2 812		
32	0 89000	0 05227	1 06530	0 29562	-8 972		
33	0 92000	0 04434	1 15333	0 27580	16 660		
34	0 95000	0 03614	1 24135	-0 24118	-26 472		
35	0 97000	0 02934	1 31471	-0 19722	-35 053		
36	1 00000	0 02262	1 38806	-0 13853	-41 867		

CIRC/RD
+ 1627SLACGFR
21.911
CAMFR
87.380

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

RT	T/C	ALPHA	UPPER	UPPER	LOWER	UPPER
1	0 00620	-1 54614	1 04166	-1 54614	1 04166	
2	0 00780	-1 54962	1 03318	-1 53771	1 04499	
3	0 00680	-1 54690	1 02588	-1 53037	1 04215	
4	0 01203	-1 48629	0 95394	1 45929	0 45072	
5	0 01720	-1 41865	0 87446	-1 38022	0 91297	
6	0 02231	-1 35081	0 79587	-1 30135	0 84618	
7	0 02732	-1 28270	0 71847	-1 22275	0 78069	
8	0 03223	-1 21423	0 64263	-1 14450	0 71697	
9	0 03703	-1 14534	0 56875	-1 06669	0 65554	
10	0 04171	-1 07599	0 49718	-0 98932	0 59662	
11	0 04627	-1 00616	0 42814	0 91245	0 54053	
12	0 05157	0 92177	0 34890	-0 82078	0 47696	
13	0 05667	0 83675	0 27358	-0 72975	0 41737	
14	0 06156	-0 75111	0 20217	-0 63934	0 36159	

PLATE III ROTOR

•714•

COORD SYSTEM ORIGIN Z -7.04800 R O.
 SECTION NO 9 SECTION JJ
 RAD 0.04003

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	T/C	UPPER ALPHA	UPPER POSITION	LOWER ALPHA	POSITION
15	0 .06624	-0 .66487	0 .13457	-0 .54953	0 .36945
16	0 .07070	-0 .57803	0 .07069	-0 .46032	0 .26078
17	0 .07412	0 .49058	0 .01047	-0 .37171	0 .21545
18	0 .07860	-0 .40251	0 .04611	-0 .28373	0 .17333
19	0 .08259	-0 .31379	0 .09903	-0 .19640	0 .13437
20	0 .08708	-0 .22451	0 .14823	-0 .10963	0 .09823
21	0 .08901	-0 .13474	-0 .19372	-0 .02334	0 .06479
22	0 .09162	-0 .04455	0 .23556	0 .06251	0 .033372
23	0 .09375	0 .04604	0 .27374	0 .14798	0 .004470
24	0 .09529	0 .13712	-0 .30826	0 .23296	-0 .02254
25	0 .09693	0 .22884	-0 .33884	0 .31729	-0 .04829
26	0 .09575	0 .32132	-0 .36505	0 .40086	-0 .01287
27	0 .09445	0 .41467	-0 .38627	0 .48356	-0 .09657
28	0 .09403	0 .50862	-0 .40188	0 .56567	-0 .11970
29	0 .08650	0 .60268	-0 .41194	0 .64766	-0 .14210
30	0 .08082	0 .69680	0 .41682	0 .72959	0 .16334
31	0 .07430	0 .79094	-0 .41672	0 .81150	-0 .18262
32	0 .06725	0 .88590	0 .41140	0 .89260	-0 .19880
33	0 .05987	0 .98192	-0 .39907	0 .97263	-0 .20394
34	0 .05227	1 .07819	0 .37726	1 .05241	0 .21397
35	0 .04434	1 .17343	0 .34296	1 .13322	-0 .20863
36	0 .03614	1 .26683	-0 .29233	1 .21588	-0 .19002
37	0 .02934	1 .34135	-0 .23520	1 .28806	-0 .15924
38	0 .02262	1 .38628	0 .19254	1 .33925	0 .12901
39	0 .02262	1 .39686	0 .17260	1 .36597	0 .12541
40	0 .02262	1 .38806	-0 .13853	1 .38806	-0 .13853
41	RAD 0 01163	CENFR AT ALPHA	-1 .53799	UPSLON	1 .03336
42	RAD 0 04003	CENTER AT ALPHA	1 .35765	UPSLON	-0 .16456

PHASE III ROTOR

*714.

	STAGE	3	ROTOR	WR	WD
CORD SYSTEM ORIGIN	2	-7.04880	R 0	W1	0.
SECTION NO	9	SECTION .09		W2	4.50000
CLINOID				CAMBFR	
3 1627				87 380	
ARFA	0 653040	SURFACE ARC LENGTH	6.86723		
SECTION C.G.		ALPHA	UP5110N		
SURFACE SURFACE SECTION C.G.		-0.00262	0.00462		
PI A/R AXIS		0.02762	-0.04430		
SLACKING AXIS (RADIAL)		0.02762	0.04430		
		0.00210	0		

PLATE III ROTOR

*7P*4

CORD CENTER ORIGIN Z -7.0000 R 0.
 SECTION NO 10 SECTION KK

MEAN INF INPUT DATA

PI	ALPHA	ZETA	THICKNESS	UPSILON
1	1.50909	43.941	0.02406	0.90496
2	-1.43057	43.561	0.04193	0.83029
3	-1.27469	42.412	0.07788	0.68549
4	1.12025	40.431	0.11430	0.54916
5	0.96771	37.671	0.15107	0.42467
6	0.80066	34.266	0.19097	0.30325
7	0.62075	30.512	0.23193	0.18860
8	0.44277	26.847	0.26772	0.09087
9	0.26670	23.311	0.29578	0.00820
10	0.09278	19.951	0.31389	-0.06091
11	0.07877	16.862	0.32114	-0.11793
12	0.24739	13.859	0.31758	-0.16414
13	0.41247	10.450	0.30532	-0.19969
14	0.57286	5.935	0.26880	-0.22310
15	0.72762	-1.538	0.26919	0.22984
16	0.87574	-14.246	0.24258	-0.20952
17	1.01635	-30.971	0.20365	-0.14875
18	1.14898	-47.058	0.15542	-0.03466
19	1.25280	-57.710	0.11964	0.11070

MEAN LINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	R/C	I/C	ALPHA	UPSILON	ZETA
1	0	0.00837	-1.50309	0.90496	43.941
2	0.0550	0.01384	1.44004	0.83922	43.338
3	0.0500	0.01934	-1.37099	0.77438	43.038
4	0.0750	0.02489	-1.30195	0.71041	42.558
5	0.1000	0.03050	-1.23290	0.64772	41.886
6	0.1250	0.03616	-1.16385	0.58669	41.033
7	0.1500	0.04189	1.09480	0.52765	39.994
8	0.1750	0.04766	-1.02576	0.47090	38.825
9	0.2000	0.05345	-0.95671	0.41658	37.534
10	0.2300	0.06039	-0.87385	0.35477	35.890
11	0.2600	0.06724	-0.79100	0.29666	34.179
12	0.2900	0.07393	-0.70814	0.24221	32.440
13	0.3200	0.08036	0.62528	0.19128	30.705
14	0.3500	0.08644	0.54243	0.14375	28.980
15	0.3800	0.09208	0.45957	0.09946	27.270
16	0.4100	0.09717	0.37671	0.05628	25.580
17	0.4400	0.10162	0.29186	0.02009	23.914

PHASE III ROTOR

.744.

CHORD SECTION ORIGIN 2 -7.04880 R 0
 SECTION NO 10 SECTION KK

MEAN LINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	P/C	AI	T/C	ALPHA	UPSILON	ZETA
18	0.4700	0.10533	-0.21100	-0.01524	22.273	
19	0.5000	0.10824	0.12814	0.04782	20.660	
20	0.5300	0.11030	0.04526	-0.07776	19.081	
21	0.5600	0.11149	0.03757	-0.10519	17.559	
22	0.5900	0.11178	0.12043	-0.13024	16.090	
23	0.6200	0.11118	0.20329	-0.15298	14.596	
24	0.6500	0.10973	0.28614	0.17338	13.051	
25	0.6800	0.10758	0.16900	0.19133	11.760	
26	0.7100	0.10494	0.45186	-0.20660	9.467	
27	0.7400	0.10196	0.53471	0.21875	7.131	
28	0.7700	0.09869	0.61757	-0.22707	4.179	
29	0.8000	0.09501	0.70043	-0.23024	-0.021	
30	0.8300	0.09065	0.78328	-0.22627	-5.836	
31	0.8600	0.08515	0.86614	-0.21202	-14.022	
32	0.8900	0.07799	0.94900	-0.18379	-23.566	
33	0.9200	0.06906	1.03185	-0.13899	-33.094	
34	0.9500	0.05878	1.14471	-0.07200	-44.856	
35	0.9700	0.05009	1.18376	-0.00996	-53.785	
36	1.0000	0.04163	1.25280	0.11070	-56.409	

(CHORD
2 R / 4RSTAGGER
16.044CAMBER
1(X) 350

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	P/C	AI	UPPER	LOWER	UPMIA	UPSILOM
1	0.00837	-1.50909	0.90496	-1.50909	0.90496	0.90496
2	0.00817	-1.26216	0.89533	-1.49978	0.90998	0.90998
3	0.00837	-1.50943	0.88726	-1.49145	0.90607	0.90607
4	0.01384	-1.45369	0.82476	-1.42639	0.85369	0.85369
5	0.01934	-1.38996	0.75407	-1.35203	0.79469	0.79469
6	0.02489	-1.32614	0.68406	-1.27776	0.73676	0.73676
7	0.03050	-1.26216	0.61509	-1.20364	0.68035	0.68035
8	0.03616	-1.19797	0.54749	-1.12974	0.62589	0.62589
9	0.04189	-1.13349	0.48154	-1.05612	0.57376	0.57376
10	0.04766	-1.06869	0.41755	-0.98283	0.52424	0.52424
11	0.05345	-1.00350	0.35567	-0.90992	0.47748	0.47748
12	0.06039	-0.92472	0.28447	-0.82298	0.42507	0.42507
13	0.06724	0.84528	0.21672	-0.73671	0.37660	0.37660
14	0.07413	-0.76512	0.15255	-0.65115	0.31186	0.31186

PHASE III ROTOR

•70C•

STAGE 3 ROTOR
 COORD SYSTEM ORIGIN Z -7.04880 R O.
 SECTION NO 10 SECTION KK
 SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

RI	I/C	ALPHA	UPPER UPSILON	LOWER ALPHA	UPSILON
15	0.08036	-0.68424	0.09200	-0.56632	0.29057
16	0.08644	-0.60261	0.03509	-0.48225	0.25241
17	0.09208	-0.52019	0.01815	-0.39895	0.21707
18	0.09717	-0.43700	0.06766	-0.31642	0.18422
19	0.10162	-0.35305	-0.11339	-0.23467	0.15358
20	0.10533	-0.26836	-0.15530	-0.15363	0.12482
21	0.10824	-0.18301	0.19335	-0.07327	0.09771
22	0.11030	0.09710	0.22755	0.00653	0.07203
23	0.11149	-0.01076	0.25792	0.08590	0.04755
24	0.11178	0.07591	0.28456	0.16494	0.02408
25	0.11116	0.16303	0.30758	0.24354	0.00164
26	0.10973	0.25054	0.32699	0.32175	0.01978
27	0.10758	0.33855	0.34283	0.39945	0.03978
28	0.10494	0.42705	0.35533	0.47495	0.05746
29	0.10196	0.51652	0.36413	0.55290	0.07337
30	0.09869	0.60723	0.36850	0.62790	0.08564
31	0.09501	0.70048	0.36676	0.70038	0.09372
32	0.09065	0.79653	0.35585	0.77004	0.09669
33	0.08515	0.89579	0.33072	0.83649	0.09331
34	0.07999	0.99380	0.28650	0.90419	0.08107
35	0.06906	1.08604	0.22213	0.97167	0.05586
36	0.05818	1.17428	0.13187	1.05514	0.01213
37	0.04619	1.24183	0.03257	1.12569	0.05248
38	0.04163	1.27402	0.02398	1.16781	0.10118
39	0.04163	1.28179	0.06240	1.20866	0.12152
40	0.04163	1.25280	0.11071	1.25280	0.11070
41	IF PAD 0.01305	CENTER AT ALPHA	-1.49967	UPSILON	0.80593
42	IF RAD 0.06620	CENTER AT ALPHA	1.21593	UPSILON	0.05572

PLATE III ROTOR

*70C.

COMPUTED CIRCUMFERNCE (ORIGIN Z -7.00000 P 0)	STAGE	3	ROTOR	HR	20
SECTION NO. 10	SECTION KK		MU	O.	FIA O
(UNROTATED)	SLAGGER		PHD	A XXXX	
2 R/30	16.004		CARRIER		
ALPHA 0.690003	SURFACE ARC LENGTH	6.56279	1KH. 150		
SECTION C G.	ALPHA	UPSETION			
CIRFAM SURFACE SECTION C.G.	0.01407	0.01066			
BLADE AXIS	0.02644	-0.01869			
SLACING AXIS (RADIAL)	-0.02644	0.01869			
	0.00210	0.			

PIAOF III ROTOR

COORD SYSTEM ORIGIN 2 -7.00000 R 0
 SECTION NO 11 SECTION LI
 MEANLINE INPUT DATA

PI	ALPHA	ZETA*	THICKNESS	UPSILON
1	-1.48469	41.825	0.02987	0.78336
2	-1.40871	41.056	0.05116	0.71579
3	-1.25816	39.143	0.09492	0.58764
4	-1.10917	36.700	0.13849	0.47057
5	-0.96156	33.924	0.17967	0.36579
6	-0.80088	30.818	0.22031	0.26422
7	-0.62764	27.458	0.25759	0.16807
8	-0.45653	24.094	0.28678	0.08592
9	-0.28763	20.825	0.30795	0.01670
10	-0.12146	17.909	0.32197	-0.04108
11	-0.04162	15.329	0.33055	-0.08922
12	0.20104	12.617	0.33634	-0.12866
13	0.35597	8.898	0.33914	-0.15804
14	0.50556	2.492	0.33719	-0.17776
15	0.64984	-9.251	0.32794	-0.16303
16	0.75400	-26.414	0.30457	-0.11434
17	0.91046	-44.165	0.26127	-0.01241
18	1.02766	-58.058	0.20737	0.15672
19	1.11752	-66.214	0.16776	0.35993

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	P/C	ALPHA	THICKNESS	ZETA*
1	0	0.01133	1.48469	0.78336
2	0.0250	0.01823	-1.41964	0.72542
3	0.0500	0.02530	-1.35458	0.66870
4	0.0750	0.03250	-1.28953	0.61359
5	0.1000	0.03977	-1.22447	0.56025
6	0.1250	0.04700	1.15941	0.50884
7	0.1500	0.05414	-1.09436	0.45955
8	0.1750	0.06112	-1.02930	0.41247
9	0.2000	0.06788	0.96425	0.36760
10	0.2300	0.07562	-0.88618	0.31659
11	0.2600	0.08291	-0.80812	0.26853
12	0.2900	0.08968	-0.73005	0.22329
13	0.3200	0.09589	0.65198	0.18077
14	0.3500	0.10150	0.57392	0.14093
15	0.3800	0.10650	0.49585	0.10370
16	0.4100	0.11087	0.41778	0.063903
17	0.4400	0.11463	0.33972	0.03685

PHASE III ROTOR

•/PC.

STAGE 3 ROTOR

COORD SYSTEM ORIGIN 7 -7 04880 R 0.
SECTION NO 14 SECTION II

M&I AMI INF COORDINATES WITH ORIGIN AT SECTION AXIS

NI	PX	PY	PZ	I/C	ALPHA	UPPER	RHO	THETA	PHI
18	0.4700	0	11780	0	26165	0	0.00703	20.117	
19	0.5000	0	12041	0	18358	-0.02058	18.785		
20	0.5300	0	12251	-0	10552	0.04614	17.482		
21	0.5600	0	12417	0	02745	-0.06979	16.242		
22	0.5900	0	12552	0	05061	-0.09165	15.051		
23	0.6200	0	12669	0	12868	0.11175	13.792		
24	0.6500	0	12764	0	20675	0.12992	12.386		
25	0.6800	0	12833	0	28484	-0.145RA	10.654		
26	0.7100	0	12865	0	36288	-0.15908	8.449		
27	0.7400	0	12848	0	44095	-0.16885	5.325		
28	0.7700	0	12772	0	51901	0.17308	0.938		
29	0.8000	0	12614	0	59708	-0.17030	-5.382		
30	0.8300	0	12322	0	67515	-0.15731	-13.856		
31	0.8600	0	11825	0	75321	0.13011	-24.767		
32	0.8900	0	11034	0	83128	-0.08357	-36.479		
33	0.9200	0	09928	0	90334	-0.01360	-46.668		
34	0.9500	0	08580	0	98741	0.08652	-57.325		
35	0.9750	0	07440	1	05247	0.20789	-65.262		
36	1.0000	0	06363	1	11752	0.35993	-67.552		
37	1.0364	2	6364	2	STAGGER 9 242	CAMBER 109.377			

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

NI	I/C	ALPHA	UPPER	UPPER ALPHA	LOWER ALPHA	POSITION
1	0.01133	-1 48469	0	78336	-1 48469	0 78336
2	0.01133	-1 48886	0	77113	-1 47326	0 78884
3	0.01133	-1 48462	0	76118	-1 46265	0 18571
4	0.01823	1 43555	0	70741	-1 40372	0 74344
5	0.02530	1 37632	0	64341	-1 33284	0 69398
6	0.03250	-1 31697	0	58068	-1 26208	0 64649
7	0.03977	-1 25735	0	51942	-1 19159	0 60107
8	0.04700	-1 19735	0	45985	-1 12148	0 55783
9	0.05414	-1 13684	0	40219	-1 05148	0 51690
10	0.06112	-1 07580	0	34667	-0 98281	0 47R27
11	0.06818	-1 01421	0	29337	-0 91429	0 44163
12	0.07562	-0 93957	0	23240	-0 87274	0 40017
13	0.08291	-0 86416	0	17470	-0 75207	0 36237
14	0.09068	-0 78796	0	12023	0 67213	0 32635

PHASE III ROTOR

71°C

COORD SYSTEM ORIGIN Z -7.04830 R O.
SECTION NO 11 SECTION 11
SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	L/C	UPPER ALPHA	UPPER BETA	UPPER GAMMA	LOWER ALPHA	LOWER BETA	LOWER GAMMA	UPSILON
15	0.09589	-0.71095	0.06897	-0.59302	0.29258			
16	0.10150	-0.63313	0.02095	-0.51470	0.26091			
17	0.10650	-0.55456	0.02382	-0.43714	0.23122			
18	0.11087	-0.47528	-0.06533	-0.36729	0.20340			
19	0.11463	-0.39546	-0.10359	-0.28398	0.17730			
20	0.11780	-0.31521	-0.13875	-0.20609	0.15278			
21	0.12041	-0.23470	-0.17085	-0.13247	0.12959			
22	0.12251	-0.15403	-0.20017	-0.05700	0.10790			
23	0.12417	-0.07323	-0.22604	0.01833	0.08736			
24	0.12552	0.00765	-0.25144	0.09358	0.06814			
25	0.12669	0.08887	-0.27393	0.16849	0.05044			
26	0.12764	0.17066	-0.29425	0.24284	0.03442			
27	0.12833	0.25354	-0.31213	0.31609	0.02036			
28	0.12865	0.33796	-0.32682	0.38780	0.00867			
29	0.12848	0.42523	-0.33728	0.45666	0.00002			
30	0.12772	0.51626	-0.34141	0.52177	-0.00474			
31	0.12614	0.61268	-0.33584	0.58148	-0.00476			
32	0.12322	0.71404	-0.31501	0.63625	0.00040			
33	0.11825	0.81852	-0.27166	0.68791	0.01143			
34	0.11034	0.91775	-0.20053	0.74480	0.03338			
35	0.09928	1.00454	-0.10340	0.81415	0.07621			
36	0.08580	1.08261	0.02546	0.89221	0.14758			
37	0.07440	1.14154	0.16685	0.96339	0.24693			
38	0.06363	1.16898	0.24711	1.00345	0.32232			
39	0.05363	1.16935	0.30300	1.05281	0.36227			
40	0.05363	1.11752	0.35993	1.11752	0.35993			
LF RAD	0.01655	CENTER AT ALPHA	-1.47236	UNPSILON	0.77231			
RF RAD	0.09146	CENTER AT ALPHA	1.08208	UNPSILON	0.27562			

PHASE III MOTOR

•7PC•

COORD SYSTEM ORIGIN	7	7	04480	R	0	MU	0	FIA	0
SECTION NO.	11		SECTION	11		PWD	3.5000		
CENTER			STAGGER			CAMFR			
2	6364		9.242			109.377			
AREA	0	756746	SURFACE	ARC LENGTH	6.39540				
SECTION C.G.			ALPHA			UPSILON			
SURFACE			0.02810			0.04281			
AIR AXIS			-0.00222			-0.00722			
STACKING			-0.00222			0.00122			
AXIS (RADIAL)			-0.00210		0.				

PLATE III RODIR

7000.

COORD SYSTEM ORIGIN Z -7.04880 R O.
 SECTION NO 12 SECTION MM
 MU O FIA O
 RHM 3.0000

MFANI INF INPUT DATA

PT	ALPHA	7FIA*	THICKNESS	UPSILON
1	1 46988	38.315	0.04052	0.68852
2	1 39601	37.331	0.06329	0.63176
3	1 24977	35.338	0.10706	0.52561
4	-1 10532	33.313	0.14741	0.42853
5	0 96254	31.238	0.18362	0.33970
6	0 80728	28.752	0.21844	0.25107
7	0 64008	25.568	0.25072	0.16584
8	-0.47513	22.079	0.27763	0.09339
9	0.31283	18.779	0.30004	0.03336
10	-0.15351	16.208	0.31908	-0.01642
11	0.00208	14.074	0.33633	-0.05836
12	0 15324	11.543	0.35572	0.09265
13	0 29881	7.390	0.37441	0.11645
14	0.43810	-0.971	0.38608	-0.12247
15	0 57006	-16.643	0.38670	0.09622
16	0 69226	-36.484	0.36656	-0.01917
17	0 80458	-53.316	0.31889	0.12393
18	0 90634	-64.882	0.25833	0.34811
19	0 98224	-71.305	0.21587	0.60917

MFANI LINE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	PCT A	T/C	ALPHA	UPSILON	ZETA*
1	0	0.01651	-1.46988	0.68852	38.315
2	0 0250	0.02422	-1.40858	0.64123	37.089
3	0 0500	0.03184	1 34727	0.59554	36.309
4	0 0750	0.03931	-1.28597	0.55116	35.475
5	0 1000	0.04659	-1.22467	0.50819	34.577
6	0 1250	0.05364	-1 16336	0.46664	33.689
7	0 1500	0.06044	-1 10206	0.42643	32.834
8	0 1750	0.06695	-1.04076	0.38751	31.985
9	0 2000	0.07318	-0.97846	0.34986	31.122
10	0 2300	0.08025	0.90589	0.30637	30.050
11	0 2600	0.08688	0.83239	0.26478	28.899
12	0 2900	0.09307	0.75877	0.22519	27.659
13	0 3200	0.09885	0.68520	0.18772	26.304
14	0 3500	0.10422	-0.61164	0.15251	24.833
15	0 3800	0.10920	0.53807	0.11964	23.305
16	0 4100	0.11280	-0.46451	0.08913	21.736
17	0 4400	0.11808	0.39095	0.06094	20.211

FRW I III RIOR

.7PC.

COORD SYSTEM ORIGIN 7 -7 0480 R O.
 SECTION NO 12 SECTION MM
 RUM 3.0000

STAGE 3. ROTOR
 R1 I AI 1/C ALPHA UPSILON ZFIA*

R1	I/C	ALPHA	UPSILON	ZFIA*
18	0 4700	0 12206	0 31738	0 03491 18 783
19	0 5000	0 12579	0 24382	0 01033 17 488
20	0 5300	0 12929	0 17026	0 01154 16 368
21	0 5600	0 13259	0 09669	0 03245 15 370
22	0 5900	0 13590	0 02313	0 05198 14 363
23	0 6200	0 13949	0 05043	0 07011 13 288
24	0 6500	0 14339	0 12400	0 08661 11 957
25	0 6800	0 14743	0 19756	0 10113 10 261
26	0 7100	0 15129	0 27112	0 11292 7 821
27	0 7400	0 15453	0 34469	0 12093 4 328
28	0 7700	0 15889	0 41825	0 12349 -1 169
29	0 8000	0 15819	0 49181	0 11690 -9 118
30	0 8300	0 15773	0 56538	0 09796 20 073
31	0 8600	0 15443	0 63894	0 06112 -33 179
32	0 8900	0 14679	0 71251	0 00083 46 310
33	0 9200	0 13396	0 78607	0 09469 -56 587
34	0 9500	0 11682	0 85963	0 22835 -65 634
35	0 9700	0 10184	0 92094	0 39323 72 703
36	1 0000	0 08199	0 98224	0 60917 -74 791

CHORD
2 45.34

STAGGER
1.853

CAMBER
113.106

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

R1	I/C	ALPHA	UPPER	UPPER	LOWFR	ALPHA	UPPER
1	0 01651	1 46988	0 68052	-1 46988	0 68052	0 68052	0 68052
2	0 01651	1 47452	0 67134	-1 45493	0 67134	0 67134	0 67134
3	0 01651	1 46798	0 65820	-1 44011	0 65820	0 65820	0 65820
4	0 02422	-1 42650	0 61752	-1 39066	0 61752	0 61752	0 61752
5	0 03184	-1 37040	0 56406	-1 32414	0 56406	0 56406	0 56406
6	0 04931	-1 31396	0 51189	-1 25798	0 51189	0 51189	0 51189
7	0 04659	-1 25710	0 46113	-1 19223	0 46113	0 46113	0 46113
8	0 01364	1 19986	0 41189	-1 12687	0 41189	0 41189	0 41189
9	0 06044	-1 14226	0 36144	-1 06186	0 36144	0 36144	0 36144
10	0 06695	-1 08426	0 31785	0 99725	0 31785	0 31785	0 31785
11	0 07318	-1 02585	0 27301	0 93306	0 27301	0 27301	0 27301
12	0 09025	0 95519	0 22116	-0 85660	0 95519	0 95519	0 95519
13	0 09688	0 88383	0 17149	-0 78083	0 88383	0 88383	0 88383
14	0 09307	0 81177	0 12406	-0 70577	0 81177	0 81177	0 81177

PLATE III ROTOR

COORD SYSTEM ORIGIN Z -7.0480 R O.
SECTION NO 12 SECTION MM
RAD 3.0MM

SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	I/C	STAGE	3. ROTOR	MU	NB	20
			UPPER	UPPER	LOWER	UPPER
			ALPHA	UPSILO	ALPHA	UPSILO
15	0	0.0885	-0.73894	0.07901	-0.63147	0.29642
16	0	10422	-0.66533	0.03648	-0.55795	0.26853
17	0	10900	-0.59107	0.00338	0.48508	0.24266
18	0	11180	-0.51621	-0.04055	-0.41281	0.21881
19	0	11808	-0.44099	0.07498	-0.34091	0.19687
20	0	12206	-0.36560	-0.10685	-0.26917	0.17666
21	0	12579	-0.29019	-0.13635	-0.19745	0.15801
22	0	12929	-0.21495	-0.16371	-0.12556	0.14064
23	0	13259	-0.13980	-0.18928	-0.05358	0.12438
24	0	13590	-0.06448	-0.21348	0.01822	0.10952
25	0	13949	0.01110	-0.23664	0.08976	0.09642
26	0	14339	0.08755	-0.25869	0.16044	0.08547
27	0	14743	0.16534	-0.27909	0.22938	0.07634
28	0	15129	0.24587	0.39679	0.29638	0.07095
29	0	15453	0.33038	-0.30996	0.35899	0.06809
30	0	15889	0.42218	-0.31561	0.41432	0.06924
31	0	15819	0.52257	0.30851	0.46106	0.07470
32	0	15773	0.63778	-0.27970	0.49897	0.08377
33	0	15443	0.74261	-0.21967	0.53527	0.09743
34	0	14679	0.84271	-0.12355	0.58230	0.12521
35	0	13396	0.92324	0.00420	0.64890	0.18518
36	0	11682	0.99017	0.16923	0.72909	0.28747
37	0	10184	1.04021	0.35608	0.80166	0.43037
38	0	08779	1.06507	0.47434	0.84351	0.54098
39	0	08799	1.05637	0.54648	0.89843	0.60076
40	0	08799	0.98224	0.60917	0.98224	0.60917
41	RAD	0.02288	CENTER AT ALPHA	-1.45186	UPSILO	0.67443
42	RAD	0.11619	CENTER AT ALPHA	0.95114	UPSILO	0.49721

PILOT III ROTOR

COORD SYSTEM ORIGIN		STAGE	3. ROTOR	NP	20
SECTION NO	12	SECTION MM	MU	O	FIA
CHORD		STAGGER	RIM	3.0000	
2.4134		1 853	CAMBER		
ALPHA	0 911752	SURFACE ARC LENGTH	6.34665		
SECTION C.G.		ALPHA	OF SECTION		
GIRDER SURFACE SECTION C.G.			0.06690	0.09445	
PLATE AXIS			0.04555	0.00062	
SLACKING AXIS (RADIAL)			0.04555	0.00062	
			-0.00210	0	

PILOT III ROTOR

COORD SYSTEM ORIGIN 2 -7.04480 R O.
SECTION NO 13 SECTION NN

STAGE 3 ROTOR

RHO 0.
RHO 2 5000
FIA 0.

MEANLINE INPUT DATA

PI	ALPHA	ZETA*	THICKNESS	UPSILON
1	1.45592	33.474	0.05425	0.61515
2	1.38439	32.690	0.07465	0.56921
3	1.24263	31.311	0.11306	0.48203
4	1.10273	30.107	0.14803	0.39988
5	0.96463	28.859	0.17963	0.32208
6	0.81455	26.924	0.21045	0.24238
7	0.65305	23.770	0.24014	0.16557
8	0.49401	20.058	0.26676	0.10159
9	-0.33815	16.695	0.29162	0.05024
10	0.18558	14.479	0.31615	0.00826
11	0.03746	12.805	0.34211	-0.02749
12	0.10545	10.461	0.37510	-0.05664
13	0.24165	5.873	0.40967	-0.07486
14	0.37064	-4.427	0.43497	0.07218
15	0.49128	23.508	0.44546	-0.02941
16	0.60052	-44.490	0.42855	0.07601
17	0.69870	-59.732	0.37651	0.26027
18	0.78502	-69.411	0.30928	0.53949
19	0.84695	-74.644	0.26399	0.85840

MEANLINE COORDINATES WITH ORIGIN AT SECTION AXIS

PI	RCL	AI	T/C	ALPHA	UPSILON	ZETA*
1	0	0	0.02343	-1.45592	0.61515	32.866
2	0	0.0250	0.03053	-1.39835	0.57810	32.560
3	0	0.0400	0.03748	-1.34078	0.54184	31.852
4	0	0.0750	0.04421	-1.28321	0.50650	31.259
5	0	0.1000	0.05073	-1.22564	0.47188	30.804
6	0	0.1250	0.05704	-1.16806	0.43783	30.389
7	0	0.1500	0.06314	-1.11049	0.40435	29.979
8	0	0.1750	0.06901	-1.05292	0.37142	29.556
9	0	0.2000	0.07466	-0.99535	0.33907	29.089
10	0	0.2300	0.08112	-0.92626	0.30112	28.442
11	0	0.2600	0.08725	-0.85718	0.26434	27.578
12	0	0.2900	0.09308	-0.7809	0.22907	26.476
13	0	0.3200	0.09863	-0.71900	0.19561	25.179
14	0	0.3400	0.10394	-0.64992	0.16419	23.692
15	0	0.3600	0.10903	-0.58083	0.13501	22.087
16	0	0.4100	0.11396	-0.51174	0.10813	20.434
17	0	0.4400	0.11876	-0.44266	0.08352	18.788

PHASE III ROTOR

•700•

COORD SYSTEM ORIGIN 2 -7.04880 R O.
 SECTION NO 13 SECTION MN RAD 2.5000
 SURFACE COORDINATES WITH ORIGIN AT SECTION AXIS

PT	T/C	UPPER ALPHA	UPPER UPSILON	LOWER ALPHA	LOWER UPSILON
15	0.09863	-0.76759	0.09226	0.67042	0.29876
16	0.10394	0.69827	0.05399	0.60156	0.27439
17	0.10903	0.62830	0.01804	-0.53336	0.25199
18	0.11346	-0.55781	-0.01552	-0.46568	0.23177
19	0.11876	-0.48694	0.04665	-0.39377	0.21370
20	0.12350	-0.41610	-0.07549	-0.33104	0.19756
21	0.12825	0.34542	0.10235	-0.26355	0.18314
22	0.13305	-0.27516	-0.12754	-0.19563	0.17011
23	0.13768	-0.20541	0.15142	0.12721	0.15813
24	0.14290	-0.13562	-0.17456	-0.05884	0.14732
25	0.14855	-0.06561	-0.19745	0.13829	0.13829
26	0.15514	0.00523	-0.22038	0.07667	0.13170
27	0.16249	0.07756	-0.24276	0.14251	0.12786
28	0.17021	0.15315	-0.26357	0.20509	0.12715
29	0.17757	0.23382	-0.28044	0.26259	0.12975
30	0.18386	0.32437	-0.28970	0.31022	0.13583
31	0.19985	0.42774	-0.28419	0.34501	0.14522
32	0.19196	0.54523	-0.25151	0.36570	0.15515
33	0.19173	0.66351	0.17912	0.38559	0.16713
34	0.18663	0.76578	-0.06283	0.42150	0.19613
35	0.17262	0.84157	0.09193	0.46386	0.27041
36	0.15181	0.89815	0.29077	0.56548	0.40442
37	0.13211	0.93881	0.52639	0.63996	0.59183
38	0.11463	0.96084	0.70379	0.68654	0.75972
39	0.11400	0.94262	0.79117	0.74555	0.83855
40	0.11400	0.84695	0.85840	0.84695	0.85840
L.F RAD	0.03059	CENTER AT ALPHA	-1.43024	UPSILON	0.59853
IT RAD	0.14045	CENTER AT ALPHA	0.82135	UPSILON	0.72030

WHALE TURBINE

.7100.

STAGGERED ROTOR

CHORD SYSTEM ORIGIN 7 -7.04880 R O.

SECTION NO 13 SECTION MN

CHORD
2.3157

AREA O 884509

STAGGER

.6 .030

SECTION C G.
SURFACE SECTION C G
RADIAL AXIS
SPACING AXIS (RADIAL)

SURFACE ARC LENGTH

6.38549

ALPHA

UPSILON

0.10295

0.16952

0.09331

0.005598

0.09331

0.005598

0.00210

0.

PRINTED IIII ROTOR

INC.

STRI	NN	RHO	STAGE	3.	ROTOR	NR	20
			CHORD	STAGGER CARRIER	TM/C	7FIA1-ZETAS	
AA	1	8 50000	4.0639	61.49	-3.48	0.02378	57.43 60.91
BB	2	8 00000	4.0029	56.60	1.74	0.02559	55.37 53.53
CC	3	7.50000	3.9854	52.29	6.24	0.03037	53.72 47.48
DD	1	7.00000	3.9558	48.24	11.44	0.03896	52.30 40.86
FF	5	6.50000	3.8854	43.95	18.51	0.05082	51.27 32.77
FF	6	6 00000	3.7695	38.82	28.85	0.06231	50.41 21.56
GG	7	5.50000	3.6694	33.21	45.13	0.07205	49.29 4.17
HH	8	5.00000	3.4555	27.03	65.20	0.08347	47.51 -17.69
JJ	9	4 50000	3.1627	21.91	87.38	0.09603	45.51 41.87
KK	10	4 00000	2.8738	16.04	100.35	0.11178	43.94 56.41
LL	11	3 50000	2.6364	9.24	109.38	0.12865	41.82 67.55
MM	12	3 00000	2.4534	1.85	113.11	0.15819	38.32 -74.79
NN	13	2 50000	2.3157	-6.03	112.59	0.19196	32.87 -79.77

THE COORDINATES FOR THIS BLADE WERE PUT ON TAPE
IN THE SAME ORDER AS PRINTED ABOVE

SECTION XV
REFERENCES

1. A.J. Wennerstrom, and W.A. Buzzell, Redesign of a Rotor for a 1500 ft/sec Transonic, High-Through-Flow, Single-Stage Axial-Flow Compressor with Low Hub/Tip Ratio, Air Force Aero Propulsion Laboratory, Wright-Patterson AFB, Ohio 45433, AFAPL-TR-2078, September 1979.
2. George R. Frost, Richard M. Hearsey, Arthur J. Wennerstrom, A Computer Program for the Specification of Axial Compressor Airfoils, Aerospace Research Laboratories, Wright-Patterson Air Force Base, Ohio 45433, ARL 72-0171,
3. Richard M. Hearsey, A Revised Computer Program for Axial Compressor Design Volume I, Aerospace Research Laboratories, Wright-Patterson Air Force Base, Ohio 45433, ARL TF 75-0001, January 1975.
4. Arthur J. Wennerstrom, Personal Communication to L.H. Smith of General Electric Company, September 12, 1980.